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**A COMPUTER PROGRAM VERSION
OF THE BROUWER ORBITAL THEORY
WITH OPTIONAL MODIFICATIONS**

HANS G. HERTZ

AUGUST 1969



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A COMPUTER PROGRAM VERSION OF THE BROUWER ORBITAL THEORY WITH OPTIONAL MODIFICATIONS

1. INTRODUCTION

This report describes a program which computes osculating elements and position-velocity vectors in an earth satellite orbit according to the Brouwer theory (Brouwer 1959). Options for certain modifications have been added. Basic orbital data are written on a tape called orbital tape which may be used as input to other programs for further processing.

The program written in Fortran IV for the IBM 360 computer is an adaptation of a program constructed and described by Repass and Chaplick (1965)*. It is intended to further revise and extend the program. In view of these intentions this report is to be considered as an interim report. No concentrated effort has been made to remove all imperfections in the documentation and to check out the program thoroughly. This will be done with the final version. Spot checks have shown that the program is working correctly and is giving correct results.

2. THE BROUWER THEORY

The Brouwer theory (Brouwer 1959) is a first order theory giving expressions for the osculating values of the Keplerian elements of a satellite orbit.

To some degree of accuracy the earth's potential at a point at distance r from the earth's center of mass and at latitude β with respect to the equator is given by (IAU 1962)

$$U = \frac{\mu}{r} \left[1 - \sum_{n=1}^{\infty} J_n \left(\frac{R}{r} \right)^n P_n(\sin \beta) \right] \quad (1)$$

where $P_n(x)$ is the Legendre polynomial of degree n . The constant $\mu = GM_E$ is the product of the gravitational constant G and the mass M_E of the earth and has the dimension length cubed/time squared. The quantity R is the mean equatorial radius of the earth. The coefficient J_n is called the coefficient of the zonal harmonic of order n or briefly the zonal harmonic of order n or the n 'th harmonic. Since the center of mass of earth is taken as origin $J_1 = 0$. The J_n are small quantities. It is customary to consider J_2 as a small quantity of the first order and all J_n with $n > 2$ as small quantities of at least second order.

*The primary purpose of this program is not to study orbits but to use the Brouwer theory in an application.

There are also longitude dependent terms in U which Brouwer neglects together with the zonal harmonics of order $n > 5$. For the purposes of his theory he writes

$$\begin{aligned}
 U = & \frac{\mu}{r} + \frac{\mu k_2}{r^3} (1 - 3 \sin^2 \beta) + \frac{\mu A_{30}}{r^4} \left(-\frac{3}{2} \sin \beta + \frac{5}{2} \sin^3 \beta \right) \\
 & + \frac{\mu k_4}{r^5} \left(1 - 10 \sin^2 \beta + \frac{35}{3} \sin^4 \beta \right) \\
 & + \frac{\mu A_{50}}{r^6} \left(\frac{15}{8} \sin \beta - \frac{35}{4} \sin^3 \beta + \frac{63}{8} \sin^5 \beta \right)
 \end{aligned} \tag{2}$$

where

$$\begin{aligned}
 k_2 &= \frac{1}{2} J_2 R^2 \\
 A_{30} &= - J_3 R^3 \\
 k_4 &= -\frac{3}{8} J_4 R^4 \\
 A_{50} &= - J_5 R^5
 \end{aligned} \tag{3}$$

In addition to the constant μ we shall also use the constant k defined by

$$k^2 = \mu \tag{4}$$

If a spherical earth were the only attracting body the potential due to the earth would be μ/r , i.e. all J_n with $n \geq 2$ would be zero. The orbit of the satellite would be Keplerian, i.e. the motion would take place in a fixed elliptic orbit according to the laws of Kepler. Using well known formulae, the rectangular coordinates and velocity components in megameters and megameters per hour respectively with respect to the equator and equinox of some epoch may be computed from the Keplerian elements

a = semi-major axis in megameters

e = eccentricity

$$\left. \begin{array}{l}
 I = \text{inclination} \\
 \Omega = \text{right ascension of ascending node} \\
 \omega = \text{argument of perigee} \\
 M = \text{mean anomaly}
 \end{array} \right\} \begin{array}{l}
 \text{with respect to} \\
 \text{equator and equinox}
 \end{array} \left. \vphantom{\begin{array}{l} I \\ \Omega \\ \omega \\ M \end{array}} \right\} \text{in degrees}$$

The elements a, e, I, Ω, ω are constants whereas M is a linear function of the time given by

$$M = M_0 + M_1 t \quad (5)$$

where

$$M_1 = k a^{-3/2} \quad (6)$$

is the mean motion of the satellite.

If τ is measured in hours and M_1 in degrees per hour then $\tau = 360/M_1$ is the period of revolution of the satellite in its closed orbit.

If a non-spherical earth is assumed, i.e. if the J_n with $n \geq 2$ are no longer all assumed to be zero, the same formulae for computing the coordinates and velocity components may be used but the quantities a, e, I, Ω, ω occurring in these formulae are no longer constants and M is no longer a linear function of the time. The variable values

$$a, e, I, \Omega, \omega, M$$

occurring in these formulae under the assumption of a non-spherical earth are called the osculating values of the Keplerian elements and are solutions of certain differential equations. It is the purpose of the Brouwer theory to find approximate expressions for the solutions of these differential equations under the assumption that all J_n with $n > 5$ are zero. The deviations of the expressions for the osculating values from the Keplerian values are of order J_2 but it must be mentioned that e occurs as a divisor.

It should be noted that in the following description of the Brouwer theory the unit of length is the megameter, the unit of time the hour and that angles are expressed in degrees. These are not the units employed by Brouwer in his paper.

In the Brouwer theory an epoch T and six constants

$$\begin{array}{l} a_0 \quad \text{in megameters} \\ e_0 \\ \left. \begin{array}{l} I_0 \\ \Omega_0 \\ \omega_0 \\ M_0 \end{array} \right\} \quad \text{in degrees} \end{array}$$

are chosen. Associated with them a solution of the equations of motion for $a, e, I, \Omega, \omega, M$ of the form

$$\begin{aligned} a_B &= a_0 + \delta a \\ e_B &= e_0 + \delta e \\ I_B &= I_0 + \delta I \\ \Omega_B &= \Omega_0 + \Omega_{10} t + \delta \Omega \\ \omega_B &= \omega_0 + \omega_{10} t + \delta \omega \\ M_B &= M_0 + M_{10} t + \delta M \end{aligned} \tag{7}$$

is constructed. The subscript B has been added to indicate that we deal here with approximate expressions based on the Brouwer theory proper.

In these expressions t is time in hours from the epoch. The quantities

$$\Omega_{10}, \omega_{10}, M_{10}$$

are constants expressed in degrees per hour and are called the secular motions of Ω_B, ω_B, M_B . The linear functions

$$\begin{aligned} \Omega'' &= \Omega_0 + \Omega_{10} t \\ \omega'' &= \omega_0 + \omega_{10} t \\ M'' &= M_0 + M_{10} t \end{aligned} \tag{8}$$

are called the secular portions of Ω_B, ω_B, M_B respectively. The secular portions of a_B, e_B, I_B are the constants defined by

$$\begin{aligned} a'' &= a_0 \\ e'' &= e_0 \\ I'' &= I_0 \end{aligned} \tag{9}$$

and thus have no linear terms.

The secular motion M_{10} of M_B is approximately equal to the mean motion the satellite would have if it were moving in a Keplerian orbit close to the actual orbit and is of the order of 180° per hour for a close earth satellite. It is computed to order J_2^2 . No confusion will arise if we call $360/M_{10}$ the period of the satellite even though the actual orbit is not closed and the motion is not periodic. But after a period which is of the order of 2 hours the satellite will return to approximately the same position in space. This is so since M_{10} deviates only by terms at least of the order J_2 from the value it would have in a nearby Keplerian orbit and since Ω_{10} and ω_{10} being at least of order J_2 are also small. Since Ω_{10} and ω_{10} which are computed to order J_2^2 are thus much smaller than M_{10} it will take the secular portions of Ω_B and ω_B to complete an arc of 360° in a time much longer than the period of satellite. In general, times of the order of 100 days are required.

The quantities

δa in megameters

δe

$\delta I, \delta \Omega, \delta \omega, \delta M$ in degrees

are series of periodic terms and are called the periodic perturbations of the respective elements. They are of order J_2 and are computed to that order only.

The Brouwer theory thus represents the osculating elements as the sums of secular portions and periodic perturbations. The periodic perturbations are not necessarily equal to zero at the epoch ($t = 0$) so that

$$a_0, e_0, I_0, \Omega_0, \omega_0, M_0$$

are not necessarily equal to the osculating values of

$$a, e, I, \Omega, \omega, M$$

at the epoch. A theory could have been constructed where this would have been the case but the periodic perturbations would have been much greater and the accuracy of the theory would have suffered.

For this reason

$$a_0, e_0, I_0, \Omega_0, \omega_0, M_0$$

are called the Brouwer mean elements for epoch T. However, the first three of these quantities are independent of the epoch. The latter three change linearly with the epoch. In particular

$$\left. \begin{aligned} a_0 &= \text{Brouwer mean semi-major axis in megameters} \\ e_0 &= \text{Brouwer mean eccentricity} \\ I_0 &= \text{Brouwer mean inclination} \\ \Omega_0 &= \text{Brouwer mean right ascension of the node} \\ \omega_0 &= \text{Brouwer mean argument of perigee} \\ M_0 &= \text{Brouwer mean mean anomaly} \end{aligned} \right\} \text{ in degrees}$$

There are two types of periodic terms occurring in the periodic terms. The long-period terms are terms whose periods are long compared with the period of the satellite while the short-period terms have periods of comparable size with the period of the satellite or less. In particular terms whose arguments are linear combinations with small integer coefficients of the secular portions of Ω_B and ω_B are long-period terms.

We thus write

$$\begin{aligned} \delta a &= \delta_L a + \delta_S a \\ \delta e &= \delta_L e + \delta_S e \\ \delta I &= \delta_L I + \delta_S I \\ \delta \Omega &= \delta_L \Omega + \delta_S \Omega \\ \delta \omega &= \delta_L \omega + \delta_S \omega \\ \delta M &= \delta_L M + \delta_S M \end{aligned} \tag{10}$$

where the subscripts L and S indicate long- and short-period terms respectively. There are no long-period terms in a.

While long-period terms are explicitly written down as Fourier series whose arguments are linear functions of the time, this is not the case for the short-period terms where the true anomaly which is not linear in t is used in the arguments.

The long-period portions of the elements are indicated by primes and are defined to be the sums of the secular portions and the long-period perturbations. Using (8) and (9) and noting that there are no long-period terms in a we find

$$\begin{aligned}
 a' &= a'' & &= a_0 \\
 e' &= e'' + \delta_L e = e_0 & &+ \delta_L e \\
 I' &= I'' + \delta_L I = I_0 & &+ \delta_L I \\
 \Omega' &= \Omega'' + \delta_L \Omega = \Omega_0 + \Omega_{10} t + \delta_L \Omega \\
 \omega' &= \omega'' + \delta_L \omega = \omega_0 + \omega_{10} t + \delta_L \omega \\
 M' &= M'' + \delta_L M = M_0 + M_{10} t + \delta_L M
 \end{aligned} \tag{11}$$

The Brouwer expressions for the osculating elements are thus equal to the sums of the long-period portions and the short-period perturbations. Thus

$$\begin{aligned}
 a_B &= a_0 & &+ \delta_s a = a' + \delta_s a \\
 e_B &= e_0 & &+ \delta_L e + \delta_s e = e' + \delta_s e \\
 I_B &= I_0 & &+ \delta_L I + \delta_s I = I' + \delta_s I \\
 \Omega_B &= \Omega_0 + \Omega_{10} t + \delta_L \Omega + \delta_s \Omega = \Omega' + \delta_s \Omega \\
 \omega_B &= \omega_0 + \omega_{10} t + \delta_L \omega + \delta_s \omega = \omega' + \delta_s \omega \\
 M_B &= M_0 + M_{10} t + \delta_L M + \delta_s M = M' + \delta_s M
 \end{aligned} \tag{12}$$

Brouwer computes the short-period perturbations by making use of the von Zeipel method (von Zeipel 1916-1918). This method furnishes the short-period perturbations as functions of the long-period portions. However, Brouwer uses in these functions the secular portions e'' , I'' instead of the long-period portions e' , I' . Since he intended to find the short-period terms to order J_2 only and since his procedure causes an error only of order J_2^2 this is legitimate. We shall come back to this point on page 8.

It should be noted that Brouwer does not compute the periodic perturbations directly but computes first the periodic perturbations in the Delaunay variables

$$L, G, H, l, g, h$$

which are defined by

$$\begin{aligned} L &= (a \mu)^{1/2} \\ G &= L(1 - e^2)^{1/2} \\ H &= G \cos I \\ l &= M \\ g &= \omega \\ h &= \Omega \end{aligned} \tag{13}$$

3. OPTIONAL MODIFICATIONS OF THE BROUWER THEORY

The program input contains an option which allows the computation of M_{10} according to Kozai rather than according to Brouwer. This is accomplished by adding to the value according to Brouwer, which is of the second order, the difference δM_{10} of the value according to Kozai which is likewise used as far as to order J_2^2 minus the value according to Brouwer. Mathematical details concerning this are given in Appendix C (pp. 189-200).

There is a further option which states whether the long-period terms are to be included or not and another option whether the short-period perturbations are to be included or not. If they are to be included two further options will be available. In one of these the short-period terms are computed, as Brouwer does, with the secular portions e'' and I'' , instead of with the long-period portions e' and I' as required by the theory. The error (see page 7) is of the second order and is of no consequence if one limits oneself as Brouwer does to short-period terms of the first order. However, it may be desirable to use second-order short-period terms in the semi-major axis a . Then it is essential to have the short-period terms in a computed with e' and I' . The second option provides for the computation of the short-period terms with e' and I' in a and also e, I, Ω, ω, M .

The second order short-period terms in a are important since they have an appreciable effect on the secular motion of M as computed from the Brouwer mean elements corresponding to a given position-velocity vector using Brouwer's or Kozai's formulation for the mean motion of the mean anomaly.

The values of these short-period terms computed by the program are based on expressions given by Kozai (Kozai 1962) and may be optionally included in any run with the program being described here. Mathematical details will be found in Appendix C (pp. 189-200).

For the purpose of studying discrepancies with other orbit generators it will be useful to have the option of adding certain terms to the Brouwer expressions even though with the added terms the expressions may no longer correspond to a gravitational solution within the order specified by Brouwer.

The terms that may be added are linear, quadratic, and cubic terms in the time in the semi-major axis, eccentricity, and inclination. The Brouwer theory provides for linear terms in time in the right ascension of the node, the argument of perigee, and the mean anomaly. Any additional linear terms and quadratic and cubic terms may be added. Finally, up to 99 sines and cosines may be added to the six elements. The arguments of the sines and cosines are arbitrary linear functions of the time. The same arguments must be used for each of the six elements. The sum of the sines and cosines for each of the six elements will be referred to as the supplementary perturbation of this element.

The optional additions to the Brouwer theory discussed may be written in the form

$$\begin{aligned}
 \delta_a a &= a_1 t & + a_2 (0.01 t)^2 + a_3 (0.01 t)^3 & + \delta_2 a + \delta_{\text{sup}} a & \text{in } a \\
 \delta_a e &= e_1 t & + e_2 (0.01 t)^2 + e_3 (0.01 t)^3 & + \delta_{\text{sup}} e & \text{in } e \\
 \delta_a I &= I_1 t & + I_2 (0.01 t)^2 + I_3 (0.01 t)^3 & + \delta_{\text{sup}} I & \text{in } I \\
 \delta_a \Omega &= d\Omega_1 t & + \Omega_2 (0.01 t)^2 + \Omega_3 (0.01 t)^3 & + \delta_{\text{sup}} \Omega & \text{in } \Omega \\
 \delta_a \omega &= d\omega_1 t & + \omega_2 (0.01 t)^2 + \omega_3 (0.01 t)^3 & + \delta_{\text{sup}} \omega & \text{in } \omega \\
 \delta_a M &= (\delta M_{10} + dM_1) t + M_2 (0.01 t)^2 + M_3 (0.01 t)^3 & + \delta_{\text{sup}} M & \text{in } M
 \end{aligned} \tag{14}$$

The quantities

a_1 in megameters per hour

e_1 per hour

$I_1, d\Omega_1, d\omega_1, dM_1$ in degrees per hour

define the optional additional linear terms and

a_2 in megameters per (100 hours)²

e_2 per (100 hours)²

$I_2, \Omega_2, \omega_2, M_2$ in degrees per (100 hours)²

and

a_3 in megameters per (100 hours)³

e_3 per (100 hours)³

$I_3, \Omega_3, \omega_3, M_3$ in degrees per (100 hours)³

the optional quadratic and cubic terms. For scaling purposes 100 hours instead of 1 hour is the unit of time in these terms.

The quantity δM_{10} is the optional reduction of the mean motion from Brouwer to Kozai. The quantity $\delta_2 a$ is the sum of the optional second order short-period terms in a and

$$\delta_{\text{sup}} a, \delta_{\text{sup}} e, \delta_{\text{sup}} I, \delta_{\text{sup}} \Omega, \delta_{\text{sup}} \omega, \delta_{\text{sup}} M$$

are the optional supplementary perturbations.

4. THE BASIC ORBIT GENERATOR FUNCTION

The first basic orbit generator function consists in computing complete expressions for the osculating values of the elements

a = semi-major axis in megameters

e = eccentricity

I = inclination

Ω = right ascension of ascending node

ω = argument of perigee

M = mean anomaly

} with respect to
equator and equinox
} in degrees

at t hours from the epoch by means of the formulae

$$\begin{aligned}
 a &= a_B + \delta_a a \\
 e &= e_B + \delta_a e \\
 I &= I_B + \delta_a I \\
 \Omega &= \Omega_B + \delta_a \Omega \\
 \omega &= \omega_B + \delta_a \omega \\
 M &= M_B + \delta_a M
 \end{aligned}
 \tag{15}$$

Here the first terms of the right hand members are given by (12) and correspond to the Brouwer theory proper with the added option that the short-period terms may be computed with either e'' , I'' or e' , I' . The second terms of the right-hand members represent the optional additional terms given by (14). Their inclusion may cause the osculating values to be no longer consistent with a gravitational solution within the order specified by Brouwer.

To compute the osculating values of the elements for any t the following quantities are required:

Geophysical constants R = mean equatorial radius of the earth in megameters

GM_E = product of the gravitational constant and mass of the earth in km cubed/seconds squared

J_2, J_3, J_4, J_5 zonal harmonics of orders 2-5

Epoch (calendar date and time of day)

Orbital parameters

Mean Brouwer elements

a_0 semi-major axis in megameters

e_0 eccentricity

I_0 inclination

Ω_0 right ascension of ascending node

ω_0 argument of perigee

M_0 mean anomaly

$\left. \begin{array}{l} \text{with respect to} \\ \text{equator and equinox} \end{array} \right\} \text{ in degrees}$

Additional Secular Coefficients

Linear coefficients

rate a_1 in megameters per hour

rate e_1 per hour

motion I_1 , and additional motions $d\Omega_1, d\omega_1, dM_1$ in degrees per hour

Quadratic Coefficients

a_2 in megameters per (100 hours)²

e_2 per (100 hours)²

$I_2, \Omega_2, \omega_2, M_2$ in degrees per (100 hours)²

Cubic Coefficients

a_3 in megameters per (100 hours)³

e_3 per (100 hours)³

$I_3, \Omega_3, \omega_3, M_3$ in degrees per (100 hours)³

Further it must be indicated whether the reduction of the mean motion of the mean anomaly from Brouwer to Kozai, the long-period perturbations, the short-period perturbations, the second order short-period terms in a , and the supplementary perturbations are to be included. If supplementary perturbations are to be included each argument occurring in them is to be defined by giving its value in degrees at an epoch to be specified and its change in degrees per hour and the coefficients of the cosine and sine of this argument in each of the six elements. The epochs to be specified for the arguments may be different for each argument and need not coincide with the epoch of the Brouwer mean elements.

Finally, an indication must be given whether the short-period terms are to be computed with e'', I'' or e', I' .

If expressions based on the Brouwer theory proper are desired the additional secular coefficients are to be put equal to zero, i.e. one has to use

$$a_1 = e_1 = I_1 = d\Omega_1 = d\omega_1 = dM_1 = 0$$

$$a_2 = e_2 = I_2 = \Omega_2 = \omega_2 = M_2 = 0$$

$$a_3 = e_3 = I_3 = \Omega_3 = \omega_3 = M_3 = 0$$

and choose the options not to include the reduction of the mean motion of the mean anomaly from Brouwer to Kozai, to include the long- and short-period perturbations, to compute the latter with e'' , I'' , not to include the second order short-period terms in a , and not to include any supplementary perturbations.

The second basic orbit generator function consists in computing for t the rectangular equatorial coordinates

x, y, z in megameters

and the equatorial velocity components

$\dot{x}, \dot{y}, \dot{z}$ in megameters per hour

This process includes solving Kepler's equation for the eccentric anomaly.

For convenience sake the radius vector in megameters and the eccentric and true anomalies in degrees are included in the output.

The results are arranged in form of one or several ephemerides. For each ephemeris a starting and ending time and a step size must be specified. For each ephemeris point

$x, y, z, \dot{x}, \dot{y}, \dot{z}$

and the osculating values are written on an orbital tape. The format is described in Section 9. Information is also written out on the standard output tape at specified frequencies.

5. ALTERNATIVE OPERATIONS OF THE PROGRAM

The input necessary for the basic orbit generator functions has been described in Section 4. However, two alternatives are available.

Instead of the Brouwer mean elements the osculating values

a = semi-major axis in megameters

e = eccentricity

I = inclination

Ω = right ascension of the ascending node

ω = argument of perigee

M = mean anomaly

$\left. \begin{array}{l} \text{with respect to} \\ \text{equator and equinox} \end{array} \right\} \text{in degrees}$

at the epoch may be used in the input. All other quantities of the input must be used as described in Section 4. The program then determines by an iteration process those Brouwer mean elements which together with the remaining quantities of the input would produce the osculating values of the input. After these Brouwer mean elements have been determined the program operates as described in Section 4.

As a second alternative, instead of the Brouwer mean elements, the values of the rectangular equatorial coordinates

x, y, z in megameters

and of the rectangular equatorial velocity components

$\dot{x}, \dot{y}, \dot{z}$ in megameters per hour

at the epoch may be used in the input. All other quantities of the input must be used as described in Section 4. The program then determines first the values of the osculating elements $a, e, I, \Omega, \omega, M$ at the epoch and then, as in the preceding case, by an iterative process, those Brouwer mean elements which would produce these osculating values. After these have been found the program operates as described in Section 4.

6. UNITS

The unit of length used in the input described in Section 4 or the alternatives described in Section 5 is the megameter. The unit of time is the hour. Angles are given in degrees. Velocity components and coefficients of linear terms are given per hour and coefficients of the quadratic and cubic terms in t are given per $(100 \text{ hour})^2$ and $(100 \text{ hours})^3$ respectively.

Other orbit generator programs have used other units. An alternate choice of units is possible with the present program. So called canonical units may be used instead of the units described above. The canonical unit of length (cul) is the mean equatorial radius of the earth and the canonical unit of time (cut) is that unit which together with the canonical unit of length causes the constant μ in expressions (1) and (2) for the earth's potential to be unity. It is

$$R^{3/2} (GM)^{-1/2} \text{ seconds}$$

if R is measured in cm and GM_E in $\text{cm}^3 \text{ sec}^{-2}$.

Coefficients of the linear terms are then expressed per cut and coefficients of quadratic and cubic terms per (cut)² and (cut)³ respectively.

7. BRIEF DESCRIPTION OF THE PROGRAM

The main program is designated as P73. It requires the following sub-routines (SR) and function programs except for library routines. Subroutines are called and function programs indicated by * are merely used by name without the command CALL.

ADLH	BRWR4	KOMEAN	SSWTCH
ALLOT*	DJUL*	MAD*	SUPP0
ATANQ*	DMAD*	PARA	SUPP1
ATANZ*	ELRV	POLVAL	TIMC4
BBRWR	HMSRZ*	RHMSZ	WRT6
BRWR1	JULCAL	RVELZ	XKEP*
BRWR2			

The listings for P73 and the subroutines listed above are given in Appendix A (pp. 21-172). An index appears on p. 22.

All routines except for

ADLH, BRWR4, KOMEAN, POLVAL, SSWTCH, SUPP0, SUPP1, WRT6

were contained, possibly under a slightly different name, in the original Repass program package.

SR ADLH is a slight modification of a program ADDL described by Agreen and Fisher (1968). SR BRWR4 is a modification of BRWR2, and KOMEAN, POLVAL, SUPP0, SUPP1, and WRT6 have been designed by the author. SSWTCH is a routine simulating the sense switches of the IBM 7094 machines.

In the listings for P73, BRWR4, and all other routines originating from a routine of the Repass program package except in the case of DMAD and MAD, the following information or parts of it, often appear:

Purpose (Brief description of function of program or routine)
 Calling sequence to be used
 Input
 Output
 Reference
 Method
 Restrictions
 Accuracy
 Required subprograms
 Timing
 Analysis

This information appears also in the listing for BRWR4. Above information in many cases is, however, characterized as not available. The list of required subprograms refers only to first level subprograms, i.e. a program called by a subprogram of some routine is not listed.

In case of the routines where a list of required subprograms is not given the required subprograms may be ascertained from the compilation.

A brief description of the present program package follows.

The main program, P73, reads the input except for the ephemeris specifications which are read by TIMC4, and the specifications for the supplementary perturbations which are read by SUPP0 if required.

The action of PARA depends on which of the three alternative input options discussed in Sections 4 and 5 are used. If Brouwer mean elements are used in the input PARA determines the values of the osculating elements at the epoch by employing the basic orbit generator functions to be described in some detail below. If the values of the osculating elements at the epoch are used in the input no processing occurs in PARA. Finally, if the position-velocity vector at the epoch is used in the input then the values of the osculating elements at the epoch are determined by RVELZ. No further processing occurs.

Thus in every case, after completion of PARA, the values of the osculating elements at the epoch will be available. With the help of ELRV the position-velocity vector at the epoch is determined. In case the values of the osculating elements at epoch or the position-velocity vector at epoch, were used in the input, i.e. if the Brouwer mean elements were not used in the input, they are now determined. This will be done by BBRWR using an iterative process. If the Brouwer mean elements are used in the input then BBRWR is bypassed. In every case a point in the program is reached when the Brouwer mean elements are known and the basic orbit generator functions discussed in Section 4 may be carried out.

These functions start out with the Brouwer mean elements. The coefficients of the secular and long-period terms are computed in BRWR1. This need be done just once regardless how many ephemeris points are to be processed. If the reduction of the mean motion of the mean anomaly from Brouwer to Kozai is to be included it is found from KOMEAN.

For the epoch and every ephemeris point to be processed, the expressions for the long-period terms are evaluated and the short-period perturbations are computed. Then the complete values of the osculating elements are formed and the position-velocity vector is obtained. All these tasks are accomplished by BRWR2 or BRWR4 according to whether e'' , I'' or e' , I' are used in the computation of the short-period perturbations. If the second order short-period

terms in a are to be included a call in BRWR2 and BRWR4 to ADLH is necessary.* If supplementary perturbations are to be included then a call in BRWR2 and BRWR4 to SUPP1 is necessary. After the complete values of the osculating elements have been formed in BRWR2 or BRWR4 the position-velocity vector is determined with the help of ELRV. One step in this procedure consists in solving Kepler's equation which requires the use of XKEP.

The writing of the orbital tape is done in the main program P73 by WRT6.

Several routines and function programs have not yet been mentioned in this brief description. ALLOT reduces an angle modulo 2π . ATANZ and ATANQ find an angle from its sine and cosine or its sine and cosine multiplied by a constant respectively. DJUL and JULCAL are used in converting from calendar date to Julian day and vice versa. DMAD and MAD compute remainders from divisions. HSMRZ and RHMSZ convert from hours, minutes, seconds to radians and from radians to days, hours, minutes, seconds respectively. Finally, POLVAL computes polynomials.

8. DESCRIPTION OF INPUT DECK

The cards of the input deck are described in the listings in Appendix A of the programs where they are read. For convenience sake the relevant information alone is listed in Appendix B (pp. 173-187).

Several cases may be run. For each case there are cards numbered 1-9 with several cards for some numbers, in which case distinguishing letters and numbers are used. Some of the cards are omitted under special conditions. After the last case a number 1 and a number 2 card must follow. The latter must have a 0 in cols. 1-3.

In each case, one or several ephemerides may be computed. Certain data for every ephemeris point are written out on a tape called orbital tape, the format of which is described in Section 9.

The quantity KPR in columns 60-65 of card 9 indicates how often the data for ephemeris points are to be written on the standard output tape.

9. ORBITAL TAPES

The orbital tape is an EBCDIC tape containing double precision data for one or several orbits. Each orbit corresponds to one file on the tape.

* In view of the discussion on page 8 it would not be realistic to include the second order short-period terms in a and to compute the short-period terms with e'' , l'' .

Each file contains a lead record and orbital data for one or several sets of equidistant dates.

The available orbital data are coordinates, velocities, and osculating elements.

Format

Lead record: NCASE orbit number
CASE object + orbit number
(object number is the integral portion, orbit number
is the decimal portion)
NTIME = 0
NPLAN = 0
NOTHER = 0
JA(I), I = 1, . . . , 8 = 0
INPUT = 0
ORB1 = 0.D 00
ORB2 = 0.D 00

Format 1X, I3, F8.3, I5, 2I4, 8I3, I4, 2F8.3

For each ephemeris point there are three record

Record 1: time in minutes from epoch

$$\left. \begin{array}{l} x \\ y \\ z \end{array} \right\} \text{ in megameters}$$

$$\left. \begin{array}{l} \dot{x} \\ \dot{y} \\ \dot{z} \end{array} \right\} \text{ in megameters per hour}$$

Format 1X, I5, 6D21.14

Record 2: time in minutes from epoch

0.D 00
0.D 00
0.D 00

Format 1X, I5, 3D23.16

Record 3: time in minutes from epoch

a	osculating value of	semi-major axis in megameters	} with respect to equator and equinox	} in degrees
e		eccentricity		
I		inclination		
Ω		right ascension of ascending node		
ω		argument of perigee		
M		mean anomaly		

Format 1X, I5, 6D21.14

There are other programs which produce the same type of tape where the quantities which are zero in the lead record are not necessarily zero and where record 2 of the orbital data contains accelerations instead of zeros. A future version of this program will produce these accelerations on the orbital tape.

Orbital tapes may be input to other programs. One such programs forms the differences of orbital data from two orbits which are represented either by two files of one orbital tape or two files from two orbital tapes.

10. DESCRIPTION OF OUTPUT ON STANDARD OUTPUT

After printing the title page, the input options from card 2 of input deck are listed. A statement concerning units follows. With this information, the value of a constant in another orbit generator program, vis SATPOS (Walther and Wales, 1967) is included. After this information, a list of the values used for the required geophysical constants is given. If supplementary perturbations are to be included the arguments and coefficients occurring in these are listed. Then the input orbital parameters in the units used in the input are listed and repeated in the alternate units. Information concerning the tolerance and maximum number of iterations allowed in solving Kepler's equation follows.

A table showing the coefficients of the secular and long-period terms constitutes the next group of data. The basis of the calculation of the mean motion of the mean anomaly is given. The values of the Brouwer mean elements, osculating elements, and the components of the position-velocity vector at the epoch are then shown in several units.

After this ephemeris data are printed. Preceding the data for the individual data points information on the limits of the ephemeris and frequency of printing is given. For each ephemeris point, in general, sixteen lines of numerical data are given in the form of two blocks. The first contains twelve, the second four lines.

The first line of the first block contains the date in several forms and the time from the epoch in minutes. Lines 2-12 give information on various portions of the osculating elements. Line 2 gives the gravitational secular portions, i.e. the values of the secular portions as derived from the Brouwer theory proper with the possible effect of the optional reduction of the mean motion of the mean anomaly from Brouwer to Kozai included. The optional secular terms are given in lines 3-5 leading to the total secular portions on line 6. Line 7 gives the long-period perturbations leading to the long-period portions in line 8. The short-period perturbations, second order short-period perturbations, and supplementary perturbations follow in lines 9-11. The complete values of the osculating elements are then given in line 12, the last line of the first block.

The first line of the second block lists the radius vector, eccentric and true anomalies. The next two lines show the number of iterations required to solve Kepler's equation and the last relative correction obtained for the approximation to the solution. The final line of the block contains the values of the components of the position-velocity vector.

There are some options to be specified in the input which allow additional output to be printed, i.e. detailed output of some phases of the computations. These additional outputs are printed if certain fields of columns on card 2 of the input deck contain quantities not zero. If a zero is in one of these fields the corresponding output will not be printed.

If the quantity in cols. 19-21 of card 2 of the input deck is not zero intermediate output from BRWR1 will be printed. It is not labelled. For identification the program listing of BRWR1 must be consulted.

If the quantity in cols. 43-45 of card 2 of the input deck is not zero intermediate output from KOMEAN will be printed. It is not labelled. For identification the program listing of KOMEAN must be consulted.

If the second order short-period terms in a are desired and if the quantity in cols. 31-33 of card 2 of the input deck is not zero details of the computation of the second order short-period terms in a are given for all ephemeris points for which data are given in the standard output. There is little labelling in this detailed output, which appears between lines 1 and 2 of the first block of twelve lines referred to above. Identification should be made with the help of the program listings of ADLH, BRWR2, or BRWR4.

Finally, if supplementary perturbations are desired and if the quantity in cols. 37-39 of card 2 of the input deck is not zero intermediate quantities in the computation of the supplementary perturbations are printed for every ephemeris point for which data are printed in the standard output. This additional output is labelled and appears also between lines 1 and 2 of the first block of twelve lines referred to above.

APPENDIX A
PROGRAM LISTINGS

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ISN 0006	7	DUM2(85),DUMF(9)	00045680
	8	AG(6),D11(6),D2(6),D3(6)	00045690
	9	AGPR(6),CIBR(6),D2BR(6),D3BR(6),ACC(3),JA(12)	00045700
		UI HENSION TOR(99),AFG0(99),ARGMOT(99),ARGC(99),CCOEF(99,6)	00045710
ISN 0007		ISCDEF(99,6),DSUP(6),ASHORT(6),ASBMC(3),VXBMC(3),DADIM(6)	00045720
ISN 0009		DIMENSION KSM(6)	00045730
	1000	CONTINUE	00045740
		COMMON DUM1,A110,GP,ERR,XX,AB,RXB,VXR,XXX,SS,EL,DUM2,DUMF,	00045750
		1CONV,AIDM,AG,D1,D2,D3,CMU,T0,DJ0,DTIMES,ARGO,ARGMOT,ARG,CCOEF,	00045760
		2SCDEF,DSUP,CJ2S,CJ3,CJ4,DLJ2S,DLJ3,DLJ4,FLSUM,DAJ2S,DAJ3,DAJ4,	00045770
		3CASUM,FEM,ASHORT,	00045780
	4	NERWRL,ALONG,NSHGT,NSEC,NSECD,NSUPP,NSUPPD,NARG,KCOUNT	00045790
	5	NKOZ,NKOZL	00045800
ISN 0010		COMMON /LAB1/ REML,TUH	00045810
		DEFINITION OF SYMBOLS	00045820
	C	ERR = TRUNCATION FACTOR (IN RADIANS) USED IN SOLUTION OF	00045830
	C	KEPLERS EQUATION	00045840
	C	GM = PRODUCT OF G (=GAUSSIAN CONSTANT SQUARED) AND M, THE MASS OF	00045850
	C	THE EARTH, IN UNITS OF KM. CUBED/SEC SQUARED	00045860
	C	FJ2=J2)	00045870
	C	FJ3=J3) HARMONICS OF EARTHS GRAVITATIONAL POTENTIAL	00045880
	C	FJ4=J4) (DIMENSIONLESS)	00045890
	C	FJ5=J5)	00045900
	C	1003 CONTINUE	00045910
ISN 0011	C	RE= EQUATORIAL RADIUS OF EARTH IN KM.	00045920
	C		00045930
	C		00045940
	C		00045950
	C		00045960
	C		00045970
	C	ALL FORMATS USED IN PROGRAM FOLLOW IMMEDIATELY.	00045980
	C		00045990
ISN 0012	1800	FORMAT(1H1)	00046000
ISN 0013	1801	FORMAT(1X,19HOSCULATING ELEMENTS,15X,F15.7,F17.9,F11.5,3F15.5)	00046010
ISN 0014	1802	FORMAT(1X,16HSECULAR PORTIONS,15X,F15.7,F17.9,F11.5,3F15.5)	00046020
ISN 0015	1803	FORMAT(1X,25HLONG PERIOD PERTURBATIONS,9X,F15.7,F17.9,F11.5,3F15.5,50046030	00046040
	1)		00046050
ISN 0016	1804	FORMAT(1X,20HLONG PERIOD PORTIONS,14X,F15.7,F17.9,F11.5,3F15.5)	00046060
ISN 0017	1805	FORMAT(1X,2CHX,Y,Z,XDOT,YDOT,ZDOT,14X,6F15.7)	00046070
ISN 0018	1806	FORMAT(1X,26HSHORT PERIOD PERTURBATIONS, 8X,F15.7,F17.9,F11.5,	00046080
	13F15.5)		00046090
ISN 0019	1808	FORMAT(94H A E I	00046100
	1	NODE OMEGA M)	00046110
ISN 0020	1810	FORMAT(//91HCOEFFICIENTS OF LONG PERIOD PERTURBATIONS (ARGUMENTS	00046120
	1ARE N TIMES SECULAR PORTION OF OMEGA)/98HON A		00046130
	2 E	1 NODE OMEGA M	00046140
	3	PI/1H0.56X,6HCUSINE/4H01,2F14.8,3F14.6,2F16.6/4H 2,2F14.8,3F14.6,2F16.6/4H 3,2F14.8,3F14.6,2F16.6/	00046150
ISN 0021	1811	FORMAT(1H0.57X,4HCUSINE/4H01,2F14.8,3F14.6,2F16.6/	00046160
	114.6,2F16.6/4H 3,2F14.8,3F14.6,2F16.6/		00046170
ISN 0022	1816	FORMAT(53X,26HIN EARTH RADII AND RADIANS/)	00046180
ISN 0023	1817	FORMAT(//53X,25HIN MEGAMETERS AND DEGREES/)	00046190
ISN 0024	1318	FORMAT(//53X,25HIN KM AND KM PER SEC/)	00046200
ISN 0025	1819	FORMAT(59X,13HNEAN ELEMENTS//)	00046210
ISN 0026	1820	FORMAT(//56X,19HOSCULATING ELEMENTS//)	00046220
ISN 0027	1821	FORMAT(61X, 9HRV VECTOR//)	00046230
ISN 0028	1822	FORMAT(1X,14HCANONICAL UNIT,3X,C24.16,3X,7HSECONDS,5X,D24.16,3X,	

```

154HOURS)
1823 FORMAT(1X,1HK,3X,D24.16)
1825 FORMAT(//48X,37HIN MEGAMETERS AND MEGAMETERS PER HOUR/)
1827 FORMAT(1X,38HCONSTANT CONI USED IN SATPOS PROGRAM =,5X,D24.16,/)
1830 FORMAT(72X)
1831 FORMAT(1H1,54X,24HINPUT ORBITAL PARAMETERS////)
1832 FORMAT(1X,43HVV-VECTOR IN MEGAMETERS,MEGAMETERS PER HOUR,6X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1833 FORMAT(1X,45HVV-VECTOR IN EARTH RADII,EARTH RADII PER CUT,.4X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1834 FORMAT(1X,39HMEAN ELEMENTS IN MEGAMETERS AND DEGREES,10X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1835 FORMAT(1X,40HMEAN ELEMENTS IN EARTH RADII AND RADIAN, 9X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1836 FORMAT(1X,45HOSCILLATING ELEMENTS IN MEGAMETERS AND DEGREES,4X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1837 FORMAT(1X,46HOSCILLATING ELEMENTS IN EARTH RADII AND RADIAN,3X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1838 FORMAT(1X,48HCOEFF. OF ADD'L LIN. TERMS OF ELEMENTS, PER HR,.1X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1839 FORMAT(1X,48HCOEFF. OF ADD'L LIN. TERMS OF ELEMENTS PER CUT,.1X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1840 FORMAT(1X, 47HCOEFF.OF QUADR. TERMS PER (100 HRS) SQUARED,2X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1841 FORMAT(1X, 47HCOEFF.OF QUADR. TERMS PER (CUT ) SQUARED,2X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1842 FORMAT(1X, 47HCOEFF.OF CUBIC TERMS PER (100 HRS) CUBED ,2X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1843 FORMAT(1X, 47HCOEFF.OF CUBIC TERMS PER (CUT ) CUBED ,2X,
13(D24.17,3X)/50X,3(D24.17,3X)/)
1844 FORMAT(1X,20HGRAV.SECULAR PORTION, 14X,F15.7,F17.9,F11.5,3F15.5)00046540
1845 FORMAT(1X,23HADDTL.TERMS LINEAR IN T,11X,F15.7,F17.9,F11.5,3F15.5)00046550
1846 FORMAT(1X,20HTERMS QUADRATIC IN T, 14X,F15.7,F17.9,F11.5,3F15.5)00046560
1847 FORMAT(1X,20HTERMS CUBIC IN T, 14X,F15.7,F17.9,F11.5,3F15.5)00046570
1848 FORMAT(1X,30H2ND ORDER SHORT-PER.TERMS IN A, 4X,F15.7) 00046580
1849 FORMAT(1X,27H5UPPLEMENTARY PERTURBATIONS,7X,F15.7,F17.9,F11.5,
13F15.5) 00046590
1851 FORMAT(14X,15I7)
1852 FORMAT(59X,13HINPUT OPTIONS////14X,
1 INPUT NCST NERR NDA INC NUNIT NBRWRI NLONG NSHORT 10SH 00046610
2SEC NSECD NSUPP NSUPPD NKDZ NKDZC//) 00046620
1854 FORMAT(1H1,55X,21HGEOPHYSICAL CONSTANTS//) 00046640
1855 FORMAT(////26X, 80HADDITIONAL TERMS LINEAR IN T,QUADRATIC AND CUBIC00046650
IC TERMS IN T NOT LISTED FOR EPOCH/52X,19H5INCE THEY ARE ZERO////)00046660
1857 FORMAT(37X, 58HALL LENGTHS IN MEGAMETERS,TIMES IN HOURS,ANGLES IN 00046680
1DEGREES/55X,22HUNLESS OTHERWISE NOTED/51X,31HALL TIMES ARE IN EPHE00046690
2MERIS TIME) 00046700
1858 FORMAT(1H1,48X,34HINPUT CONVERTED TO ALTERNATE UNITS////) 00046710
1859 FORMAT( 38X,1HA,26X,1HE,26X,1H1/ 00046720
136X,4HNDGE,23X,5HMEGA,24X,1HM////) 00046730
1860 FORMAT(27X,3(D24.17,3X)) 00046740
1861 FORMAT(////36X,1HX,26X,1HY,26X,1HZ/ 00046750
136X,5HX DOT,22X,5HY DOT,22X,5HZ DOT////) 00046760
1862 FORMAT(47X,38HIN EARTH RADII AND EARTH RADII PER CUT//) 00046770
1863 FORMAT(////40X,52HTOLERANCES USED IN DETERMINING BROWNER MEAN ELEM00046780
ENTS////) 00046790
00046240
00046250
00046260
00046270
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00046690
00046700
00046710
00046720
00046730
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00046760
00046770
00046780
00046790

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ISN 0063	1864	FORMAT(///47X,29NUMBER OF ITERATIONS REQUIRED,3X,16)	00046800
ISN 0064	1865	FORMAT(8X,6(1010.4,6X))	00046810
ISN 0065	1866	FORMAT(///44X,36HMAXIMUM NUMBER OF ITERATIONS ALLOWED,3X,16)	00046820
ISN 0066	1867	FORMAT(11X,33HNO. OF IT. IN SOLV. KEPLER'S EQ'N,18)	00046830
ISN 0067	1868	FORMAT(11X,38LAST REL. CORR. IN SOLV. KEPLER'S EQ'N,10.2)	00046840
ISN 0068	1869	FORMAT(11X,27X,52HTOLERANCE USED IN SOLV. KEPLER'S EQ'N FOR (E2-E1)00046850 1)/52.3X. D8.2.3X /44X,36HMAXIMUM NUMBER OF ITERATIONS ALLOWED,3X,00046860 215)	00046870
ISN 0069	1870	FORMAT(12X,5HNZERO,20X,7HNOUE 10,17X,8HOMEGA 10,21X,4HMM 10,22X,5HPO00046880 11 10)	00046890
ISN 0070	1871	FORMAT(1X,5(1024.17,2X1)///)	00046900
ISN 0071	1872	FORMAT(62X,7HMOVATIONS//)	00046910
ISN 0072	1873	FORMAT(62X,1MA,26X,1ME,26X,1H1/ 160X,4HNOUE,23X,5HOMEGA,24X,1HM///)	00046920
ISN 0073	1874	FORMAT(40X,53HMOVATIONS AND COEFFICIENTS OF LONG PERIOD PERTURBATIONS,00046940 1S)	00046950
ISN 0074	1875	FORMAT(1H1,38X,56HMEAN ELEMENTS,OSCULATING ELEMENTS,AND RV VECTOR 1AT EPOCH)	00046960
ISN 0075	1876	FORMAT(1H1,57X,18DATA FOR THE EPOCH)	00046970
ISN 0076	1877	FORMAT(1H1,57X,9HEPHEMERIS,3X,16/////)	00046980
ISN 0077	1878	FORMAT(///26X,80HE1 AND E2 ARE TWO SUCCESSIVE APPROXIMATIONS TO THE00047000 1E SOLUTION OF KEPLER'S EQUATION)	00047010
ISN 0078	1879	FORMAT(11X,27HRAU,VECT,ECC,AND TRUE AN.,7X,F15,7.28X,F15,5.15X, 1F15.5)	00047020
ISN 0079	1880	FORMAT(///25X,46HMO EXPRESS LENGTHS IN CANONICAL UNIT DIVIDE BY, 00047040 13X,1024.17/26X,52HMO EXPRESS VELOCITIES IN CANONICAL UNITS MULTIPLY00047050 2 BY,3X,D24.17)	00047060
ISN 0080	1881	FORMAT(40X,18HANOALISTIC PERIOD,3X,D24.17,3X,5HMOURS///)	00047070
ISN 0081	1882	FORMAT(///56X,SHOREIT,3X,F8.3///)	00047080
ISN 0082	1883	FORMAT(///22X,87HTHE SHORT PERIOD TERMS ARE COMPUTED WITH E' AND 1" AS SPECIFIED BY BROUWER,AJ,NOV. 1959)	100047090
ISN 0083	1884	FORMAT(///49X,34HNO SHORT PERIOD TERMS ARE INCLUDED)	00047100
ISN 0084	1885	FORMAT(///9X,113HTHE SHORT PERIOD TERMS ARE COMPUTED WITH E' AND 00047120 11' INSTEAD OF WITH E' AND 1" AS SPECIFIED BY BROUWER,AJ,NOV. 1959)00047130	
ISN 0085	1886	FORMAT(18X,6HNUMBER,3X,16,3X,77HOF ARGUMENTS IN SUPPLEMENTARY PER-00047140 1TUBATIONS IS NOT A POSITIVE INTEGER LE,99.)	00047150
ISN 0086	1887	FORMAT(20X,65HREDUCTION OF MEAN MOTION OF MEAN ANOMALY FROM BROUWER00047160 1R TO KZAI = .3X,D24.17//)	00047170
ISN 0087	1888	FORMAT(42X,49HMEAN MOTION OF MEAN ANOMALY COMPUTED ACCORDING TO) 00047180	
ISN 0088	1889	FORMAT(62X, 7H8ROUWER//)	00047190
ISN 0089	1890	FORMAT(33X,66HKOZAI BY APPLYING REDUCTION TO VALUE COMPUTED ACCORD00047200 1ING TO BROUWER//)	00047210
ISN 0090	1899	FORMAT(1X,13,F8.3,15,214.813, 14,2F8.3)	00047220
ISN 0091	6002	FORMAT(12A61	00047230
ISN 0092	6003	FORMAT(1513)	00047240
ISN 0093	6004	FORMAT(1012.6,4012.5/ D24.17)	00047250
ISN 0094	6005	FORMAT (//1X0PF8.3,1X34H E00047260 1QUATORIAL RADIUS OF EARTH IN KM/ 2 1X,1PD14.6,3X31H GM (KM, CUBED/SECONDS SQUARED)///18X45H HARMON00047280 3ICS OF EARTHS GRAVITATIONAL POTENTIAL/1XD13.6,4X3H J2/1X,D13.6,4X 43H J3/1X,D13.6,4X3H J4/1X,D13.6,4X3H J5) 6006 FORMAT(13,F8.3,12,1X,F4.2 /3024.17/3024.17/3024.17/00047300 13024.17/3024.17/3024.17/3024.17/3024.17) 1004 CONTINUE 00047320 00047330 00047340 00047350	
ISN 0095			
ISN 0096			
ISN 0097	6010	FORMAT(///33X,4HJD =,3X,D24.17,5X,7H EPOCH ,16,4(1X,12),1X, 1F6.0///)	

C LUTION OF KEPLER'S EQUATION FROM 00048480
 C CARD 3. 00048490
 C ANY NEGATIVE INTEGER = OMIT CARD 3. TOLERANCE FOR EC- 00048500
 C OR 0 00048510
 C CENTRIC ANOMALY IS .100-13 AND 00048520
 C THE MAXIMUM NUMBER OF ITERATIONS 00048530
 C IS 50. 00048540
 C COLUMNS 10-12 BROWER TRUNCATION INDICATOR IN SUBROUTINE BBWR 00048550
 C (NDA). 00048560
 C ANY POSITIVE = REAL TOLERANCES TO BE USED IN DETERMINING MEAN 00048570
 C INTEGER BROWER ELEMENTS FROM CARD 6. 00048580
 C ANY NEGATIVE 00048590
 C INTEGER OR 0 = OMIT CARD 6. THE FOLLOWING TOLERANCES 00048600
 C ARE USED IN DETERMINING BROWER MEAN 00048610
 C ELEMENTS. 00048620
 C SEMIMAJOR AXIS = .500-10 KM 00048630
 C ECCENTRICITY = .500-14 00048640
 C INCLINATION = .500-11 DEGREES 00048650
 C RIGHT ASCENSION OF ASCENDING NODE = .500-11 DEGREES 00048660
 C ARGUMENT OF PERIGEE = .500-11 DEGREES 00048670
 C MEAN ANOMALY = .500-11 DEGREES 00048680
 C 00048690
 C 7001 CONTINUE 00048700
 C COLUMNS 13-15 MAXIMUM NUMBER OF ITERATIONS ALLOWED IN DETERMINING 00048710
 C BROWER MEAN ELEMENTS(IRC). 00048720
 C COLUMNS 16-18 TYPE OF UNITS USED ON CARDS 8A-8H(UNIT). 00048730
 C 0 = UNIT OF LENGTH = MEGAMETERS.UNIT OF TIME = HOUR. 00048740
 C ANGLES IN DEGREES 00048750
 C COEFFICIENTS OF LINEAR TERMS ARE GIVEN PER HOUR 00048760
 C COEFFICIENTS OF QUADRATIC TERMS ARE GIVEN PER 00048770
 C (100 HOURS) SQUARED 00048780
 C COEFFICIENTS OF CUBIC TERMS ARE GIVEN PER 00048790
 C (100 HOURS) CUBED. 00048800
 C OTHER THAN 0 = UNIT OF LENGTH = EARTH'S RADIUS.UNIT OF 00048810
 C TIME = CUT. ANGLES IN RADIANS. 00048820
 C (VANGUARD UNITS) 00048830
 C COEFFICIENTS OF LINEAR TERMS ARE GIVEN PER CUT 00048840
 C COEFFICIENTS OF QUADRATIC TERMS ARE GIVEN PER 00048850
 C (CUT) SQUARED. 00048860
 C COEFFICIENTS OF CUBIC TERMS ARE GIVEN PER 00048870
 C (CUT) CUBED. 00048880
 C COLUMNS 19-21 INTERMEDIATE OUTPUT FROM SUBROUTINE BBWR1 INDICATOR 00048890
 C (NBR1). 00048900
 C 0 = DO NOT PRINT INTERMEDIATE OUTPUT. 00048910
 C OTHER THAN 0 = PRINT INTERMEDIATE OUTPUT. 00048920
 C COLUMNS 22-24 LONG PERIOD PERTURBATIONS INDICATOR (MLONG) 00048930
 C 0 = DO NOT INCLUDE LONG PERIOD PERTURBATIONS 00048940
 C OTHER THAN 0 = INCLUDE LONG PERIOD PERTURBATIONS 00048950
 C COLUMNS 25-27 SHORT PERIOD PERTURBATIONS INDICATOR (NSHORT) 00048960
 C 0 = DO NOT INCLUDE SHORT PERIOD PERTURBATIONS 00048970
 C ANY NEGATIVE INTEGER = THE SHORT PERIOD PERTURBATIONS ARE COM- 00048980
 C PUTED WITH E' AND I'. 00048990
 C ANY POSITIVE INTEGER = THE SHORT PERIOD PERTURBATIONS ARE COM- 00049000
 C PUTED WITH E' AND I'. 00049010
 C 7002 CONTINUE 00049020
 C COLUMNS 28-30 SECOND ORDER SHORT-PERIOD TERMS IN SEMIMAJOR AXIS 00049030

15N 0145

15N 0146

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C          INDICATOR (NSEC)                                00049040
C          0 = DO NOT INCLUDE SECOND ORDER SHORT-PERIOD TERMS. 00049050
C          OTHER THAN 0 = INCLUDE SECOND ORDER SHORT-PERIOD TERMS. 00049060
C          COLUMNS 31-33 INTERMEDIATE OUTPUT OF SECOND ORDER SHORT-PERIOD 00049070
C          PERTURBATIONS IN SEMI-MAJOR AXIS INDICATOR (NSEC) 00049080
C          0 = DO NOT PRINT INTERMEDIATE OUTPUT. 00049090
C          OTHER THAN 0 = PRINT INTERMEDIATE OUTPUT. 00049100
C          NOTE: INTERMEDIATE DATA OUTPUT AT EPOCH WILL NOT BE PRINTED 00049110
C          UNLESS EPOCH DATE IS INCLUDED AMONG THE DATES IN REQUESTED 00049120
C          EPOCHES. 00049130
C          COLUMNS 34-36 SUPPLEMENTARY PERTURBATIONS INDICATOR (NSUPP) 00049140
C          0 = DO NOT INCLUDE SUPPLEMENTARY PERTURBATIONS 00049150
C          OTHER THAN 0 = INCLUDE SUPPLEMENTARY PERTURBATIONS. 00049160
C          COLUMNS 37-39 INTERMEDIATE OUTPUT OF SUPPLEMENTARY PERTURBATIONS 00049170
C          INDICATOR (NSUPPD) 00049180
C          0 = DO NOT PRINT INTERMEDIATE OUTPUT. 00049190
C          OTHER THAN 0 = PRINT INTERMEDIATE OUTPUT. 00049200
C          NOTE: INTERMEDIATE DATA OUTPUT AT EPOCH WILL NOT BE PRINTED 00049210
C          UNLESS EPOCH DATE IS INCLUDED AMONG THE DATES IN REQUESTED 00049220
C          EPOCHES. 00049230
C          COLUMNS 40-42 MEAN MOTION INDICATOR (NKOZ) 00049240
C          0 = USE BROUWER'S MEAN MOTIONS. 00049250
C          OTHER THAN 0 = USE KZAI'S MEAN MOTIONS. 00049260
C          COLUMNS 43-45 INTERMEDIATE OUTPUT OF COMPUTATION OF KOZAI MEAN 00049270
C          MOTIONS INDICATOR (NKOZD) 00049280
C          0 = DO NOT PRINT INTERMEDIATE OUTPUT. 00049290
C          OTHER THAN 0 = PRINT INTERMEDIATE OUTPUT. 00049300
C          READ 6003, INPUT, NCST, NERR, NDA, IRC, NUNIT, NBRWR1, 00049310
C          1, NLONG, NSHORT, NSEC, NSECD, NSUPP, NSUPPD, NKOZ, NKOZD 00049320
C          WRITE(6, 1800) 00049330
C          WRITE(6, 1852) 00049340
C          1, NLONG, NSHORT, NSEC, NSECD, NSUPP, NSUPPD, NKOZ, NKOZD 00049350
C          NSECE=NSECO NCST, NERR, NDA, IRC, NUNIT, NBRWR1 00049360
C          NSUPPE=NSUPPD 00049370
C          NSECD=0 00049380
C          NSUPPD=0 00049390
C          503* IF (INPUT) 300, 300, 2 00049400
C          IF INPUT=+00 OR -XX, ALL CASES HAVE BEEN RUN. GO TO 300 (END 00049410
C          TEST = CHANGE TRUNCATION FACTOR - YES OR NO 00049420
C          2 IF (NERR) 4, 3 00049430
C          CARD NO. 3 TOLERANCE AND MAXIMUM NUMBER OF ITERATIONS ALLOWED IN 00049440
C          SOLVING KEPLER'S EQUATION. 00049450
C          TOLERANCE IS UPPER LIMIT OF (E2-E1)/E2, WHERE E1 AND 00049460
C          E2 ARE VALUES OF THE ECCENTRIC ANOMALY IN TWO SUCCESSIVE 00049470
C          ITERATIONS. 00049480
C          NOTE: THIS CARD IS REQUIRED IF AND ONLY IF COLUMNS 7-9 00049490
C          OF CARD 2 (RUN CONTROL CARD) CONTAIN A POSITIVE INTEGER. 00049500
C          COLUMNS 1- 8 ECCENTRIC ANOMALY TOLERANCE IN FORMAT D8.2. 00049510
C          COLUMNS 9-12 MAXIMUM NO OF ITERATIONS ALLOWED IN SOLUTION OF KEP- 00049520
C          LER'S EQUATION. 00049530
C          00049540
C          00049550
C          00049560
C          00049570
C          00049580
C          00049590

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ISN 0157	C	3 READ 0040.ERR.NMAX CARDS NO. 4 AND 5 EARTH CONSTANTS CARDS	00049600 00049610 00049620 00049630 00049640 00049650 00049660 00049670 00049680 00049690 00049700 00049710 00049720 00049730 00049740 00049750 00049760 00049770 00049780 00049790 00049800 00049810 00049820 00049830 00049840 00049850 00049860 00049870 00049880 00049890 00049900 00049910 00049920 00049930 00049940 00049950 00049960 00049970 00049980 00049990 00050000 00050010 00050020 00050030 00050040 00050050 00050060 00050070 00050080 00050090 00050100 00050110 00050120 00050130 00050140 00050150
	C	NOTE: THESE CARDS ARE REQUIRED IF AND ONLY IF COLUMNS 4-6 OF CARD NO. 2 (THE RUN CONTROL CARD) CONTAIN A POSITIVE INTEGER.	
	C	FOR CARD 4	
	C	COLUMNS 1-12 CONSTANT OF ATTRACTION IN (KM CUBED / SEC SQUARED)	
	C	COLUMNS 13-24 J2)	
	C	COLUMNS 25-36 J3) COEFFICIENTS OF ZONAL HARMONICS	
	C	COLUMNS 37-48 J4) OF DEGREES 2-5	
	C	COLUMNS 49-60 J5)	
	C	FORMAT IS D12.6.4D12.5	
	C	FOR CARD 5	
	C	COLUMNS 1-24 MEAN EQUATORIAL RADIUS OF THE EARTH IN MEGAMETERS	
	C	FORMAT IS D24.17	
ISN 0158	C	4 IF (NCST)5.11.10	
	C	TO NEXT STEP.	
ISN 0159	C	5 IF (NCST+1)7.6.7	
	C	USE SRY PACKAGE CONSTANTS IF NCST = -1	
ISN 0160	C	GM=3.9862688D+05	
ISN 0161	C	FJ2=1.08219D-03	
ISN 0162	C	FJ3=-2.285C-06	
ISN 0163	C	FJ4=-2.123D-06	
ISN 0164	C	FJ5=-2.32D-07	
ISN 0165	C	RE=6378.368D0	
ISN 0166	C	GO TO 11	
ISN 0167	C	7 IF (NCST+2)9.8.11	
	C	USE GODDARD EARTH CONSTANTS WITH HARMONICS = 0 IF NCST = -2	
ISN 0168	C	FJ2=0.0D0	
ISN 0169	C	FJ3=0.0D0	
ISN 0170	C	FJ4=0.0D0	
ISN 0171	C	FJ5=0.0D0	
ISN 0172	C	GO TO 11	
ISN 0173	C	USE INTERNATIONAL CONSTANTS WITH HARMONICS = 0 IF NCST = -3	
ISN 0174	C	GM=3.98626873D+5	
ISN 0175	C	FJ2=0.0D0	
ISN 0176	C	FJ3=0.0D0	
ISN 0177	C	FJ4=0.0D0	
ISN 0178	C	FJ5=0.0D0	
ISN 0179	C	RE=6378.368D0	
	C	GO TO 11	

ISN 0214
 ISN 0215
 ISN 0216
 ISN 0217
 ISN 0218

DA(2)=EAD(2)
 DO 916 N=3,6
 916 DAIN=EAD(N)*0.01745329200
 IF (IRC)917,917,918
 917 IRC=50

C CARD NO. 7 ORBIT-OBJECT AND EPOCH CARD
 C COLUMNS 1-3 ORBIT NUMBER
 C COLUMNS 4-11 OBJECT + ORBIT NUMBER (ORBIT NUMBER IS DECIMAL PORTION TO 0.001)
 C COLUMNS 12-13 LAST TWO DIGITS OF YEAR
 C COLUMNS 14-15 MONTH
 C COLUMNS 16-17 DAY
 C COLUMN 18 BLANK
 C COLUMNS 19-20 HOUR
 C COLUMNS 21-22 MINUTES
 C COLUMNS 23 BLANK
 C COLUMNS 24-27 SECONDS TO HUNDRETHS OF SECONDS. NO DECIMAL POINT.

C CARDS NO. 8A THROUGH 8H (8 CARDS)
 C UNITS ARE DEFINED BY QUANTITY IN COLUMNS 16-18 OF CARD NO.2 (THE RUN CONTROL CARD).
 C THE FORMAT OF EACH OF CARDS 8A-8H IS 3D24.17.

A = SEMIMAJOR AXIS
 E = ECCENTRICITY
 I = INCLINATION TO THE EQUATOR
 NODE = RIGHT ASCENSION OF ASCENDING NODE
 OMEGA = ARGUMENT OF PERIGEE
 M = MEAN ANOMALY

C CARD 8A (COLUMNS 1-3 OF CARD 2 CONTAIN EITHER +01 OR +04)
 C COLUMNS 1-24 OSCULATING (+01) OR BROWER MEAN (+04) A
 C COLUMNS 25-48 OSCULATING (+01) OR BROWER MEAN (+04) E
 C COLUMNS 49-72 OSCULATING (+01) OR BROWER MEAN (+04) I
 C CARD 8B (COLUMNS 1-3 OF CARD 2 CONTAIN EITHER +01 OR +04)
 C COLUMNS 1-24 OSCULATING (+01) OR BROWER MEAN (+04) NODE
 C COLUMNS 25-48 OSCULATING (+01) OR BROWER MEAN (+04) OMEGA
 C COLUMNS 49-72 OSCULATING (+01) OR BROWER MEAN (+04) M
 C NOTE: THESE COMPONENTS ARE AT EPOCH REFERRED TO THE EQUATOR AND EQUINOX.

C CARD 8A (COLUMNS 1-3 OF CARD 2 CONTAIN +03)
 C COLUMNS 1-24 X COMPONENT OF POSITION VECTOR
 C COLUMNS 25-48 Y COMPONENT OF POSITION VECTOR
 C COLUMNS 49-72 Z COMPONENT OF POSITION VECTOR

C 1011 CONTINUE
 C CARD 8B (COLUMNS 1-3 OF CARD 2 CONTAIN +03)
 C COLUMNS 1-24 X COMPONENT OF VELOCITY VECTOR
 C 25-48 Y COMPONENT OF VELOCITY VECTOR
 C 49-72 Z COMPONENT OF VELOCITY VECTOR
 C NOTE: THESE COMPONENTS ARE AT EPOCH REFERRED TO THE EQUATOR

ISN 0219

00050720
 00050730
 00050740
 00050750
 00050760
 00050770
 00050780
 00050790
 00050800
 00050810
 00050820
 00050830
 00050840
 00050850
 00050860
 00050870
 00050880
 00050890
 00050900
 00050910
 00050920
 00050930
 00050940
 00050950
 00050960
 00050970
 00050980
 00050990
 00051000
 00051010
 00051020
 00051030
 00051040
 00051050
 00051060
 00051070
 00051080
 00051090
 00051100
 00051110
 00051120
 00051130
 00051140
 00051150
 00051160
 00051170
 00051180
 00051190
 00051200
 00051210
 00051220
 00051230
 00051240
 00051250
 00051260
 00051270

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ISN 0220
ISN 0221
ISN 0222
ISN 0223
ISN 0224
ISN 0225
ISN 0226

AND EQUINOX.

FOR ALL VALUES IN COLUMNS 1-3 OF CARD 2:
CARD 8C
COLUMNS 1-24 COEFFICIENT OF LINEAR TERM IN A
COLUMNS 25-48 COEFFICIENT OF LINEAR TERM IN E
COLUMNS 49-72 COEFFICIENT OF LINEAR TERM IN I

CARD 8D
COLUMNS 1-24 COEFFICIENT OF ADDITIONAL LINEAR TERM IN NODE
COLUMNS 25-48 COEFFICIENT OF ADDITIONAL LINEAR TERM IN OMEGA
COLUMNS 49-72 COEFFICIENT OF ADDITIONAL LINEAR TERM IN M

CARD 8E
COLUMNS 1-24 COEFFICIENT OF QUADRATIC TERM IN A
COLUMNS 25-48 COEFFICIENT OF QUADRATIC TERM IN E
COLUMNS 49-72 COEFFICIENT OF QUADRATIC TERM IN I

CARD 8F
COLUMNS 1-24 COEFFICIENT OF QUADRATIC TERM IN NODE
COLUMNS 25-48 COEFFICIENT OF QUADRATIC TERM IN OMEGA
COLUMNS 49-72 COEFFICIENT OF QUADRATIC TERM IN M

CARD 8G
COLUMNS 1-24 COEFFICIENT OF CUBIC TERM IN A
COLUMNS 25-48 COEFFICIENT OF CUBIC TERM IN E
COLUMNS 49-72 COEFFICIENT OF CUBIC TERM IN I

CARD 8H
COLUMNS 1-24 COEFFICIENT OF CUBIC TERM IN NODE
COLUMNS 25-48 COEFFICIENT OF CUBIC TERM IN OMEGA
COLUMNS 49-72 COEFFICIENT OF CUBIC TERM IN M

918 READ 6006, NCASE, CASE, NYE, NME, NDE, NME, NME, TSE, (AINPUT(N), N=1, 24)
DO 421 I = 1, 12
421 JA(I) = 0
ORBI = 0.0 00
ORB2 = 0.0 00
WRITE (21, 3999) NCASE, CASE, (JA(I), I=1, 12), ORBI, ORB2
NVE = 1900 + NYE

C CONVERT EPOCH CALENDAR DATE TO EPOCH JULIAN DATE AT 0 HOURS
C UNIVERSAL TIME.
C DJD = DJUL(NME, NDE, NYE)

C CONVERT EPOCH UNIVERSAL TIME IN HOURS, MINUTES, AND SECONDS
C TO SECONDS.

00051280
00051290
00051300
00051310
00051320
00051330
00051340
00051350
00051360
00051370
00051380
00051390
00051400
00051410
00051420
00051430
00051440
00051450
00051460
00051470
00051480
00051490
00051500
00051510
00051520
00051530
00051540
00051550
00051560
00051570
00051580
00051590
00051600
00051610
00051620
00051630
00051640
00051650
00051660
00051670
00051680
00051690
00051700
00051710
00051720
00051730
00051740
00051750
00051760
00051770
00051780
00051790
00051800
00051810
00051820
00051830

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1SN 0228      TME=NMH*3000
1SN 0229      TME=NMH*60
1SN 0230      TSEP=TM+TME*TS
1SN 0231      DLE=JOG*(TSEP/8.64E 04)
1SN 0232      IF(NSUPP)9181,9182,9181
1SN 0233      READ CANDS 81,8J,..... FROM SR SUPPO
1SN 0234      9181 CALL SUPPO
1SN 0235      9183 WRITE(6,1886)NARG
1SN 0236      STOP
1SN 0237      9184 IF(NARG-99)9182,9182,9183
1SN 0238      9182 WRITE(6,1831)
1SN 0239      WRITE(6,1882)CASE
1SN 0240      WRITE(6,6010)JL,NYE,NME,NDE,NME,NMNE,TSE
1SN 0241      GO TO (8210,8220,8220,8230).INPUT
1SN 0242      8210 IF(NUNIT)8212,8211,8212
1SN 0243      8211 WRITE(6,1836)(AINPUT(I),I=1,6)
1SN 0244      GO TO 8250
1SN 0245      8212 WRITE(6,1837)(AINPUT(I),I=1,6)
1SN 0246      GO TO 8250
1SN 0247      8220 IF(NUNIT)8222,8221,8222
1SN 0248      8221 WRITE(6,1832)(AINPUT(I),I=1,6)
1SN 0249      GO TO 8250
1SN 0250      8222 WRITE(6,1833)(AINPUT(I),I=1,6)
1SN 0251      GO TO 8250
1SN 0252      8230 IF(NUNIT)8232,8231,8232
1SN 0253      8231 WRITE(6,1834)(AINPUT(I),I=1,6)
1SN 0254      GO TO 8250
1SN 0255      8232 WRITE(6,1835)(AINPUT(I),I=1,6)
1SN 0256      8250 IF(NUNIT)8252,8251,8252
1SN 0257      8251 WRITE(6,1838)(AINPUT(I),I=7,12)
1SN 0258      WRITE(6,1840)(AINPUT(I),I=13,16)
1SN 0259      WRITE(6,1842)(AINPUT(I),I=19,24)
1SN 0260      GO TO 8260
1SN 0261      8252 WRITE(6,1839)(AINPUT(I),I=7,12)
1SN 0262      WRITE(6,1841)(AINPUT(I),I=13,16)
1SN 0263      WRITE(6,1843)(AINPUT(I),I=19,24)
1SN 0264      8260 CONTINUE
1SN 0265      IF(NUNIT)8100,8110,8100
1SN 0266      C
1SN 0267      C IF ELEMENTS ARE INPUT INDICES 4,10,16,22 ARE CHANGED TO REFER TO
1SN 0268      C MEAN ANOMLY AND 6,12,18,24 TO NODE
1SN 0269      C
1SN 0270      8100 GO TO(8101,8102,8102,8101).INPUT
1SN 0271      8101 DO 8104 I=1,24
1SN 0272      8104 A(I)=AINPUT(I)
1SN 0273      A(I 4)=AINPUT( 6)
1SN 0274      A(I 6)=AINPUT( 4)
1SN 0275      A(I 10)=AINPUT(12)
1SN 0276      A(I 12)=AINPUT(10)
1SN 0277      A(I 16)=AINPUT(18)
1SN 0278      A(I 18)=AINPUT(16)
1SN 0279      A(I 22)=AINPUT(24)
1SN 0280      A(I 24)=AINPUT(22)
1SN 0281      AIDM( 1)=AINPUT( 1)*REM
1SN 0282      AIDM( 2)=AINPUT( 2)
1SN 0283      00051840
1SN 0284      00051850
1SN 0285      00051860
1SN 0286      00051870
1SN 0287      00051880
1SN 0288      00051890
1SN 0289      00051900
1SN 0290      00051910
1SN 0291      00051920
1SN 0292      00051930
1SN 0293      00051940
1SN 0294      00051950
1SN 0295      00051960
1SN 0296      00051970
1SN 0297      00051980
1SN 0298      00051990
1SN 0299      00052000
1SN 0300      00052010
1SN 0301      00052020
1SN 0302      00052030
1SN 0303      00052040
1SN 0304      00052050
1SN 0305      00052060
1SN 0306      00052070
1SN 0307      00052080
1SN 0308      00052090
1SN 0309      00052100
1SN 0310      00052110
1SN 0311      00052120
1SN 0312      00052130
1SN 0313      00052140
1SN 0314      00052150
1SN 0315      00052160
1SN 0316      00052170
1SN 0317      00052180
1SN 0318      00052190
1SN 0319      00052200
1SN 0320      00052210
1SN 0321      00052220
1SN 0322      00052230
1SN 0323      00052240
1SN 0324      00052250
1SN 0325      00052260
1SN 0326      00052270
1SN 0327      00052280
1SN 0328      00052290
1SN 0329      00052300
1SN 0330      00052310
1SN 0331      00052320
1SN 0332      00052330
1SN 0333      00052340
1SN 0334      00052350
1SN 0335      00052360
1SN 0336      00052370
1SN 0337      00052380
1SN 0338      00052390

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ISN 0279	AICH(3)=AINPUT(3)*CONV	00052400
ISN 0280	AICH(4)=AINPUT(4)*CONV	00052410
ISN 0281	AICH(5)=AINPUT(5)*CONV	00052420
ISN 0282	AICH(6)=AINPUT(6)*CONV	00052430
ISN 0283	AICH(7)=AINPUT(7)*REM/TUH	00052440
ISN 0284	AICH(8)=AINPUT(8)/TUH	00052450
ISN 0285	AICH(9)=AINPUT(9)*CONV/TUH	00052460
ISN 0286	AICH(10)=AINPUT(10)*CONV/TUH	00052470
ISN 0287	AICH(11)=AINPUT(11)*CONV/TUH	00052480
ISN 0288	AICH(12)=AINPUT(12)*CONV/TUH	00052490
ISN 0289	AICH(13)=AINPUT(13)*REM*1.D 04/TUH**2	00052500
ISN 0290	AICH(14)=AINPUT(14)*1.D 04/TUH**2	00052510
ISN 0291	AICH(15)=AINPUT(15)*CONV*1.D 04/TUH**2	00052520
ISN 0292	AICH(16)=AINPUT(16)*CONV*1.D 04/TUH**2	00052530
ISN 0293	AICH(17)=AINPUT(17)*CONV*1.D 04/TUH**2	00052540
ISN 0294	AICH(18)=AINPUT(18)*CONV*1.D 04/TUH**2	00052550
ISN 0295	AICH(19)=AINPUT(19)*REM*1.D 06/TUH**3	00052560
ISN 0296	AICH(20)=AINPUT(20)*1.D 06/TUH**3	00052570
ISN 0297	AICH(21)=AINPUT(21)*CONV*1.D 06/TUH**3	00052580
ISN 0298	AICH(22)=AINPUT(22)*CONV*1.D 06/TUH**3	00052590
ISN 0299	AICH(23)=AINPUT(23)*CONV*1.D 06/TUH**3	00052600
ISN 0300	AICH(24)=AINPUT(24)*CONV*1.D 06/TUH**3	00052610
ISN 0301	GO TO 8120	00052620
ISN 0302	8102 DO 8103 I=1,24	00052630
ISN 0303	8103 AI(1)=AINPUT(1)	00052640
ISN 0304	AI(10)=AINPUT(12)	00052650
ISN 0305	AI(12)=AINPUT(10)	00052660
ISN 0306	AI(16)=AINPUT(18)	00052670
ISN 0307	AI(18)=AINPUT(16)	00052680
ISN 0308	AI(22)=AINPUT(24)	00052690
ISN 0309	AI(24)=AINPUT(22)	00052700
ISN 0310	AICH(1)=AINPUT(1)*REM	00052710
ISN 0311	AICH(2)=AINPUT(2)*REM	00052720
ISN 0312	AICH(3)=AINPUT(3)*REM	00052730
ISN 0313	AICH(4)=AINPUT(4)*REM/TUH	00052740
ISN 0314	AICH(5)=AINPUT(5)*REM/TUH	00052750
ISN 0315	AICH(6)=AINPUT(6)*REM/TUH	00052760
ISN 0316	GO TO 8107	00052770
ISN 0317	8110 GO TO 8111,8112,8112,8111,INPUT	00052780
ISN 0318	8111 DO 8114 I=1,24	00052790
ISN 0319	8114 AICH(1)=AINPUT(1)	00052800
ISN 0320	AI(1)=AINPUT(1)/REM	00052810
ISN 0321	AI(2)=AINPUT(2)	00052820
ISN 0322	AI(3)=AINPUT(3)/CONV	00052830
ISN 0323	AI(4)=AINPUT(6)/CONV	00052840
ISN 0324	AI(5)=AINPUT(5)/CONV	00052850
ISN 0325	AI(6)=AINPUT(7)/CONV	00052860
ISN 0326	AI(7)=AINPUT(7)*TUN/REM	00052870
ISN 0327	AI(8)=AINPUT(8)*TUN	00052880
ISN 0328	AI(9)=AINPUT(5)*TUN/CONV	00052890
ISN 0329	AI(10)=AINPUT(12)*TUN/CONV	00052900
ISN 0330	AI(11)=AINPUT(11)*TUN/CONV	00052910
ISN 0331	AI(12)=AINPUT(10)*TUN/CONV	00052920
ISN 0332	AI(13)=AINPUT(13)*TUN**2*1.D-04/REM	00052930
ISN 0333	AI(14)=AINPUT(14)*TUN**2*1.D-04	00052940
ISN 0334	AI(15)=AINPUT(15)*TUN**2*1.D-04/CONV	00052950

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ISN 0335      AI(16)=AINPUT(18)*TUN**2*1.D-04/CONV
ISN 0336      AI(17)=AINPUT(17)*TUN**2*1.D-04/CONV
ISN 0337      AI(18)=AINPUT(16)*TUN**2*1.D-04/CONV
ISN 0338      AI(19)=AINPUT(19)*TUN**3*1.D-06/REM
ISN 0339      AI(20)=AINPUT(20)*TUN**3*1.D-06
ISN 0340      AI(21)=AINPUT(21)*TUN**3*1.D-06/CONV
ISN 0341      AI(22)=AINPUT(24)*TUN**3*1.D-06/CONV
ISN 0342      AI(23)=AINPUT(23)*TUN**3*1.D-06/CONV
ISN 0343      AI(24)=AINPUT(22)*TUN**3*1.D-06/CONV
ISN 0344      GO TO 8120
ISN 0345      8112 DO 8113 I=1,24
ISN 0346      8113 AIDM(I)=AINPUT(I)
ISN 0347      DO 8116 I=1,3
ISN 0348      AI(I)=AINPUT(I)/REM
ISN 0349      8116 AI(I+3)=AINPUT(I+3)*TUN/REM
ISN 0350      GO TO 8121
ISN 0351      8120 NYE19=NYE-1900
ISN 0352      WRITE(6,18581)
ISN 0353      WRITE(6,1882)CASE
ISN 0354      WRITE(6,6010)DJL,NYE,NME,NDE,NNE,NMNE,TSE
ISN 0355      GO TO (9210,9220,9220,9230).INPUT
ISN 0356      9210 IF(NUNIT)9211,9212,9211
ISN 0357      9211 WRITE(6,1836)(AIDM (I),I=1,6)
ISN 0358      GO TO 9250
ISN 0359      9212 WRITE(6,1837)(AI(1),AI(2),AI(3),AI(6),AI(5),AI(4))
ISN 0360      GO TO 9250
ISN 0361      9220 IF(NUNIT)9222,9221,9222
ISN 0362      9221 WRITE(6,1833)(AI (I),I=1,6)
ISN 0363      GO TO 9250
ISN 0364      9222 WRITE(6,1832)(AIDM (I),I=1,6)
ISN 0365      GO TO 9250
ISN 0366      9230 IF(NUNIT)9222,9231,9232
ISN 0367      9231 WRITE(6,1835)(AI(1),AI(2),AI(3),AI(6),AI(5),AI(4))
ISN 0368      GO TO 9250
ISN 0369      9232 WRITE(6,1834)(AIDM (I),I=1,6)
ISN 0370      9250 IF(NUNIT)9252,9251,9252
ISN 0371      9251 WRITE(6,1839)(AI(7),AI(8),AI(9),AI(12),AI(11),AI(10))
ISN 0372      WRITE(6,1841)(AI(13),AI(14),AI(15),AI(18),AI(17),AI(16))
ISN 0373      WRITE(6,1843)(AI(19),AI(20),AI(21),AI(24),AI(23),AI(22))
ISN 0374      GO TO 9260
ISN 0375      9252 WRITE(6,1838)(AIDM (I),I=7,12)
ISN 0376      WRITE(6,1840)(AIDM (I),I=13,18)
ISN 0377      WRITE(6,1842)(AIDM (I),I=19,24)
ISN 0378      9260 CONTINUE
ISN 0379      WRITE(6,1869)ERR,NMAX
ISN 0380      WRITE(6,1878)
C
C      CONVERT EPOCH UNIVERSAL TIME IN HOURS, MINUTES, AND SECONDS
C      TO EPOCH UNIVERSAL TIME IN RADIAN.
C
C      2142 TIME0=HMSRZ(NME,NMNE,TSE)
C      GO TO(3002,1,28,3002).INPUT
C      CCNVERT TO KM AND DEGREES
C      3002 AI(1)=AI(1)*RE
C      AI(3)=AI(3)*CONV
C      AAMA=AI(4)
00052960
00052970
00052980
00052990
00053000
00053010
00053020
00053030
00053040
00053050
00053060
00053070
00053080
00053090
00053100
00053110
00053120
00053130
00053140
00053150
00053160
00053170
00053180
00053190
00053200
00053210
00053220
00053230
00053240
00053250
00053260
00053270
00053280
00053290
00053300
00053310
00053320
00053330
00053340
00053350
00053360
00053370
00053380
00053390
00053400
00053410
00053420
00053430
00053440
00053450
00053460
00053470
00053480
00053490
00053500
00053510

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1SN 0386      A1(4)=A1(6)*CONV      00053520
1SN 0387      A1(5)=A1(5)*CONV      00053530
1SN 0388      A1(6)=A1(6)*CONV      00053540
                                00053550
                                00053560
                                00053570
                                00053580
1SN 0389      C 28 CONTINUE          00053590
                                00053600
                                00053610
                                00053620
                                00053630
                                00053640
                                00053650
                                00053660
                                00053670
                                00053680
                                00053690
                                00053700
                                00053710
                                00053720
                                00053730
                                00053740
                                00053750
                                00053760
                                00053770
                                00053780
                                00053790
                                00053800
                                00053810
                                00053820
                                00053830
                                00053840
                                00053850
                                00053860
                                00053870
                                00053880
                                00053890
                                00053900
                                00053910
                                00053920
                                00053930
                                00053940
                                00053950
                                00053960
                                00053970
                                00053980
                                00053990
                                00054000
                                00054010
                                00054020
                                00054030
                                00054040
                                00054050
                                00054060
                                00054070

C          CUNVERT INPUT PARAMETERS TO OSCILLATING ELEMENTS

C          C 28 CONTINUE
C          DN = REDUCTION OF MEAN MOTION OF MEAN ANOMALY FROM BROUWER
C          KOZAI IN RADIANS PER SECOND.
C          CALL PARA(INPUT,A1,AGM,GOIF,NMAX,NIT,NSHORT,DN)
C          CALL ELAV (FX,VX,EF,RADV,A,PER,EN,GM,ERR,GOIF,NMAX,NIT)
C          ED=ER*CONV
C          FC=F*CONV
C          RADVM=RADV*1.0-03
C          IF (INPUT-4)190,8190,8190
C          UN = REDUCTION OF MEAN MOTION OF MEAN ANOMALY FROM BROUWER
C          KOZAI IN RADIANS PER SECOND.
C          CALL BRWR(DA,A,IRC,NN,GOIF,NMAX,NIT,NSHORT,DN)
C          AR=A1IC(1)/HE
C          DO 8069 J=1,6
C          A108R(J)=A10(1)
C          A108R(1)=A108R(1)/1.0 03
C          DO 8068 J=3,6
C          A108R(J)=A108R(J)*CONV
C          RATE(1)=0.0 0C
C          RATE(2)=0.0 0C
C          RATE(3)=0.0 0C
C          RATE(4)=SS(3)*CONV*3.6D 03
C          RATE(5)=SS(2)*CONV*3.6D 03
C          RATE(6)=SS(1)*CONV*3.6D 03
C          ENC=XX(1)*CONV*3.6D 03
C          RATE(7)=RATE(4)+RATE(5)
C          DO 8070 I=1,7
C          DO 8070 J=1,6
C          LPCOE(1,J)=0.0 0C
C          LPCOE(2,2)=EL(10)
C          LPCOE(2,4)=EL(11)
C          LPCOE(2,6)=EL(12)
C          DO 8071 J=1,6
C          LPCOE(3,J)=LPCOE(2,J)*EL(13)
C          LPCOE(4,1)=EL(8)
C          LPCOE(4,3)=EL(9)
C          LPCOE(4,5)=EL(7)
C          LPCOE(5,1)=EL(5)
C          LPCOE(5,3)=EL(6)
C          LPCOE(5,5)=EL(4)
C          LPCOE(6,1)=EL(2)
C          LPCOE(6,3)=EL(3)
C          LPCOE(6,5)=EL(1)
C          DO 8073 J=1,6
C          LPCOE(7,J)=LPCOE(4,J)*LPCOE(5,J)
C          DO 8072 I=3,7
C          DO 8072 J=1,6
C          LPCOE(1,J)=LPCOE(1,J)*CONV
C          WRITE(6,18C0)
C          WRITE(6,1874)

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ISN 0435	WRITE(6,1882)CASE	00054080
ISN 0436	WRITE(6,6010)DJL,NYE,NME,NDE,NHE,NMNE,TSE	00054090
ISN 0437	DA=DN*CONV*3600.0 (0	00054100
ISN 0438	WRITE(6,1872)	00054110
ISN 0439	WRITE(6,1882)	00054120
ISN 0440	IF(NKQZ)7003,7CC4,7C03	00054130
ISN 0441	7003 WRITE(6,1890)	00054140
ISN 0442	WRITE(6,1867)DN	00054150
ISN 0443	GO TO 7C05	00054160
ISN 0444	7004 WRITE(6,1889)	00054170
ISN 0445	7005 WRITE(6,1870)	00054180
ISN 0446	WRITE(6,1871)END,RATE(4),RATE(5),RATE(6),RATE(7)	00054190
ISN 0447	PAN = 3.60 02/RATE(6)	00054200
ISN 0448	WRITE(6,1881)PAN	00054210
ISN 0449	WRITE(6,1873)	00054220
ISN 0450	WRITE(6,1838)(AIDM (1),I=7,12)	00054230
ISN 0451	WRITE(6,1840)(AIDM (1),I=13,18)	00054240
ISN 0452	WRITE(6,1842)(AIDM (1),I=19,24)	00054250
ISN 0453	WRITE(6,1830)	00054260
ISN 0454	WRITE(6,1830)	00054270
ISN 0455	WRITE(6,1810)((LPCDEF(1,J),I=1,7),J=1,3)	00054280
ISN 0456	WRITE(6,1811)((LPCDEF(1,J),I=1,7),J=4,6)	00054290
ISN 0457	WRITE(6,1830)	00054300
ISN 0458	WRITE(6,1830)	00054310
ISN 0459	WRITE(6,1830)	00054320
ISN 0460	WRITE(6,1830)	00054330
ISN 0461	WRITE(6,1857)	00054340
ISN 0462	IF (NSHORT)8074,8075,8076	00054350
ISN 0463	WRITE(6,1883)	00054360
ISN 0464	GO TO 8077	00054370
ISN 0465	WRITE(6,1884)	00054380
ISN 0466	GO TO 8077	00054390
ISN 0467	WRITE(6,1885)	00054400
ISN 0468	WRITE(6,1880)REM,CON2	00054410
ISN 0469	WRITE(6,1875)	00054420
ISN 0470	WRITE(6,1882)CASE	00054430
ISN 0471	WRITE(6,6010)DJL,NYE,NME,NDE,NHE,NMNE,TSE	00054440
ISN 0472	WRITE(6,1859)	00054450
ISN 0473	WRITE(6,1819)	00054460
ISN 0474	WRITE(6,1816)	00054470
ISN 0475	WRITE(6,1860)AR,A110(2),A110(3)	00054480
ISN 0476	WRITE(6,1860)A110(4),A110(5),A110(6)	00054490
ISN 0477	WRITE(6,1817)	00054500
ISN 0478	WRITE(6,1860)(A110BR(J),J=1,3)	00054510
ISN 0479	WRITE(6,1860)(A110CR(J),J=4,6)	00054520
ISN 0480	DO 8150 J=1,6	00054530
ISN 0481	A11BR(J)=XX(J)	00054540
ISN 0482	A1ER(J)=XX(J)*6	00054550
ISN 0483	ASHR(J)=ASHORT(J)	00054560
ISN 0484	ABR(J)=AB(J)	00054570
ISN 0485	ABRP=ABR(11)/RE	00054580
ISN 0486	WRITE(6,1820)	00054590
ISN 0487	WRITE(6,1816)	00054600
ISN 0488	WRITE(6,1860)ABRR,ABR(2),ABR(3)	00054610
ISN 0489	WRITE(6,1860)ABR(4),ABR(5),ABR(6)	00054620
ISN 0490	A11BR(1)=A11BR(11)/1.0 03	00054630

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1SN 0491      A1BR(1)=A1BR(1)/1.D 03
1SN 0492      ASR(1)=ASR(1)/1.D 03
1SN 0493      ARE(1)=ARE(1)/1.D 03
1SN 0494      DO 8151 J=3,6
1SN 0495      A1BR(J)=A1BR(J)*CONV
1SN 0496      A1BR(J)=A1BR(J)*CONV
1SN 0497      ASR(J)=ASR(J)*CONV
1SN 0498      ARE(J)=ARE(J)*CONV
1SN 0499      8151 ABR(J)=ABR(J)*CONV
1SN 0500      WRITE(6,1817)
1SN 0501      WRITE(6,1860)(ABR(J),J=1,3)
1SN 0502      WRITE(6,1860)(ABR(J),J=4,6)
1SN 0503      DO 8152 J=1,3
1SN 0504      RXBM(J)=RXBM(J)/1.D 03
1SN 0505      VXBMC(J)=VXBMC(J)*3.D 00
1SN 0506      VXBMC(J)=VXBMC(J)/REM
1SN 0507      8152 VXBMC(J)=VXBMC(J)*TUM/REM
1SN 0508      WRITE(6,1861)
1SN 0509      WRITE(6,1862)
1SN 0510      WRITE(6,1860)(RXBMC(J),J=1,3)
1SN 0511      WRITE(6,1860)(RXBMC(J),J=1,3)
1SN 0512      WRITE(6,1860)(RXBMC(J),J=1,3)
1SN 0513      WRITE(6,1860)(VXBMC(J),J=1,3)
1SN 0514      WRITE(6,1860)(VXBMC(J),J=1,3)
1SN 0515      WRITE(6,1818)
1SN 0516      WRITE(6,1860)(RXB(J),J=1,3)
1SN 0517      WRITE(6,1860)(VXB(J),J=1,3)
1SN 0518      GO TO(8271,8271,8271,8272).INPUT
1SN 0519      8271 WRITE(6,1863)
1SN 0520      WRITE(6,1868)
1SN 0521      WRITE(6,1817)
1SN 0522      WRITE(6,1865)(JADIN(J),J=1,6)
1SN 0523      WRITE(6,1894)N
1SN 0524      WRITE(6,1866)INC
1SN 0525      8272 DO 8153 J=1,6
1SN 0526      DLL(J)=A1BR(J)-A1BR(J)
1SN 0527      8153 DLS(J)=ASHR(J)-A1BR(J)
1SN 0528      WRITE(6,1876)
1SN 0529      WRITE(6,1882)CASE
1SN 0530      WRITE(6,1830)
1SN 0531      WRITE(6,1830)
1SN 0532      WRITE(6,1830)
1SN 0533      WRITE(6,1830)
1SN 0534      WRITE(6,6085)DL,NYE,NME,NDE,NHE,NMNE,TSE,DMIN
1SN 0535      WRITE(6,6083)
1SN 0536      8303 WRITE(6,1802)(A1BR(J),J=1,6)
1SN 0537      WRITE(6,1803)(DLL(J),J=1,6)
1SN 0538      WRITE(6,1804)(A1BR(J),J=1,6)
1SN 0539      WRITE(6,1806)(DLS(J),J=1,6)
1SN 0540      WRITE(6,1848)DASUM
1SN 0541      WRITE(6,1849)(DSUP(J),J=1,6)
1SN 0542      WRITE(6,1801)(ABR(J),J=1,6)
1SN 0543      WRITE(6,1830)
1SN 0544      WRITE(6,1879)RADVM,ED,FD
1SN 0545      WRITE(6,1867)NIT
1SN 0546      WRITE(6,1868)GOLF

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15N 0588	8057	WRITE(6,6085)DM,NY,NM,ND,IM,IM,TS,OMIN	00055760
15N 0589	8056	IF(INSORT)8062,8062,8063	00055770
15N 0590	8062	CALL BRWR2(DTIMES, E,F,RADV,GDIF,NMAX,NITS,	00055780
15N 0591	GO TO 8064		00055790
15N 0592	8063	CALL BRWR4(DTIMES, E,F,RADV,GDIF,NMAX,NITS,	00055800
15N 0593	8064	ED=ESCONV	00055810
15N 0594	FL=FSCONV		00055820
15N 0595	RADVM=RAOV*1.D-03		00055830
15N 0596	DO 8050 J=1,6		00055840
15N 0597	AGER(J)=AG(J)		00055850
15N 0598	D1BR(J)=D1(J)		00055860
15N 0599	D2BR(J)=D2(J)		00055870
15N 0600	D3BR(J)=D3(J)		00055880
15N 0601	A1BR(J)=XK(J)		00055890
15N 0602	A1ER(J)=XK(J*6)		00055900
15N 0603	ASHR(J)=ASHORT(J)		00055910
15N 0604	ABR(J)=AB(J)		00055920
15N 0605	AGER(1)=AG(1)/1.D 03		00055930
15N 0606	D1BR(1)=D1(1)/1.D 03		00055940
15N 0607	D2BR(1)=D2(1)/1.D 03		00055950
15N 0608	D3BR(1)=D3(1)/1.D 03		00055960
15N 0609	A1BR(1)=A1BR(1)/1.D 03		00055970
15N 0610	A1ER(1)=A1ER(1)/1.D 03		00055980
15N 0611	ASHR(1)=ASHR(1)/1.D 03		00055990
15N 0612	ABR(1)=ABR(1)/1.D 03		00056000
15N 0613	DO 8051 J=3,6		00056010
15N 0614	AGER(J)=AG(J)*CONV		00056020
15N 0615	D1BR(J)=D1(J)*CONV		00056030
15N 0616	D2BR(J)=D2(J)*CONV		00056040
15N 0617	D3BR(J)=D3(J)*CONV		00056050
15N 0618	A1BR(J)=A1BR(J)*CONV		00056060
15N 0619	A1ER(J)=A1ER(J)*CONV		00056070
15N 0620	ASHR(J)=ASHR(J)*CONV		00056080
15N 0621	ABR(J)=ABR(J)*CONV		00056090
15N 0622	DO 8052 J=1,3		00056100
15N 0623	RXBR(J)=RXB(J)/1.D 03		00056110
15N 0624	VXBR(J)=VXB(J)*3.D 00		00056120
15N 0625	DO 8053 J=1,6		00056130
15N 0626	DL(J)=A1BR(J)-A1BR(J)		00056140
15N 0627	DLS(J)=ASHR(J)-A1BR(J)		00056150
15N 0628	IF(KCOUNT)8404,8403,8404		00056160
15N 0629	8403	WRITE(6,6083)	00056170
15N 0630	WRITE(6,1844)(AGBR(J),J=1,6)		00056180
15N 0631	WRITE(6,1845)(D1BR(J),J=1,6)		00056190
15N 0632	WRITE(6,1846)(D2BR(J),J=1,6)		00056200
15N 0633	WRITE(6,1847)(D3BR(J),J=1,6)		00056210
15N 0634	WRITE(6,1802)(A1ER(J),J=1,6)		00056220
15N 0635	WRITE(6,1803)(DL(J),J=1,6)		00056230
15N 0636	WRITE(6,1804)(A1BR(J), J=1,6)		00056240
15N 0637	WRITE(6,1806)(DLS(J),J=1,6)		00056250
15N 0638	WRITE(6,1848)DASUP		00056260
15N 0639	WRITE(6,1849)(DSUP(J),J=1,6)		00056270
15N 0640	WRITE(6,1801)(ABR(J),J=1,6)		00056280
15N 0641	WRITE(6,1830)		00056290
15N 0642	WRITE(6,1879)RADVP,ED,FD		00056300
15N 0643	WRITE(6,1867)INIT		00056310

15N 0644	WRITE(6,1868)GDI	00056320
15N 0645	WRITE(6,1805)(RXBM(J),J=1,3),(VXB(J),J=1,3)	00056330
15N 0646	WRITE(6,1830)	00056340
15N 0647	WRITE(6,1830)	00056350
15N 0648	WRITE(6,1830)	00056360
15N 0649	WRITE(6,1830)	00056370
15N 0650	MWT=DTIMES/6-D 01 + 1-D-08	00056380
15N 0651	CALL WRT6(MWT,RXB,VXB,ACC,ABR,21,22,23,1,0,0)	00056390
15N 0652	KCOUNT=KCOUNT+1	00056400
15N 0653	IF(KCOUNT-KPR)8406,8405,8405	00056410
15N 0654	8405 KCOUNT=0	00056420
15N 0655	8406 IF(KEF-MEFL)8061,2060,8061	00056430
15N 0656	8060 KEF=0	00056440
15N 0657	8061 CONTINUE	00056450
15N 0658	IF (DTIMES -XLAP)7016,7998,7998	00056460
15N 0659	7998 IF (LAST12,7999,42	00056470
15N 0660	7999 ENC FILE 21	00056480
15N 0661	GO TO 7900	00056490
15N 0662	300 PRINT 6039	00056500
15N 0663	STOP	00056510
15N 0664	END	00056520

***** F U R T R A N C R O S S R E F E R E N C E L I S T I N G *****

SYMBOL	INTERNAL STATEMENT NUMBERS														
	0138	0165	0178	0181	0184	0185	0186	0187	0192	0198	0207	0383	0397	0485	
RE															
RX	0005	0391													
SS	0005	0009													
TS	0582	0588							0556						
T0	0006	0009													
VX	0005	0391	0551	0552	0553										
XX	0005	0009	0481	0482	0601	0602									
ABRN	0605	0484	0485	0488	0488	0489	0489	0489	0493	0493	0498	0498	0500	0501	
	0621	0640	0651												
ACC	0005	0111	0651												
ARG	0006	0009													
CJJ	0009	0189													
CJ4	0009	0190													
CMU	0009	0195													
DAU	0605	0121	0122	0123	0124	0125	0126	0127	0128	0129	0130	0131	0132	0209	
	0216														
DJL	0231	0240	0354	0436	0471	0534	0574								
DJG	0009	0227	0231	0504	0579										
DLI	0005	0526	0537	0626	0635										
DLS	0005	0527	0539	0627	0637										
ENO	0409	0446													
ERR	0009	0119	0157	0379	0391										
FJ2	0134	0161	0168	0174	0180	0184	0188	0207							
FJ3	0135	0162	0169	0175	0180	0185	0189	0207							
FJ4	0136	0163	0170	0176	0180	0186	0190	0207							
FJ5	0137	0164	0171	0177	0180	0187	0207								
IRC	0147	0150	0217	0218	0396	0524									
KEF	0108	0586	0586	0655	0656										
KPR	0554	0566	0567	0653											
KSW	0007	0105	0106												
MWT	0650	0651													
MDA	0147	0150	0208												
NDZ	0220	0227	0240	0354	0436	0471	0534								
NHE	0220	0228	0240	0354	0381	0436	0471	0534							
NIT	0390	0391	0396	0545	0590	0592	0643								
NME	0220	0227	0240	0354	0436	0471	0534								
NVE	0220	0226	0226	0227	0240	0351	0354								
PAN	0447	0448							0436	0471	0534				
PER	0391	0557													
REC	0192	0196													
REM	0009	0180	0181	0182	0201	0277	0283	0289	0295	0310	0311	0312	0313	0314	
	0348	0349	0468	0505	0506	0550	0569								
	0348	0349	0503	0516	0623										
RXB	0005	0009													
THE	0228	0230													
TSE	0220	0230	0240	0354	0381	0436	0471	0534							
TUH	0010	0197	0198	0205	0283	0284	0285	0286	0287	0288	0289	0290	0291	0292	
	0298	0299	0300	0313	0314	0315	0326	0327	0328	0329	0330	0331	0332	0333	
	0339	0340	0341	0342	0343	0349	0366								
TUS	0196	0197	0205												
VAB	0005	0009	0504	0517	0624										
VXE	0005	0551	0552												
XXX	0005	0009	0409												
AAMA	0385	0388													
ABRN	0485	0488													

***** G A T R A N C R U S I R E F E R E N C E L I S T I N G *****

SYMBOL	INTERNAL STATEMENT NUMBERS
XLAF	0564 0658
XLAS	0564 0571
ALLOT	0580
ALLB*	0605 0481 0490 0495 0495 0526 0536 0601 0609 0618 0618 0626 0634
BBR4*	0396
BBR42	0550
BBR4	0592
CCCE*	0006
DACIN	0006 0127 0128 0129 0130 0131 0132 0211 0522
CAJ25	0009
DASUM	0009 0540 0628
DLJ25	0009
DLSUM	0009
D5QRT	0184 0196 0196
HMSR2	0381
IDINT	0578
INPUT	0147 0150 0155 0241 0260 0317 0355 0382 0390 0395 0518
KL43T	0504 0659
KPAGE	0107
NCASL	0220
NR02D	0009 0147 0150
NLUNG	0009 0147 0150
NSECD	0009 0147 0150 0151 0153 0562
NSPIL	0151 0502
NSP F	0009 0147 0150 0232
NUNIT	0147 0150 0242 0247 0252 0256 0265 0356 0361 0366 0370
NYE19	0351
NYM19	0585
RADVM	0394 0544 0595 0642
RHMS2	0582
RXBMC	0006
SCCEP	0006 0009
SUPPC	0233
TIME*	0004
TIME*	0381 0577 0586 0511
VAENC	0006
VAENC	0005 0135 0143 0245 0248 0250 0253 0255 0257 0258 0259 0261 0262 0263 0268 0269 0270 0271 0272
AIINPUT	0005 0240 0242 0275 0276 0277 0278 0279 0280 0281 0282 0283 0284 0285 0286 0287 0288 0289 0290 0291
AIINPUT	0273 0293 0294 0295 0296 0297 0298 0299 0300 0301 0302 0303 0304 0305 0306 0307 0308 0309 0310 0311 0312
AIINPUT	0313 0314 0315 0316 0317 0318 0319 0320 0321 0322 0323 0324 0325 0326 0327 0328 0329 0330 0331 0332 0333 0334
AIINPUT	0335 0336 0337 0338 0339 0340 0341 0342 0343 0344 0345 0346 0347 0348 0349
ARGNLT	0006 0009 0483 0603
ASHRT	0006 0009 0483 0603
AIIOBR	0005 0394 0400 0400 0402 0402 0478 0479
LTAKES	0009 0140 0571 0572 0573 0574 0577 0590 0592 0650 0658
JULCAL	0584
KCOUNT	0009 0587 0628 0652 0652 0653 0654
LCP023	0117 0118
LPC023	0004 0005 0413 0414 0415 0416 0418 0418 0419 0420 0421 0422 0423 0424 0425 0426 0427 0429 0429
LPC023	0429 0432 0432 0435 0436
NERK1	0005 0147 0150 0390 0396 0462 0589
NSHRT	0009 0147 0150 0390 0396 0462 0589
NSUPP*	0009 0147 0150 0390 0396 0462 0589

*****O T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL INTERNAL STATEMENT NUMBERS
NSUPPE 0152 C503
SLITET 0117
SSATCH 0106

*****ORTRAN CROSS REFERENCE LISTING*****

LABEL	DEFINED	REFERENCES		CROSS	REFERENCE		LISTING
		0119	0118		0167	0172	
1	0119	0118	0382				
2	0156	0155					
3	0157	0156	0156				
4	0158	0156	0156				
5	0159	0158					
6	0160	0159					
7	0167	0159	0159				
8	0168	0167					
9	0173	0167					
10	0180	0158					
11	0183	0158	0166	0167	0172	0179	
28	0389	0382					
42	0559	0659	0659				
100	0111	0110					
300	0662	0155	0155				
421	0221	0221					
422	0567	0566	0566				
423	0568	0566					
913	0208						
914	0209	0208					
915	0213	0208	0208				
916	0216	0215					
917	0218	0217	0217				
918	0220	0217					
1000	0008						
1031	0003						
1003	0011						
1004	0096						
1005	0144						
1011	0219						
1800	0012	0148	0194	0202	0433	0570	
1801	0013	0542	0640				
1802	0014	0536	0634				
1803	0015	0537	0635				
1804	0016	0538	0636				
1805	0017	0547	0645				
1806	0018	0539	0637				
1808	0019	0520					
1810	0020	0455					
1811	0021	0456					
1816	0022	0474	0487				
1817	0023	0477	0499	0521			
1818	0024	0515					
1819	0025	0473					
1820	0026	0486					
1821	0027	0508					
1822	0028	0205					
1823	0029	0204					
1825	0030	0512					
1827	0031	0203					
1830	0032	0453	0454	0457	0458	0459	0460
1831	0033	0238	0238	0531	0532	0533	0543
1832	0034	0248	0364	0646	0647	0648	0649
1833	0035	0250	0362				

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

LABEL	DEFINED	REFERENCES
2142	0381	
3002	0383	0382 0362
3999	0090	0225
5034	0155	
5042	0117	
6002	0091	0139 0143
6003	0092	0147
6004	0093	0180
6005	0094	0207
6006	0095	0220
6010	0097	0240 0354 0436 0471
6039	0098	0462
6040	0099	0157
6041	0100	0116
6050	0101	0209
6083	0102	0535 0629
6085	0103	0534 0568
7001	0145	
7002	0146	
7003	0441	0440 0440
7004	0444	
7005	0445	0443
7316	0572	0658
7900	0139	0661
7998	0659	0658 0658
7999	0660	0659
8001	0105	
8002	0106	0104
8050	0604	0596
8051	0621	0613
8052	0624	0622
8053	0627	0625
8054	0589	0587 0587
8057	0528	0587
8058	0113	0112
8059	0576	0575
8060	0656	0655
8061	0657	0655 0655
8062	0590	0589 0589
8063	0592	0589
8064	0593	0591
8070	0413	0411 0412
8071	0418	0417
8072	0432	0430 0431
8073	0429	0428
8074	0463	0462
8075	0465	0462
8076	0467	0462
8077	0468	0464 0466
8098	0402	0401
8099	0399	0398
8100	0266	0265 0265
8101	0267	0266 0266
8102	0302	0266 0266

***** O F T H A N C R O S S R E F E R E N C E L I S T I N G *****

LABEL	DEFINED	REFERENCES
8103	0303	0302
8104	0208	0267
8107	0283	0316
8110	0317	0245
8111	0318	0317 0317
8112	0345	0317 0317
8113	0346	0345
8114	0319	0318
8116	0349	0347
8120	0351	0301 0344
8121	0326	0350
8150	0464	0480
8151	0498	0494
8152	0506	0502
8155	0527	0525
8190	0396	0395
8191	0397	0395
8210	0242	0241
8211	0243	0242
8212	0245	0242 0242
8220	0247	0241 0241
8221	0248	0247
8222	0250	0247 0247
8230	0252	0241
8231	0253	0252
8232	0255	0252 0252
8250	0256	0244 0246
8251	0257	0249 0251 0254
8252	0261	0256
8260	0264	0260
8271	0519	0518
8272	0525	0518
8303	0536	0518
8403	0629	0628
8404	0650	0628
8405	0654	0652
8406	0655	0653
9141	0211	0210
9181	0233	0210
9182	0238	0232 0232
9183	0235	0232 0237 0237
9184	0237	0234 0234
9210	0356	0355
9211	0357	0356
9212	0359	0356
9220	0361	0355 0355
9221	0362	0361
9222	0364	0361
9230	0366	0355
9231	0367	0366
9232	0365	0366
9250	0370	0360
9251	0371	0363 0365 0368
9252	0375	0370 0370

***** ORTRAN CROSS REFERENCE LISTING*****

LAUCL	DEFINED	REFERENCES
9260	0378	0374

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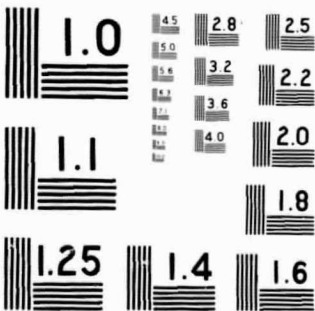
C/COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=58,SOURCE=EBDCIC,NOLIST,NODECK,LOAD=MAP,NODEIT,LD,XREF
CSUBROUTINE ADLH
CSUBROUTINE ADLH
SUBROUTINE ADLH (DLJ2S,DLJ3,DLJA,DLJSUM,TH,E,G,A,R,ET,F,CAPG,CAPL,
1CMU,CJ2S,CJ3,CJA,C)
IMPLICIT REAL*8(A-H,O-Z,S)
DOUBLE PRECISION DLJ2S,DLJ3,DLJA,DLJSUM,TH,E,G,A,R,ET,F,CAPG,CAPL,
1CMU,CJ2S,CJ3,CJA,D
DIMENSION D(19)
C(1)=TH*TH
C(2)=D(1)*D(1)
C(3)=E*F
C(4)=E*D(3)
C(5)=ET*ET*ET
C(6)=(3.000*CMU**4*CJ2S/(128.000*CAPG**7))*(8.000*D(1))*(1.000-5.0
1D00*(1))*(3.000*(5.000-18.000*D(1))+5.000*D(2))-2.000*D(3)*(1.000-D
2(1))*(1.000-15.000*D(1))+D(1)*D(2)*D(3)*D(4)*D(5)
C(7)=(3.000*CMU**4*CJ2S/(512.000*CAPL*CAPG**6))*(A/R)**3
C(8)=8.000*(9.000-26.000*D(1)+4.000*D(2))+4.000*D(3)*(137.000-
198.000*D(1))+37.000*D(2))+16.000*D(5))*(1.000-3.000*D(1))**2
C(9)=2.000*(16.000*(1.000-3.000*D(1))**2*(1.000-D(5))/D(3)+
14.000*(19.000-36.000*D(1)+35.000*D(2))+D(3)*(73.000-234.000*D(1)+
212.000*D(2)))*D(3)*D(4)
C(10)=4.000*D(3)*(4.000*(1.000-3.000*D(1))**2*(1.000-D(5))/D(3)+290.000
1D00-66.000*D(1))+45.000*D(2))*D(3)*D(4)*D(5)+2.000*D(4)*(11.000-30.000
2D(1))+27.000*D(2))*D(3)*D(4)*D(5)
C(11)=3.000*D(4)*(1.000-3.000*D(1))*D(3)*D(4)*D(5)+2.000*D(3)*
1(1.000-3.000*D(1))*(1.000-D(5))/D(3)+8.000*D(3)*D(4)*D(5)+E*
2(4.000*(1.000-3.000*D(1))*(1.000-D(5))/D(3)+32.000-D(3))*(17.000-
3147.000*D(1))*D(3)*D(4)*D(5)+2.000*D(3)*D(4)*D(5)
C(12)=4.000*(13.000-27.000*D(1))*D(3)*D(4)*D(5)+E*(28.000*(1.000-
13.000*D(5))*(1.000-3.000*D(1))*D(3)*D(4)*D(5)+E*(28.000*(1.000-
23.000*D(1))*(1.000-D(5))/D(3)+32.000*(1.000-4.000*D(1))-D(3))*D(5.
3000-77.000*D(1))*D(3)*D(4)*D(5)
C(13)=2.000*D(3)*(1.000-3.000*D(1))*D(5)+5.000*(1.000-D(5))/D(3)+4.
1D(1)*D(3)*D(4)*D(5)+2.000*D(3)*D(4)*D(5)*(1.000-3.000*D(1))*D(3)*D(4)
2D(1)+2.000*D(3)
C(14)=9.000*D(4)*D(3)*D(4)*D(5)+54.000*D(3)*D(4)*D(5)+4.000*
1G*(E*(148.000-13.000*D(3))*D(3)*D(4)*D(5)+20.000*(2.000
2+7.000*D(3))*D(3)*D(4)*D(5)+F*(F+G))
C(14)=(D(14)+3.000*(28.000+17.000*D(3))*D(3)*D(4)*D(5)+4.000*D(3)+
154.000*D(3)*D(4)*D(5)+9.000*(F+G)*D(4)*D(5)+4.000*
2G)*(1.000-D(1))**2
DLJ2S=D(6)+D(7)*(C(14)*D(9)+D(10)+4.000*(1.000-D(1))*(D(11)+D(12)+
1D(13)))*D(14)
D(15)=(-3.000/8.000)*D(14)*D(15)*D(16)*D(17)*D(18)*D(19)
C(16)=5.000/8.000*D(14)*D(15)*D(16)*D(17)*D(18)*D(19)
C(17)=(-2.000/64.000)*D(14)*D(15)*D(16)*D(17)*D(18)*D(19)
D(18)=5.000/16.000*(1.000-D(1))*(1.000-7.000*D(1))
C(19)=(-35.000/64.000)*D(14)*D(15)*D(16)*D(17)*D(18)*D(19)
DLJ3=CMU**3*CJ3/CAPL**5*(D(15)*E*(A/R)**4*D(16)*D(17)*D(18)*D(19)
1(G)+C(16)*A/R)**4*D(15)*D(16)*D(17)*D(18)*D(19)
DLJA=CMU**4*CJA/CAPL**7*(D(17)*E*(A/R)**5-1.000/ET**7*(1.000+3.000
1*(C(13)+2.000)*D(16))*(A/R)**5*D(18)*D(19)+3.000*D(3)+4.000
2*ET**7)*D(18)*D(19)+D(19)*(A/R)**5*D(18)*D(19)+4.000*(F+G))
DLJSUM=DLJ2S+DLJ3+DLJA
RETURN
END

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***** C R O S S R E F E R E N C E L I S T I N G *****

[illegible]

N 7
289



LEVEL 16 (1 JULY 68)

OS/360 FORTRAN H

DATE 69.206/19.05.09

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COMPILER OPTIONS = NAME= MAIN,OPT=00,LINECT=58,SOURCE,EBDCIC,NOLIST,NODECK,LOAD,MAP,NOEDIT,ID,XREF
CSUBROUTINE ALLOT
ISN 0002 REAL FUNCTION ALLOT (X)
ISN 0003 IMPLICIT REAL*8(A-H,O-Z,$)
C
C VERSION OF 07/22/63
C FORTRAN FUNCTION
C
C PURPOSE
C REDUCES AN ANGLE OF ANY MAGNITUDE AND SIGN BY MODULUS 2 PI
C AND ADDS 2 PI IF ANGLE IS NEGATIVE. THE RESULTING ANGLE IS
C POSITIVE BETWEEN 0, AND +2 PI RADIANS.
C
C CALLING SEQUENCE
C D NAME = ALLOT (X)
C NAME CHANGED 11/15/68 FROM ALLOTZ TO ALLOT
C
C INPUT X = ANGLE IN RADIANS
C
C X MUST BE AVAILABLE IN CALLING PROGRAM IN DOUBLE PRECISION
C FORM.
C
C OUTPUT NAME = ANGLE IN RADIANS BETWEEN 0 AND + 2 PI RADIANS
C
C NAME IS RETURNED TO CALLING PROGRAM IN DOUBLE PRECISION
C FORM.
C
C REFERENCE
C *****
C
C METHOD *****
C
C RESTRICTIONS
C *****
C
C 1 CONTINUE
C
C ACCURACY
C INTERNAL ARITHMETIC IS PERFORMED IN DOUBLE PRECISION.
C
C REQUIRED SUBPROGRAMS
C DMAD
C
C TIMING
C NO ESTIMATE AVAILABLE
C
C ***** START PROGRAM *****
C

```

ISN 0005
ISN 0006
ISN 0007
ISN 0008
ISN 0009

2 ALLOT =DMAC(X.6.283185307179586D0)
3 IF (ALLOT) 4.E+5
4 ALLOT =ALLOT +6.283185307179586D0
5 RETURN
END

00058580
00058590
00058600
00058610
00058620

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL INTERNAL STATEMENT NUMBERS
X 0002 0005
DMAC 0005
ALLCT 0002 0005 0006 0007 0007

***** DRY HAN C R O S S R E F E R E N C E L I S T I N C *****

LABEL	DEFINIC	REFERENCES
1	0004	
2	0005	
3	0006	
4	0007	0006
5	0008	0006 0006

COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=58,SOURCE=EBDCIC,NOLIST,NODECK,LOAD,MAP,NODEIT,ID,XREF

C SUBROUTINE ATANG

C REAL FUNCTION ATANG(S,C)

C IMPLICIT REAL*8(A-H,O-Z,S)

C

C VERSION OF 03/03/64

C FORTRAN FUNCTION

C

C PURPOSE

C COMPUTES THE ARCTANGENT OF AN ANGLE WITH PROPER ALLOCATION

C OF QUADRANT TO THE ANGLE BETWEEN 0 AND + 2 PI RADIANS.

C

C CALLING SEQUENCE

C NAME = ATANG (S,C)

C

C INPUT

C S = DSIN(A)

C C = DCOS(A)

C WHERE D IS AN ARBITRARY POSITIVE CONSTANT (NORMALLY

C D = *1.0)

C

C OUTPUT

C NAME = ANGLE A IN RADIANS BETWEEN 0 AND + 2 PI RADIANS

C

C REFERENCE

C REFER TO MATHEMATICAL DESCRIPTION IN SUBPROGRAM WRITEUP

C

C METHOD

C USES FORTRAN FUNCTION DATAN. TESTS THE SIGNS OF

C SINE AND COSINE, THEN ADDS OR SUBTRACTS APPROPRIATE

C FRACTIONS OF 2 PI RADIANS TO ASSIGN ANGLE TO PROPER

C QUADRANT BETWEEN 0 AND + 2 PI RADIANS

C

C RESTRICTIONS

C ATANG(0/0) = 0 BY DEFINITION

C

C ACCURACY

C *****

C

C REQUIRED SUBPROGRAMS

C NONE

C

C TIMING

C NO ESTIMATE AVAILABLE

C

C

C

C

C

C

C

C

C

C

C

C

C

C

C

C

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C

C

C

C

C

C

ISN 0004

ISN 0005

ISN 0006

ISN 0007

ISN 0008

ISN 0009

ISN 0010

ISN 0011

ISN 0012

ISN 0013	110 ADD=3.14159265DC	00060810
ISN 0014	GO TO 124	00060820
ISN 0015	112 ATANQ=3.14159265DC	00060830
ISN 0016	RETURN	00060840
ISN 0017	114 ACC=3.14159265DC	00060850
ISN 0018	GO TO 132	00060860
ISN 0019	116 IF (S) 118,120,122	00060870
ISN 0020	118 ACC=6.28318531DC	00060880
ISN 0021	GO TO 132	00060890
ISN 0022	120 ATANQ=C.000	00060900
ISN 0023	RETURN	00060910
ISN 0024	122 ACC=0.000	00060920
ISN 0025	124 IF (LARS(S)-DARS(C)) 126,128,130	00060930
ISN 0026	126 ATANQ=ATAN(S/C)+ADD	00060940
ISN 0027	RETURN	00060950
ISN 0028	128 ATANQ=C.78539816300+ADD	00060960
ISN 0029	RETURN	00060970
ISN 0030	130 ATANQ=1.570796230-DATAN(C/S)+ADD	00060980
ISN 0031	RETURN	00060990
ISN 0032	132 IF (DARS(S)-DARS(C)) 126,134,136	00061000
ISN 0033	134 ATANQ=-0.78539816300+ADD	00061010
ISN 0034	RETURN	00061020
ISN 0035	136 ATANQ=-1.57079633DC-DATAN(C/S)+ADD	00061030
ISN 0036	RETURN	00061040
ISN 0037	END	00061050

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS
C	0002 CC04 0025 0026 0030 0032 0035
S	0002 CC05 0012 0019 0025 CC26 0030 0032 0035
ADD	0013 CC17 0020 CC24 CC26 CC28 0030 0033 0035
DABS	0025 CC25 CC32 0032
ATANQ	0002 0006 CC08 0010 0015 CC22 0026 0028 0030 0033 0035
DATAN	0026 0030 0035

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

LABEL	DEFINED	REFERENCES
100	0005	C004
102	0006	C005
104	0008	0005
106	0010	0005
108	0012	0004
110	0013	0012
112	0015	0012
114	0017	0012
116	0019	C004
118	0020	0019
120	0022	0019
122	0024	C019
124	0025	0014
126	0026	0025 0032
128	0028	0025
130	0030	0025
132	0032	C018 0021
134	0033	0032
136	0035	0032

```

COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=58,SOURCE=EBCDIC,NOLIST,NODECK,LOAD,MAP,NODEIT,ID,XREF
CSUBROUTINE ATANZ
15N 0002 REAL FUNCTION ATANZ(S,C)
15N 0003 IMPLICIT REAL*8(A-H,O-Z,S)
C
C CALLING SEQUENCE
C D NAME = ATANZ(S,C)
C
C INPUT
C S = SINE OF ANGLE (+ OR -)
C C = COSINE OF ANGLE (+ OR -)
C INPUT ARGUMENTS MUST BE AVAILABLE IN CALLING PROGRAM IN
C DOUBLE PRECISION FORM.
C
C OUTPUT
C NAME = ANGLE IN RADIANS BETWEEN 0 AND + 2 PI RADIANS
C
C NAME IS RETURNED TO CALLING PROGRAM IN DOUBLE PRECISION
C FORM.
C
C REFERENCE
C *****
C
C METHOD
C USES FORTRAN MONITOR FUNCTION DATANZ. IF ARGUMENT
C RETURNED BY DATANZ IS -, 2 PI RADIANS ARE ADDED.
C
C RESTRICTIONS
C *****
C
C ACCURACY
C INTERNAL ARITHMETIC IS PERFORMED IN DOUBLE PRECISION.
C
C REQUIRED SUBPROGRAMS
C NONE
C
15N 0004 2 CONTINUE
C
C TIMING
C NO ESTIMATE AVAILABLE
C
C
C ***** START PROGRAM *****
C
15N 0005 ATANZ = DATANZ(S,C)
15N 0006 3 1= (ATANZ) 4.5.5
15N 0007 4 ATANZ=ATANZ*6.2831853071795866DC
15N 0008 5 RETURN
15N 0009 END

```

***** F O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

SYMBOL	INTERNAL STATEMENT NUMBERS
C	C002 0005
S	C002 0005
ATAN2	C002 0005 C006 C007 0007
DATAN2	C005

***** F O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

LABEL	DEFINED	REFERENCES
2	0004	
3	0006	
4	0007	0006
5	0008	0006 0006

LEVEL 16 (1 JULY 68) OS/360 FORTRAN H DATE 69.206/19.05.16

COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=58,SOURCE=EBDCIC,NCLIST,NODECK,LOAD,MAP,NODEDIT,ID,XREF

1SN 0002 CSUBROUTINE BRWR
1SN 0003 IMPLICIT REAL*8(A-H,J-Z,S)
VERSION OF 10/02/63
FORTRAN SUBROUTINE

PURPOSE COMPUTES BROUWER MEAN ORBITAL ELEMENTS FROM OSCULATING
ORBITAL ELEMENTS BY MEANS OF AN ITERATIVE PROCESS.
CALLING SEQUENCE
CALL BRWR(DA,AT,J,K,GDIF,NMAX,NIT,NSHORT,CN)

INPUT
DA(1)= TRUNCATION FACTOR FOR SEMI-MAJOR AXIS - KILOMETERS
DA(2)= TRUNCATION FACTOR FOR ECCENTRICITY - DIMENSIONLESS
DA(3)= TRUNCATION FACTOR FOR INCLINATION - RADIAN
DA(4)= TRUNCATION FACTOR FOR RIGHT ASCENSION OF ASCENDING
NODE - RADIAN
DA(5)= TRUNCATION FACTOR FOR ARGUMENT OF PERIGEE - RADIAN
DA(6)= TRUNCATION FACTOR FOR MEAN ANOMALY - RADIAN

OSCULATING ORBITAL ELEMENTS AT EPOCH TIME
AT(1)= SEMI-MAJOR AXIS - KILOMETERS
AT(2)= ECCENTRICITY - DIMENSIONLESS
AT(3)= INCLINATION - RADIAN
AT(4)= RIGHT ASCENSION OF ASCENDING NODE - RADIAN
AT(5)= ARGUMENT OF PERIGEE - RADIAN
AT(6)= MEAN ANOMALY - RADIAN

J = MAXIMUM NUMBER OF ITERATIONS ALLOWED
NMAX = MAXIMUM NUMBER OF ITERATIONS ALLOWED
IN SOLVING KEPLER'S EQUATION
NSHORT SHORT-PERIOD PERTURBATIONS INCLUDED (00 = NO
LT=0
THE SHORT PERIOD PERTURBATIONS ARE
COMPUTED WITH E' AND 1".
GT=0
THE SHORT PERIOD PERTURBATIONS ARE
COMPUTED WITH E' AND 1'.)

OUTPUT
K = SERIAL NUMBER OF CURRENT ITERATION
GDIF = LAST CORRECTION (E2-E1)/E2
NIT = IN SOLVING KEPLER'S EQUATION.
NIT = SERIAL NUMBER OF CURRENT ITERATION IN SOLVING
KEPLER'S EQUATION
DN = REDUCTION OF MEAN MOTION OF MEAN ANOMALY FROM
BROUWER TO KUZAI IN RADIAN PER SECOND.

OUTPUT VIA COMMON

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00061990
00062000
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00062020
00062030
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00062060
00062070
00062080
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00062100
00062110
00062120
00062130
00062140
00062150
00062160
00062170
00062180
00062190
00062200
00062210

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C          BREWER MEAN ELEMENTS
C          A110(1) = SEMI-MAJOR AXIS
C          A110(2) = ECCENTRICITY
C          A110(3) = INCLINATION
C          A110(4) = RIGHT ASCENSION OF ASCENDING NODE
C          A110(5) = ARGUMENT OF PERIGEE
C          A110(6) = MEAN ANOMALY
C          REQUIRED SUBPROGRAMS
C          07/22/63 ALLOT
C          2 CONTINUE
C          07/17/63 BWR1
C          01/31/64 BWR2
C          BWR4
C          TIMING
C          NO ESTIMATE AVAILABLE
C          PROGRAM MODIFICATIONS
C          ***** START PROGRAM *****
C          100 FORMAT (//20H ***** WARNING ***** /79H NO CONVERGENCE IN BRR*****
C          1R SUBROUTINE. BREWER MEAN ELEMENTS ARE NOT ACCURATE //)
C          DIMENSION DUM1(100)
C          DIMENSION DAT(6),AT(6),A110(6),DUMX(18),AC(6),DAT(6),TLAT(6)
C          COMMON DUM1,A110,DUMX,AC
C          K = 0
C          DO 10 N=1,6
C          10 A110(N) = AT(N)
C          1 CALL BWR1(DN)
C          IF(ISHORT)101,101,102
C          101 CALL BWR2(3.000, E,F,RADV,GDIF,NMAX,NIT)
C          GO TO 102
C          102 CALL BWR4(6.000, E,F,RADV,GDIF,NMAX,NIT)
C          103 K = K + 1
C          IF (K-J) 4,4,3
C          3 PRINT 100
C          RETURN
C          4 DO 5 N=1,6
C          DAT(N) = AT(N) - AC(N)
C          5 TLAT(N) = CABS(DAT(N))
C          DO 6 N=1,6
C          IF (TLAT(N)-DA(N)) 6,6,8
C          6 CONTINUE
C          7 RETURN
C
ISN 0004
00062220
00062230
00062240
00062250
00062260
00062270
00062280
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00062760
00062770

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ISN 0028
ISN 0029
ISN 0030
ISN 0031
ISN 0032
ISN 0033
ISN 0034
ISN 0035

9 DO 9 N=1,6
9 A110(N) = A110(N) + DAT(N)
A110(3)=ALLOT(A110(3))
A110(4)=ALLOT(A110(4))
A110(5)=ALLOT(A110(5))
A110(6)=ALLOT(A110(6))
GO TO 1
END

00062780
00062790
00062800
00062810
00062820
00062830
00062840
00062850

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS
E	0014 0016
F	0014 0016
J	0002 0018
K	0017 0017 0018
N	0010 0011 0021 0022
AC	0007 0008 0022
AT	0002 0007 0011 0022
DA	0002 0007 0025
DN	0002 0012
DAT	0007 0022 0023 0029
NIT	0002 0014 0016
A110	0007 0008 0011 0029
DAES	0023 0030 0031 0032 0033
DUNX	0007 0008
DUN1	0006 0008
GDIF	0002 0014 0016
NMAX	0002 0014 0016
RADV	0014 0016
TDAT	0007 0023 0025
ALLUT	0030 0031 0032 0033
BERWR	0002
BRWH1	0012
BRWR2	0014
BRWR4	0016
NSHGT	0002 0013

*****F D R T R A N C R O S S R E F E R E N C E L I S T I N G*****

LABEL	DEFINED	REFERENCES
1	0012	0034
2	0004	
3	0019	0018
4	0021	0018 0018
5	0023	0021
6	0026	0024 0025 0025
7	0027	
8	0028	0025
9	0029	0028
10	0011	0010
100	0005	0019
101	0014	0013 0013
102	0016	0013
103	0017	0015

15N 0018	8 F(3)=F(2)+F(2)	00063960
15N 0019	9 F(4)=F(2)+F(3)	00063970
15N 0020	10 F(5)=F(2)+F(2)	00063980
15N 0021	11 F(6)=F(4)+F(4)	00063990
15N 0022	12 F(7)=F(2)+F(2)	00064000
15N 0023	13 F(11)=F(2)+F(2)	00064010
15N 0024	14 F(12)=F(2)+F(2)	00064020
15N 0025	15 F(13)=F(1)+F(2)	00064030
15N 0026	16 F(14)=F(3)+F(2)	00064040
15N 0027	17 F(15)=F(2)+F(1)	00064050
15N 0028	18 F(16)=F(2)+F(1)	00064060
15N 0029	19 F(17)=F(1)+F(1)	00064070
15N 0030	20 F(18)=F(1)+F(1)	00064080
15N 0031	21 F(19)=F(1)+F(1)	00064090
15N 0032	22 F(11)=F(1)+F(1)	00064100
15N 0033	23 F(12)=F(1)+F(1)	00064110
15N 0034	24 F(13)=F(1)+F(1)	00064120
15N 0035	25 F(14)=F(1)+F(1)	00064130
15N 0036	26 F(15)=F(1)+F(1)	00064140
15N 0037	27 F(16)=F(1)+F(1)	00064150
15N 0038	28 F(17)=F(1)+F(1)	00064160
15N 0039	29 F(18)=F(1)+F(1)	00064170
15N 0040	30 F(19)=F(1)+F(1)	00064180
15N 0041	31 F(16)=F(1)+F(1)	00064190
15N 0042	32 F(17)=F(1)+F(1)	00064200
15N 0043	33 F(18)=F(1)+F(1)	00064210
15N 0044	34 F(19)=F(1)+F(1)	00064220
15N 0045	35 F(10)=F(1)+F(1)	00064230
15N 0046	36 F(11)=F(1)+F(1)	00064240
15N 0047	37 F(12)=F(1)+F(1)	00064250
15N 0048	38 F(13)=F(1)+F(1)	00064260
15N 0049	39 F(14)=F(1)+F(1)	00064270
15N 0050	40 F(15)=F(1)+F(1)	00064280
15N 0051	41 F(16)=F(1)+F(1)	00064290
15N 0052	42 F(17)=F(1)+F(1)	00064300
15N 0053	43 F(18)=F(1)+F(1)	00064310
15N 0054	44 F(19)=F(1)+F(1)	00064320
15N 0055	45 F(10)=F(1)+F(1)	00064330
15N 0056	46 F(11)=F(1)+F(1)	00064340
15N 0057	47 F(12)=F(1)+F(1)	00064350
15N 0058	48 F(13)=F(1)+F(1)	00064360
15N 0059	49 F(14)=F(1)+F(1)	00064370
15N 0060	50 F(15)=F(1)+F(1)	00064380
15N 0061	51 F(16)=F(1)+F(1)	00064390
15N 0062	52 F(17)=F(1)+F(1)	00064400
15N 0063	53 F(18)=F(1)+F(1)	00064410
15N 0064	54 F(19)=F(1)+F(1)	00064420
15N 0065	55 F(10)=F(1)+F(1)	00064430
15N 0066	56 F(11)=F(1)+F(1)	00064440
15N 0067	57 F(12)=F(1)+F(1)	00064450
15N 0068	58 F(13)=F(1)+F(1)	00064460
15N 0069	59 F(14)=F(1)+F(1)	00064470
15N 0070	60 F(15)=F(1)+F(1)	00064480
15N 0071	61 F(16)=F(1)+F(1)	00064490
15N 0072	62 F(17)=F(1)+F(1)	00064500
15N 0073	63 F(18)=F(1)+F(1)	00064510

ISN 0113
ISN 0114
ISN 0115
ISN 0116
ISN 0117
ISN 0118
ISN 0119
ISN 0120
ISN 0121
ISN 0122
ISN 0123
ISN 0124

WRITE(6,1002)(F(J),J=1, 9)
WRITE(6,1003)
WRITE(6,1002)(G(J),J=1,13)
WRITE(6,1003)
WRITE(6,1002)(GP(J),J=1, 5)
WRITE(6,1003)
WRITE(6,1002)(S(J),J=1, 5)
WRITE(6,1003)
WRITE(6,1002)A2,A3
WRITE(6,1009)
BY RETURN
END

0065080
0065090
0065100
0065110
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*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL INTERNAL STATEMENT NUMBERS

DCS	0032	
DLJ3	0013	
DLJ4	0013	
DSIN	0067	
DSUP	0012	0013
DUM1	0010	0013
DUM2	0010	0013
ECA1	0013	
NARG	0013	
NKZ	0013	0084
NSEC	0013	
BRW1	0002	
CCCF	0012	0013
CAJ2S	0013	
DASUM	0013	
CLJ2S	0013	
DLSUM	0013	
DSGT	0024	0063 0080
AKZU	0013	
NLONG	0013	
NSEC	0013	
NSUPP	0013	
SCCF	0012	0013
ARGMT	0012	0013
ASHGT	0012	0013
CTIME	0013	
KCLUNT	0013	
KMEAN	0085	
NGRW1	0013	0100
ASHGT	0013	
NSUPD	0013	

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

REFERENCES

LABEL	DEFINED
1	0005
5	0014
6	0015
7	0017
8	0018
9	0019
10	0020
11	0021
12	0022
13	0023
14	0024
15	0025
16	0026
17	0027
18	0028
19	0029
20	0030
21	0031
22	0032
23	0033
24	0034
25	0035
26	0036
27	0037
28	0038
29	0039
30	0040
31	0041
32	0042
33	0043
34	0044
35	0045
36	0046
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39	0049
40	0050
41	0051
42	0052
43	0053
44	0054
45	0055
46	0056
47	0057
48	0058
49	0059
50	0060
51	0061
52	0062
53	0063
54	0064
55	0065
56	0066
57	0067

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

LABEL	DEFINED	REFERENCES	
		0054	
58	0068		
59	0069		
60	0070		
61	0071		
62	0072		
63	0073		
64	0074		
65	0075		
66	0076		
67	0077		
68	0078		
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70	0080		
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74	0087		
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77	0090		
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83	0096		
84	0097		
85	0098		
86	0099		
87	0123		
89	0101		
1000	0006		
1001	0007		
1002	0008		
1003	0009		
2001	0004		
8000	0085		
		0100	0100
		0101	0122
		0102	
		0103	0105
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LEVEL 16 ( 1 JULY 68)
DATE 69.206/19.05.22
OS/360 FORTRAN H
COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=59,SOURCE=ERCDIC,NOLIST,NODECK,LOAD,MAP,NODEIT,IO,XREF
C SUBROUTINE BRWR2
C SUBROUTINE BRWR2 (/DT/, E,FAN,RADV,GCIF,NMAX,NIT)
C IMPLICIT REAL*8(A-H,O-Z,S)
C
C VERSION OF 01/31/64
C FORTRAN SUBROUTINE
C
C PURPOSE
C COMPUTES THE VALUES OF THE OSCILLATING ELEMENTS AND THE
C POSITION-VELOCITY VECTOR FOR ANY TIME.
C THE SHORT-PERIOD PERTURBATIONS ARE COMPUTED WITH E* AND I*.
C
C DUM1 IS A DUMMY VARIABLE INSERTED AS FIRST VARIABLE IN COMMON IN
C BRWR1,BRWR2,AND BRWR4 TO PERMIT SHIFTING OF VARIABLES IN COMMON
C AREA IF DESIRED. THE DIMENSION OF DUM1 MAY BE CHANGED BUT SHOULD
C BE THE SAME IN SUBROUTINES BRWR1,BRWR2,AND BRWR4,AND THE CALLING
C PROGRAM.
C
C CALLING SEQUENCE
C CALL BRWR2(DT,E,FAN,RADV,GCIF,NMAX,NIT)
C
C INPUT
C DT = TIME ELAPSED FROM EPOCH OF MEAN ELEMENTS - SECONDS
C NMAX = MAXIMUM NUMBER OF ITERATIONS ALLOWED
C SEE SUBROUTINE BRWR1 FOR INPUT VIA COMMON
C
C OUTPUT
C E = ECCENTRIC ANOMALY RADIANS
C FAN = TRUE ANOMALY RADIANS
C RADV = RADIUS VECTOR KILOMETERS
C GCIF = LAST CORRECTION (E2-E1)/E2
C IN SOLVING KEPLER'S EQUATION.
C NIT = SERIAL NUMBER OF CURRENT ITERATION IN SOLVING
C KEPLER'S EQUATION
C
C 1000 CONTINUE
C OUTPUT VIA COMMON
C OSCILLATING ORBITAL ELEMENTS AT TIME T = EPOCH TIME + DT
C
C A(1) = SEMI-MAJOR AXIS - KILOMETERS
C A(2) = ECCENTRICITY - DIMENSIONLESS
C A(3) = INCLINATION - RADIANS
C A(4) = HEIGHT ASCENSION OF ASCENDING NODE - RADIANS
C A(5) = ARGUMENT OF PERIGEE - RADIANS
C A(6) = MEAN ANOMALY - RADIANS
C
C REQUIRED SUBPROGRAMS
C ADLH
C 07/22/63 ALLOT
C 03/03/64 ATAND
C 03/02/64 ELRV
C 09/12/63 SUPPL
C KPEP
C
ISN 0002
ISN 0003
ISN 0004

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C      TIMING      NO ESTIMATE AVAILABLE
C
C
C
C      PROGRAM MODIFICATIONS
C
C
C      STATEMENT AL(1)=AL(1) ADDED 7/26/68
C
C***** START PROGRAM *****
C
C      DIMENSION DUM1(100)
C      DIMENSION AL(100),GP(5),AL(100),A(6),RX(3),VX(3),EL(13),F(9),
C      B(9),C(10),D(23),G(13),X(28),S(3)
C      DIMENSION AG(6),D1(6),D2(6),D3(6),AIDH(24)
C      DIMENSION T0(99),ARGO(99),ARGMT(99),ARG(99),CCDEF(99,6),
C      ISCDEF(99,6),DSUP(6),ASHORT(6),DADLH(19)
C
C      COMMON DUM1,A110,GP,ERR,A11,A1,A,RX,VX,ENO,ECA1,DIE,D11,A1,UL,
C      I,CI,UG,MI,UM,S,EL,A2,A3,B,C,D,G,X,F,
C      ICONV,AIDH,AG,D1,D2,D3,CMU,T0,DJ0,DTIME,S,ARGO,ARGMT,ARG,CCDEF,
C      ISCDEF,DSUP,CJ2S,CJ3,CJ4,DLJ2S,DLJ3,DLJ4,DLSUM,DAJ2S,DAJ3,DAJ4,
C      4DASUM,REM , ASHORT
C      5      NBRWR1,NLONG,NSHORT,NSEC,NSECC,NSUPP,NSUPPE,NARG,KCOUNTC0065980
C      6,NK0Z,NK0ZD
C      1850 FORMAT(/29X,74HC CONTRIBUTION OF J2 SQUARE, J3 * J4 TO SECOND ORDER,00066000
C      1 SHORT PERIOD TERMS IN A/ 39X,3(F15.7,3X)/)
C
C      DASUM=0.D 00
C      DO 700 I=1,6
C      700 DSUP(I)=0.D 00
C      COMPUTE SECULAR TERMS
C
C      S1 = S(1)
C      S2 = S(2)
C      S3 = S(3)
C      DS1 = S1
C      DS2 = S2
C      DS3=S3
C      AG(1)=A110(1)
C      AG(2)=A110(2)
C      AG(3)=A110(3)
C      AG(4)=A110(4) +DS2*DT
C      AG(5)=A110(5) +DS2*DT
C      AG(6)=A110(6) +DS1*DT
C      AG4=AG(4)
C      AG5=AG(5)
C      AG6=AG(6)
C      AG4)=ALLOT (AG4)
C      AG5)=ALLOT (AG5)
C      AG6)=ALLOT (AG6)
C      C1(1)=DT*AIDH(7)*1.D 00/3.6D 00
C      C1(2)=DT*AIDH(8)/3.6D 03
C      DO 1002 I=3,6
C      1002 D1(1)=DT*AIDH(1+6)/(CONV*3.6D 03)

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15N 0036      D2(1)=DT**2*ALDH(13)/1.296D 04      00066300
15N 0037      D2(2)=DT**2*ALDH(14)/1.296D 07      00066310
15N 0038      D2 10C3 I=3+6                        00066320
15N 0039      1003 D2(1)=DT**2*ALDH(1+12)/(CONV*1.296D 07) 00066330
C          TO ALLOW FOR THE FACT THAT ALH(I),I=13,18 ARE PER (10CMRS)SQUARED00066340
15N 0040      D2 1013 I=1+6                        00066350
15N 0041      1013 D2(1)=1.0-C4+D2(1)              00066360
15N 0042      D3(1)=DT**3*ALDH(19)/4.6656D 07      00066370
15N 0043      D3(2)=DT**3*ALDH(20)/4.6656D 10      00066380
15N 0044      D3 1004 I=3+6                        00066390
15N 0045      1004 D3(1)=DT**3*ALDH(1+18)/(CONV*4.6656D 10) 00066400
C          TO ALLOW FOR THE FACT THAT #IDH(I),I=19,24 ARE PER (10CMRS)CURED 00066410
15N 0046      D3 1014 I=1+6                        00066420
15N 0047      1014 D3(1)=1.0-06+D3(1)              00066430
15N 0048      D3 10C5 I=1+6                        00066440
15N 0049      1005 A11(1)=AG(1)*G[(1)+D2(1)+D3(1)] 00066450
C          A114=A11(4)                            00066460
C          A115=A11(5)                            00066470
C          A116=A11(6)                            00066480
9          A11(9) = ALLOT (A116)                  00066490
10         A11(5) = ALLOT (A115)                  00066500
11         A11(4) = ALLOT (A114)                  00066510
C          COMPUTE LONG PERIOD TERMS              00066520
C          C                                          00066530
15         X(1)=A11(5)+A11(5)                    00066540
16         X(2)=X(1)+A11(5)                      00066550
17         X(3)=X(2)+X(1)                        00066560
18         X(4)=X(3)+X(1)                        00066570
19         X(5)=X(4)+X(1)                        00066580
20         A11(6)=A11(5)+EL(1)*X(3)+EL(2)*X(4)+EL(3)*X(5) 00066590
21         A11(5)=A11(5)+EL(4)*X(3)+EL(5)*X(4)+EL(6)*X(5) 00066600
22         A11(4)=A11(4)+EL(7)*X(3)+EL(8)*X(4)+EL(9)*X(5) 00066610
23         C1E=EL(10)*COS(X(1))+EL(11)*DSIN(A11(5))+EL(12)*DSIN(X(2)) 00066620
24         D11=EL(13)*D1E                          00066630
25         A11(3)=A11(3)+D11                      00066640
26         A11(2)=A11(2)+D1E                      00066650
C          A11(1)=A11(1)                          00066660
28         A11=ALLOT(A11(6))                      00066670
29         G1 =ALLOT(A11(5))                      00066680
30         H1 =ALLOT(A11(4))                      00066690
C          TO WRITE LONG PERIOD TERMS              00066700
C          C                                          00066710
C          C                                          00066720
C          C                                          00066730
C          C                                          00066740
31         ECA1=X(1)*AL1+A11(2)*X(6)+X(7)*ERF,GE IF,NMAX,NIT) 00066750
33         X(8)=1.0-D2-A11(2)*X(7)                00066760
34         X(9)=1.0-C/X(8)                        00066770
35         X(10)=X(9)*X(9)                        00066780
36         X(11)=X(9)*X(10)                      00066790
37         X(12)=X(6)*X(9)*X(2)                  00066800
38         X(13)=X(9)*X(7)-A11(2)                00066810
C          X12 = X(12)                             00066820
C          X13 = X(13)                             00066830
C          DX13 = X13                              00066840
C          X(14) = ATANG(DX12,DX13)                00066850

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15N 0132	CAJ4 =2.D CO*CAPL*RE*DLJ4	00067420
15N 0133	DASUM=2.D CO*CAPL*RE*DLSUM	00067430
15N 0134	A(1)=A(1)+DASUM*1.D 03	00067440
15N 0135	IF(KCOUNT)6210+8259,6210	00067450
15N 0136	8299 IF(NSCJ)8300+6210,8300	00067460
15N 0137	8300 WRITE(6,1853)DAJ2S,DAJ3,DAJ4	00067470
15N 0138	6210 IF(NSUPP)621,63,621	00067480
15N 0139	621 CALL SUPP1	00067490
15N 0140	A(1)=A(1)+CSUP(1)*1.D 03	00067500
15N 0141	A(2)=A(2)+DSUP(2)	00067510
15N 0142	DC 629 I=3,6	00067520
15N 0143	629 A(1)=A(1)+CSUP(1)/CONV	00067530
	C	00067540
	C	00067550
	53 CALL ELHV (RX,VX,E,FAN,RADVA,P,EN,GP(1),ERR,GOIF,NMAX,NIT)	00067560
	RETURN	00067570
	END	00067580
		00067590

*****F U P T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS															
A	0006	0009	0050	0099	0100	0104	0105	0106	0108	0130	0134	0140	0140	0141	0141	0143
B	0006	0009	0077	0091	0094	0099	0101									0144
C	0006	0009	0050	0099	0102	0103										
D	0006	0009	0091	0092	0095	0095	0099	0100	0102	0102						
E	0002	0144														
F	0006	0009														
G	0006	0009	0095	0099												
I	0012	0013	0034	0035	0035	0038	0039	0039	0040	0041	0041	0044	0045	0045	0046	0049
P	0049	0049	0049	0049	0107	0108	0108	0120	0128	0128	0142	0143	0143	0143		
S	0006	0009	0014	0015	0016											
L	0115	0116														
X	0006	0009	0030	0037	0037	0038	0038	0039	0040	0040	0041	0041	0041	0041	0042	0043
	0004	0004	0072	0072	0073	0073	0074	0074	0075	0075	0075	0076	0076	0076	0077	0078
	0078	0079	0081	0083	0084	0085	0085	0085	0086	0086	0086	0087	0087	0088	0088	0089
	0090	0091	0091	0092	0092	0093	0093	0093	0094	0094	0094	0095	0095	0095	0095	0096
	0096	0096	0097	0097	0097	0097	0098	0098	0098	0098	0099	0099	0099	0100	0101	0102
	0103															
AG	0007	0009	0020	0021	0022	0023	0024	0025	0026	0027	0028	0029	0030	0031	0049	
A1	0006	0009	0001	0002	0003	0003	0003	0003	0003	0003	0003	0003	0003	0003	0003	0003
A2	0009															
A3	0009															
DT	0002	0003	0004	0005	0005	0005	0005	0005	0005	0005	0005	0005	0005	0005	0005	0005
D1	0007	0009	0032	0033	0035	0035	0035	0035	0035	0035	0035	0035	0035	0035	0035	0035
D2	0007	0009	0036	0036	0036	0036	0036	0036	0036	0036	0036	0036	0036	0036	0036	0036
D3	0007	0009	0042	0043	0045	0045	0045	0045	0045	0045	0045	0045	0045	0045	0045	0045
EL	0006	0009	0001	0001	0001	0001	0001	0001	0001	0001	0001	0001	0001	0001	0001	0001
EN	0144															
ET	0119	0122	0125													
GP	0006	0009	0144													
G1	0009	0070	0084	0084	0084	0084	0084	0084	0084	0084	0084	0084	0084	0084	0084	0084
H1	0009	0071	0103													
RA	0117	0118	0119													
RX	0006	0009	0144													
S1	0014	0017														
S2	0015	0016														
S3	0019	0019														
TH	0110	0122	0129													
TU	0008	0009														
UG	0009	0102	0105													
UH	0009	0103	0106													
UL	0009	0101	0104													
VX	0006	0009	0144													
AG4	0026	0029														
AG5	0027	0030														
AG6	0028	0031														
AL1	0009	0009	0072	0096	0101											
ARG	0008	0009														
ALL	0020	0009	0049	0050	0051	0052	0053	0054	0055	0056	0056	0057	0059	0061	0062	0063
	0064															0064
CJ3	0009	0122	0129													
CJ4	0009	0122	0129													
CMU	0009	0120	0122	0129												

SYMBOL	INTERNAL STATEMENT NUMBERS
000	0009
001	0017
002	0025
003	0031
004	0038
005	0044
006	0050
007	0056
008	0062
009	0068
010	0074
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058	0362
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062	0386
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***** DATA TRAN CROSS REFERENCE LIST IN G*****

SYMBOL INTERNAL STATEMENT NUMBERS

OLSON	009	0122	0129	0133
OSGAT	0115	0115	0120	0121
AKGZD	0009			
ALCNG	0009			
NSSCU	0009	0126	0136	
NSUPP	0009	0138		
SCLEF	0009	0009		
SUPPL	0139			
APRML	0113	0118	0120	0122 0129
ARGCT	0008	0009		
ASHOT	0008	0009	0108	
UTIME	0005			
EPRIE	0114			
FPRIME	0116	0119	0122	0129
GPRIME	0114	0122	0129	
KCOUNT	0009	0125	0135	
NRHRT	0009			
NSHRT	0009			
NSUPPD	0009			
RPRIME	0118	0124	0129	

***** DRYMAN CROSS REFERENCE LISTING *****

REFERENCES

DEFINED

LABEL

9 0033
 10 0034
 11 0035
 12 0036
 13 0037
 14 0038
 15 0039
 16 0040
 17 0041
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*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

LABEL	DEFINED	REFERENCES
1003	0039	0038
1004	0045	0044
1005	0049	0048
1013	0041	0040
1014	0047	0046
1828	0123	0128 0129
1850	0010	0137
1860	0124	0127
2000	0130	0125 0126
2001	0127	0126 0126
2002	0126	0125
6200	0110	0109 0109
6210	0138	0104 0135 0136
6220	0168	0107
8299	0136	0135
8300	0137	0136 0136

LEVEL 16 (3 JULY 68) OS/360 FORTRAN M DATE 69.206/19.05.26

COMPILER OPTIONS - 4ME= MAIN,OPT=00,LINECNT=58,SOURCE=EBDCIC,NOLIST,NODECK,LOAD,MAP,NOEDIT,ID,XREF
 CSUBROUTINE BRWR4
 ISN 0002 SUBROUTINE BRWR4 (/DT/, E,FAN,RADV,GDIF,NMAX,NIT)
 ISN 0003 IMPLICIT REAL*8(A-H,O-Z,S)

C
 C VERSION OF 01/31/64
 C FORTRAN SUBROUTINE
 C THE SHORT PERIOD TERMS COMPUTED IN THIS ROUTINE ARE COMPUTED WITH
 C E' AND I' INSTEAD OF WITH E" AND I" AS SPECIFIED BY BROUWER,
 C AJANOV, 1959.
 C OTHERWISE THE ROUTINE IS THE SAME AS BRWR2.
 C
 C PURPOSE
 C COMPUTES THE VALUES OF THE OSCILLATING ELEMENTS AND THE
 C POSITION-VELOCITY VECTOR FOR ANY TIME.
 C THE SHORT-PERIOD PERTURBATIONS ARE COMPUTED WITH E' AND I'.
 C
 C DUM1 IS A DUMMY VARIABLE INSERTED AS FIRST VARIABLE IN COMMON IN
 C BRWR1,BRWR2,AND BRWR4 TO PERMIT SHIFTING OF VARIABLES IN COMMON
 C AREA IF DESIRED. THE DIMENSION OF DUM1 MAY BE CHANGED BUT SHOULD
 C BE THE SAME IN SUBROUTINES BRWR1,BRWR2,AND BRWR4,AND THE CALLING
 C PROGRAM.
 C CALLING SEQUENCE
 C CALL BRWR4(DT,E,FAN,RADV,GDIF,NMAX,NIT)
 C
 C INPUT
 C DT = TIME ELAPSED FROM EPOCH OF MEAN ELEMENTS - SECONDS
 C NMAX = MAXIMUM NUMBER OF ITERATIONS ALLOWED
 C SEE SUBROUTINE BRWR1 FOR INPUT VIA COMMON
 C
 C OUTPUT
 C E = ECCENTRIC ANOMALY RADIANS
 C FAN = TRUE ANOMALY RADIANS
 C RADV = RADII'S VECTOR KILOMETERS
 C GDIF = LAST CORRECTION (E2-E1)/E2
 C NIT = IN SOLVING KEPLER'S EQUATION.
 C NIT = SERIAL NUMBER OF CURRENT ITERATION IN SOLVING
 C KEPLER'S EQUATION

1000 CONTINUE
 C
 C OUTPUT VIA COMMON
 C OSCILLATING ORBITAL ELEMENTS AT TIME T = EPOCH TIME + DT
 C
 C A(1) = SEMI-MAJOR AXIS - KILOMETERS
 C A(2) = ECCENTRICITY - DIMENSIONLESS
 C A(3) = INCLINATION - RADIANS
 C A(4) = RIGHT ASCENSION OF ASCENDING NODE - RADIANS
 C A(5) = ARGUMENT OF PERIGEE - RADIANS
 C A(6) = MEAN ANOMALY - RADIAN
 C
 C REQUIRED SUBPROGRAMS
 C 07/22/63 ALLOT
 C 03/03/64 ADLH
 C 03/02/64 ATANG
 C 03/02/64 ELRV

ISN 0004


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DO 1002 I=3,6
1002 D1(I)=DT*AIDH(I+6)/(CONV*3.6D 03)
D2(I)=DT**2*AIDH(I+3)/1.296D 04
D3(I)=DT**3*AIDH(I+1)/1.296D 07
DO 1003 I=3,6
1003 D2(I)=DT**2*AIDH(I+12)/(CONV*1.296D 07)
C TO ALLOW FOR THE FACT THAT AIDH(I+1)=13.18 ARE PER (100HRS)SQUARED
DO 1013 I=1,6
1013 D2(I)=1.4-C*4D2(I)
D3(I)=DT**3*AIDH(I+19)/4.6656D 07
D4(I)=DT**4*AIDH(I+20)/4.6656D 10
DO 1004 I=3,6
1004 D3(I)=DT**3*AIDH(I+18)/(CONV*4.6656D 10)
C TO ALLOW FOR THE FACT THAT AIDH(I+1)=19.24 ARE PER (100HRS)CUBED
DO 1014 I=1,6
1014 D3(I)=1.4-D*6D3(I)
DO 1005 I=1,6
1005 A11(I)=AG(I)+D1(I)+D2(I)+D3(I)
A114=A11(I+4)
A115=A11(I+5)
A116=A11(I+6)
9 A11(6) = ALLOT (A116)
10 A11(5) = ALLOT (A115)
11 A11(4) = ALLOT (A114)
C COMPUTE LONG PERIOD TERMS
C
15 X(1)=A11(5)*A11(5)
16 X(2)=X(1)*A11(5)
17 X(3)=X(2)*A11(5)
18 X(4)=X(3)*A11(5)
19 X(5)=X(4)*A11(5)
20 A1(6)=A11(6)+EL(1)*X(3)+EL(2)*X(4)+EL(3)*X(5)
21 A1(5)=A11(5)+EL(4)*X(3)+EL(5)*X(4)+EL(6)*X(5)
22 A1(4)=A11(4)+EL(7)*X(3)+EL(8)*X(4)+EL(9)*X(5)
23 D1E=EL(10)*DCOS(X(1))+EL(11)*DSIN(A11(5))+EL(12)*DSIN(X(2))
24 D1I=EL(13)*DIE
25 A1(3)=A11(3)+D1I
26 A1(2)=A11(2)+D1E
A1(1)=A11(1)
28 A1=ALLOT(A1(6))
29 G1 =ALLOT(A1(5))
30 H1 =ALLOT(A1(4))
C TO WRITE LONG PERIOD TERMS
C
C COMPUTE SHORT PERIOD TERMS
C
F2=A1(2)**2
B1=1.-D 00-F2
B2=CSORT(B1)
B3 =1*B2
B8=1.-D 00/B3
B9=1.-D 00/(B1*B1)
C1=GP(2)/A1(1)**2
C2=1.-5C 00*C1*B9
C9=C2/6.-D 00
O=BCOS(A1(3))

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ISN 0134 IF(NSEC162C0.6210.6200
ISN 0135 6200 TH=DCOS(A1(3))
ISN 0136 ECC=A1(2)
ISN 0137 GPRIME=A1(5)
ISN 0138 APRIME=A1(1)*1.D-C3/REM
ISN 0139 EPRIME=ECA1
ISN 0140 U=DSORT((1.D 00+A1(2))/(1.D 00-A1(2)))*DTAN(ECA1/2.D C0)
ISN 0141 FPRIME=2.D 00*ATAN(U)
ISN 0142 RA=1.D 00-A1(2)*COS(ECA1)
ISN 0143 RPRIME=RA*APRIME
ISN 0144 ET=DSORT(RA*(1.D C0+A1(2)*COS(FPRIME)))
ISN 0145 CAPL=CSORT(CMU*APRIME)
ISN 0146 CAPG=CAPL*DSORT(1.D 00-A1(2)**2)
ISN 0147 CALL ALUM(DLJ2S,DLJ3,DLJ4,DLSUM,
1 TH,ECC,GPRIME,APRIME,RPRIME,ET,FPRIME,CAPG,CAPL,CMU,
2CJ2S,CJ3,CJ4,DADLH)
DLPRL=(1.125J C0/CAPG**7)*(FJ2**2)*RA**(-3)
DLPRL=DLPRL*32**3
DLPRL=DLPRL*(1.D C0-5.D 00*TH**2)*(1.D 00-TH**2)
ECC=FPRIME-AL1
3000 IF(ECC-180.D 00/CUNV)3002+3001.3001
3001 ECC=ECC-360.D 00/CUNV
GO TO 3000
3002 IF(ECC+180.D 00/CUNV)3003+3004.3004
3003 ECC=ECC+360.D 00/CUNV
GO TO 3002
3004 DLPRL=DLPRL*ECC
STERM=DSIN(2.D 00*(FPRIME+G1))
DLPRL=DLPRL*STERM
DLSUM=DLSUM+DLPRL
1828 FORMAT(61X,D21.14)
1860
2002 IF(NSEC2000.20C2.2000
2001 WRITE(6,1860)
WRITE(6,1828)TH,ECC,GPRIME,APRIME,RPRIME,ET,EPRIME,RA,U,AL1,G1,
1ECC,FPRIME,CAPG,CAPL,CMU,CJ2S,CJ3,CJ4
WRITE(6,1828)DLJ2S,DLJ3,DLJ4,DLSUM
2000 CAJ2S=2.D 00*CAPL*REM*DLJ2S
DAJ3 =2.D 00*CAPL*REM*DLJ3
DAJ4 =2.D 00*CAPL*REM*DLJ4
DAPRL=2.D 00*CAPL*REM*DLPRL
DASUM=2.D 00*CAPL*REM*DLSUM
DSCA=1.D-0.3*(A1(1)-A1(10))**2/(4.D 00*A1(10)(1))
DASUM=DASUM+DSCA
A1(1)=A1(1)+DASUM*1.D 03
IF(KCOUNT)6210.8259.A210
8299 IF(NSEC218300.621C.8300
8300 WRITE(6,1850)DAJ2S,DAJ3,DAJ4,DAPRL,DSCA
6210 IF(NSUPP)621.63.621
621 CALL SUPP1
A1(1)=A1(1)+DSUP(1)*1.D 03
A1(2)=A1(2)+DSUP(2)
DO 629 I=3+6
629 A1(I)=A1(I)+DSUP(I)/CONV
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C C COMPUTE POSITION AND VELOCITY COMPONENTS
C 63 CALL ELRV (RX,X,E,FAN,RADV,A,P,EN,GP(1),ERR,GDIF,NMAX,NIT)
RETURN
END

ISN 0187
ISN 0188
ISN 0189

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS																		
A	0006	0009	0123	0124	0125	0129	0130	0131	0133	0175	0177	0177	0183	0183	0184	0184	0186	0186	0187
B	0006	0009																	
C	0006	0009																	
D	0006	0009																	
E	0002	0187																	
F	0006	0009																	
G	0006	0009																	
I	0013	0014	0035	0036	0036	0039	0040	0040	0041	0042	0042	0045	0046	0046	0047	0048	0048	0049	0050
P	0050	0050	0050	0050	0132	0133	0133	0133	0167	0167	0185	0186	0186	0186					
Q	0082	0083	0084	0084	0084	0085	0086	0087											
S	0006	0009	0015	0016	0017														
U	0140	0141	0158																
X	0006	0009	0057	0058	0058	0059	0059	0060	0061	0061	0062	0062	0063	0063	0063	0064	0064	0064	0064
	0065	0065	0067	0067	0068	0068	0069	0069	0099	0099	0100	0100	0101	0101	0102	0102	0103	0103	0103
	0103	0104	0106	0106	0108	0109	0110	0110	0111	0111	0111	0112	0112	0112	0113	0113	0114	0114	0115
	0115	0116	0116	0117	0117	0118	0118	0118	0119	0119	0119	0120	0120	0120	0120	0120	0120	0120	0121
	0121	0121	0122	0122	0122	0123	0123	0123	0123	0124	0124	0124	0124	0125	0125	0126	0127	0127	0127
	0128	0128																	
AG	0007	0009	0021	0022	0023	0024	0025	0026	0027	0028	0029	0030	0031	0032	0050				
A1	0006	0009	0062	0063	0064	0067	0068	0069	0070	0071	0072	0073	0079	0082	0093	0093	0094	0097	0098
	0103	0118	0121	0122	0122	0135	0136	0137	0138	0140	0140	0140	0142	0144	0146				
A2	0009																		
A3	0009																		
B1	0074	0075	0076	0078	0078	0094	0119												
B2	0075	0076	0102	0126	0149														
B3	0076	0077																	
B8	0077	0116																	
B9	0078	0080	0124																
C1	0079	0080	0123	0124															
C2	0080	0081																	
C9	0081	0083	0092	0096	0127														
DT	0002	0024	0025	0026	0033	0034	0036	0037	0038	0040	0043	0044	0046						
D1	0007	0009	0033	0034	0036	0050													
D2	0007	0009	0037	0038	0040	0042	0042	0050											
D3	0007	0009	0043	0044	0046	0048	0048	0050											
D5	0086	0088	0091	0092															
D6	0087	0089	0091	0092															
D7	0085	0090	0127																
EL	0006	0009	0062	0062	0062	0063	0063	0064	0064	0064	0064	0065	0065	0065	0066				
EN	0187																		
ET	0144	0147	0166																
F2	0073	0074																	
GP	0006	0009	0011	0079	0187														
G1	0009	0071	0109	0109	0127	0159	0168												
G9	0053																		
M1	0009	0072	0128																
RA	0142	0143	0144	0148	0168														
RX	0006	0009	0187																
S1	0015	0018																	
S2	0016	0019																	
S3	0017	0020																	
TH	0135	0147	0150	0150	0168														
T0	0008	0009																	

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS
UG	0009 0127 0130
UH	0009 0128 0131
UL	0009 0126 0129
VX	0006 0009 0127
AG4	0027 0030
AG5	0028 0031
AG6	0029 0032
AL1	0009 0070 0097 0121 0126 0151 0168
ARG	0008 0009
ALL	0006 0009 0050 0051 0052 0053 0054 0055 0056 0057 0057 0058 0060 0062 0063 0064 0065 0067 0068
CJ3	0009 0147 0166
CJ4	0009 0147 0168
CMU	0009 0145 0147 0168
C10	0083 0091 0128
DJ0	0009
DS1	0018 0026
DS2	0019 0025
DS3	0020 0024
D1E	0009 0065 0066 0068 0124
D11	0009 0066 0067 0125
C19	0088 0117 0120
D20	0089 0120
D21	0090 0127
D22	0091 0125
D23	0092 0124
ECC	0136 0147 0166
ENC	0009
EQC	0151 0152 0153 0155 0156 0156 0158 0168
ERR	0009 0097 0157
FAN	0002 0167
FJ2	0011 0148
G11	0054 0055 0096
G12	0055 0124
G13	0096 0120
NIT	0002 0057 0187
REM	0009 0011 0138 0170 0171 0172 0173 0174
X12	0104 0105
X13	0106 0107
ADLM	0147
AIDH	0007 0009 0033 0034 0036 0037 0038 0040 0043 0044 0046
ARGO	0008 0009
ALLC	0006 0009 0021 0022 0023 0024 0025 0026 0123 0124 0125 0175 0175
ALL4	0051 0056
ALL5	0052 0055
ALL6	0053 0054
CAFG	0146 0147 0148 0168
CAPL	0145 0146 0147 0168 0170 0171 0172 0173 0174
CJ25	0009 0147 0168
CONV	0009 0036 0040 0046 0152 0153 0155 0156 0186
DAJ3	0009 0171 0180
DAJ4	0009 0172 0180
DCUS	0060 0061 0065 0062 0115 0118 0118 0135 0142 0144
DLJ3	0009 0147 0169 0171

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS									
DLJ4	0009	0147	0159	0172						
DSIN	0059	0065	0065	0093	0113	0114	0122	0159		
DSQA	0175	0176	0180							
DSUP	0008	0009	0014	0183	0184	0186				
DTAN	0140									
DUM1	0005	0009								
DX12	0105	0108								
DX13	0107	0108								
ECA1	0009	0097	0139	0140	0142					
ELRV	0187									
GDIF	0002	0097	0187							
NARG	0009									
NKQ2	0009									
NMAX	0002	0097	0187							
NSEC	0009	0134								
NADV	0002	0187								
XKEP	0097									
ALLUT	0030	0031	0032	0054	0055	0056	0070	0071	0072	0129 0130 0131
ATANG	0108									
BRW4	0002									
CCLEF	0008	0009								
DADLM	0008	0147	0187							
CAJ25	0009	0170	0180							
CAPRL	0173	0180								
DASUM	0009	0012	0174	0176	0176	0177				
DATAN	0141									
DLJ25	0009	0147	0155	0170						
DLPHL	0148	0149	0145	0150	0150	0159	0158	0160	0161	0169 0173
DLSUM	0009	0147	0161	0161	0169	0174				
DSOAT	0075	0091	0140	0144	0145	0146				
NKQZD	0009									
NLONG	0009									
NSECD	0009	0165	0175							
NSUPP	0009	0181								
SCOFF	0008	0009								
STERM	0159	0160								
SUPPL	0182									
APRIME	0138	0143	0145	0147	0168					
ARGMCT	0008	0009								
ASPORT	0008	0009	0133							
DTIMES	0009									
EPRIME	0139	0168								
FPRIME	0141	0144	0147	0151	0159	0168				
GPRIME	0137	0147	0168							
KCCUNT	0009	0164	0178							
NBR#R1	0009									
NSHORT	0009									
NSUPPD	0009									
RPRIME	0143	0147	0168							

*****ORTRAN CROSS REFERENCE LISTING*****

REFERENCES

DEFINED

LABEL

9	0054	
10	0055	
11	0056	
15	0057	
16	0058	
17	0059	
18	0060	
19	0061	
20	0062	
21	0063	
22	0064	
23	0065	
24	0066	
25	0067	
26	0068	
28	0070	
29	0071	
30	0072	
31	0057	
33	0098	
34	0099	
35	0100	
36	0101	
37	0102	
38	0103	
40	0109	
41	0110	
42	0111	
43	0112	
44	0113	
45	0114	
46	0115	
47	0116	
48	0117	
49	0118	
50	0119	
51	0120	
52	0121	
53	0122	
54	0123	
55	0124	
56	0125	
57	0126	
58	0127	
59	0128	
60	0129	
61	0130	
62	0131	
63	0187	
621	0182	0181 0181
629	0186	0185
700	0014	0013
1000	0004	
1002	0036	0035

***** O P T R A N C R O S S R E F E R E N C E L I S T I N G *****

LABEL	JEFINEL	REFERENCES
1003	0040	CC39
1004	0046	CC45
1005	0050	CC49
1013	0042	0041
1014	0048	0047
1828	0162	0167 0168 0169
1850	0010	0180
1860	0123	0166
2000	0170	0164 0164 0165
2001	0106	0165 0165
2002	0165	0164
3000	0152	0154
3001	0153	0152 0152
3002	0155	0152 0157
3003	0156	0155
3004	0158	0155 0155
6200	0135	0134 0134
6210	0181	0134 0178 0178 0179
6220	0153	0132
6239	0179	0178
6300	0180	0179 0179


```

1SN 0008      7 RM(3)=28.000
1SN 0009      8 RM(4)=31.000
1SN 0010      9 RM(5)=30.000
1SN 0011     10 RM(6)=31.000
1SN 0012     11 RM(7)=30.000
1SN 0013     12 RM(8)=31.000
1SN 0014     13 RM(9)=31.000
1SN 0015     14 RM(10)=30.000
1SN 0016     15 RM(11)=31.000
1SN 0017     16 RM(12)=30.000

C
1SN 0018     17 Y=NY-1800
1SN 0019     18 YL=IDINT((Y-1.0001/4.000)
1SN 0020     19 YC=IDINT((Y+99.000)/100.000)-1
1SN 0021     20 RY=Y-YL
1SN 0022     21 DJUL=RY*365.000+YL*366.000-YC+2378495.500
1SN 0023     22 TD=ND
1SN 0024     23 DC 24 N=1,NM
1SN 0025     24 DJUL=DJUL+RM(N)
1SN 0026     25 IF (NM-2) 29,29,26
1SN 0027     26 IF (Y-100.000) 27,29,27
1SN 0028     27 IF (MAD (NY,4)) 29,28,29
1SN 0029     28 DJUL=DJUL+1.000
1SN 0030     29 DJUL=DJUL+TD
1SN 0031     RETURN
1SN 0032     END
00086970
00086980
00086990
00087000
00087010
00087020
00087030
00087040
00087050
00087060
00087070
00087080
00087090
00087100
00087110
00087120
00087130
00087140
00087150
00087160
00087170
00087180
00087190
00087200
00087210
00087220

```

***** D E T A I L C R O S S R E F E R E N C E L I S T I N G *****

SYMBOL	INTERNAL STATEMENT NUMBERS
N	0024 0025
Y	0016 0019 0020 0021 0027
NO	0002 0023
NV	0002 0024 0025
NY	0002 0018 0028
RM	0009 0006 0007 0008 0009 0010 0011 0012 0013 0014 0015 0016 0017 0025
RY	0021 0022
TD	0023 0030
YC	0020 0022
YL	0019 0021 0022
MAU	0028
DJUL	0002 0022 0025 0029 0029 0030 0030
DUINT	0019 0020

***** REFERENCE LISTING *****

REFERENCES

USING

LEVEL

1 0004
 2 0006
 3 0007
 4 0008
 5 0009
 6 0010
 7 0011
 8 0012
 9 0013
 10 0014
 11 0015
 12 0016
 13 0017
 14 0018
 15 0019
 16 0020
 17 0021
 18 0022
 19 0023
 20 0024
 21 0025
 22 0026
 23 0027
 24 0028
 25 0029
 26 0030
 27 0031
 28 0032
 29 0033

0024
 0026
 0027
 0028
 0029
 0030
 0031
 0032
 0033

LEVEL 16 (1 JULY 68)

DS/390 PORTMAN H

DATE 69.206/19.05.33

```

      COMPILER OPTICS - NAME= MAIN,OPT=00,LINE=NT=56,SURCE,SECTIC,NCLIST,NODECK,LOAD,MAP,NODEIT,10,XREF
      CSURROUTINE CMAD
      ISN 0002      REAL FUNCTION CMAD(X,Y)
      ISN 0003      IMPLICIT REAL*8(A-T,O-Z*)
      ISN 0004      IF(Y.EQ.0.0)GO TO 10
      ISN 0005      K=XY
      ISN 0006      CMAD=X-K*Y
      ISN 0007      RETURN
      ISN 0008      10 CMAD=CMAD*DDG
      ISN 0009      RETURN
      ISN 0010      END
      ISN 0011
      COB7230
      COB7240
      COB7250
      COB7260
      COB7270
      COB7280
      COB7290
      COB7300
      COB7310
      COB7320
```

 C R O S S R E F E R E N C E L I S T I N G *****

SYMBOL INTERNAL STATEMENT NUMBERS

K	0000 0007
X	0002 0006 0007
Y	0002 0004 0006 0007
DNAL	0002 0007 0009

*****ORTRAN CROSS REFERENCE LISTING*****

LAULL DEFINED REFERENCES
10 0009 0004

```

LEVEL 16 ( 1 JULY 68)                                OS/360  FORTRAN H                                DATE  69.206/19.05.35

      COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=59,SOURCE=EBDCIC,NCLIST,NODECK,LOAD,MAP,NOEDIT,IC,AREF
      CSUBROUTINE ELRV
      SUBROUTINE ELRV (X,VX,E,F,RADV,A,P,EN,GM,ERR,GDIF,NMAX,NIT)
      IMPLICIT REAL*8(A-H,O-Z,S)
      C
      C NAME CHANGED 11/12/68 FROM ELRVZ TO ELRV
      C
      C
      C VERSION OF 03/02/64
      C FORTRAN SUBROUTINE
      C
      C PURPOSE
      C CONVERTS OSCILLATING ORBITAL ELEMENTS INTO GEGCENTRIC
      C EQUATORIAL INERTIAL RECTANGULAR COORDINATES OF POSITION
      C AND VELOCITY.
      C
      C CALLING SEQUENCE
      C CALL ELRV (X,VX,E,F,RADV,A,P,EN,GM,ERR,GDIF,NMAX,NIT)
      C
      C INPUT
      C A(1) = SEMI-MAJOR AXIS
      C A(2) = ECCENTRICITY
      C A(3) = INCLINATION
      C A(4) = RIGHT ASCENSION OF ASCENDING NODE
      C A(5) = ARGUMENT OF PERIGEE
      C A(6) = MEAN ANOMALY
      C
      C GM = THE PRODUCT OF G, THE GAUSSIAN CONSTANT SQUARED,
      C AND M, THE MASS OF THE EARTH
      C ERR = TRUNCATION FACTOR REQUIRED IN XKEPZ FUNCTION
      C NMAX = MAXIMUM NUMBER OF ITERATIONS ALLOWED
      C IN SOLVING KEPLER'S EQUATION
      C IN RADIANS
      C
      C UNITS OF INPUT ARGUMENTS A(1) AND GM ARE ARBITRARY
      C BUT MUST BE MUTUALLY CONSISTENT.
      C INPUT ARGUMENTS MUST BE AVAILABLE IN CALLING PROGRAM IN
      C DOUBLE PRECISION FORM.
      C
      C OUTPUT
      C X(1)
      C X(2) THE 3 RECTANGULAR COORDINATES OF POSITION
      C X(3)
      C
      C VX(1)
      C VX(2) THE 3 RECTANGULAR COMPONENTS OF VELOCITY
      C VX(3)
      C E = ECCENTRIC ANOMALY
      C F = TRUE ANOMALY
      C RADV = RADIUS VECTOR
      C
      C P = ANOMALISTIC PERIOD
      C EN = MEAN ANGULAR MOTION
      C GDIF = LAST CORRECTION (E2-E1)/E2
      C
      C0089100
      C0089110
      C0089120
      C0089130
      C0089140
      C0089150
      C0089160
      C0089170
      C0089180
      C0089190
      C0089200
      C0089210
      C0089220
      C0089230
      C0089240
      C0089250
      C0089260
      C0089270
      C0089280
      C0089290
      C0089300
      C0089310
      C0089320
      C0089330
      C0089340
      C0089350
      C0089360
      C0089370
      C0089380
      C0089390
      C0089400
      C0089410
      C0089420
      C0089430
      C0089440
      C0089450
      C0089460
      C0089470
      C0089480
      C0089490
      C0089500
      C0089510
      C0089520
      C0089530
      C0089540
      C0089550
      C0089560
      C0089570
      C0089580
      C0089590
      C0089600
      C0089610
      C0089620
      C0089630

```


SYMBOL	INT.	VAL.	STATE	UNIT	0009	0010	0011	0020	0022	0032	0033	0035
A	0002	0005	0000	0017	0009	0009	0010	0011	0020	0022	0032	0033
B	0002	0012	0013	0013	0010							
C	0004	0014	0015	0015	0010	0010	0017	0017	0018	0018	0019	0019
D	0004	0022										
E	0004	0041	0042									
F	0004	0037	0038									
G	0004	0037	0038									
H	0004	0037	0038									
I	0004	0037	0038									
J	0004	0037	0038									
K	0004	0037	0038									
L	0004	0037	0038									
M	0004	0037	0038									
N	0004	0037	0038									
O	0004	0037	0038									
P	0004	0037	0038									
Q	0004	0037	0038									
R	0004	0037	0038									
S	0004	0037	0038									
T	0004	0037	0038									
U	0004	0037	0038									
V	0004	0037	0038									
W	0004	0037	0038									
X	0004	0037	0038									
Y	0004	0037	0038									
Z	0004	0037	0038									
AA	0004	0037	0038									
AB	0004	0037	0038									
AC	0004	0037	0038									
AD	0004	0037	0038									
AE	0004	0037	0038									
AF	0004	0037	0038									
AG	0004	0037	0038									
AH	0004	0037	0038									
AI	0004	0037	0038									
AJ	0004	0037	0038									
AK	0004	0037	0038									
AL	0004	0037	0038									
AM	0004	0037	0038									
AN	0004	0037	0038									
AO	0004	0037	0038									
AP	0004	0037	0038									
AQ	0004	0037	0038									
AR	0004	0037	0038									
AS	0004	0037	0038									
AT	0004	0037	0038									
AU	0004	0037	0038									
AV	0004	0037	0038									
AW	0004	0037	0038									
AX	0004	0037	0038									
AY	0004	0037	0038									
AZ	0004	0037	0038									
BA	0004	0037	0038									

***** O R T R A N C R U S S R E F E R E N C E L I S T I N G *****

***** REFERENCES

***** DEFINED

4	0012
7	0020
8	0021
5	0022
10	0024
11	0025
100	0004


```

C***** START PROGRAM *****00097560
C
ISN 0005
ISN 0006
ISN 0007
ISN 0008
ISN 0009
ISN 0010
ISN 0011

2 TH=1*
3 TM=1*
4 TM=DSIGN(TM,TH)
5 TRS=DSIGN(TS,TH)
6 HMSRZ=TH*.2617953677591494DC + TM*.363323129985824D-3
1 +TRS*.272205216683040D-5
RETURN
END
00097570
00097580
00097590
00097600
00097610
00097620
00097630
00097640
00097650

```

***** F O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

SYMBOL	INTERNAL STATEMENT NUMBERS
IM	0002 0005
IM	0002 0006
TH	0005 0007 CCCB 0009
TM	0006 0007 C007 0009
TS	0002 0008
TBS	0008 0009
DSIGN	0007 0008
HMSNZ	CCC2 0009

***** F O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

LABEL	DEFINED	REFERENCES
2	0005	
3	0006	
4	0007	
5	0008	
6	0009	
100	0004	

LEVEL 16 (1 JULY 68) OS/360 FORTRAN M DATE 69.206/19.05.42

COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECT=58,SOURCE=EBDCIG,NULIST,NODECK,LOAD,MAP,NOEDIT,LD,XREF

CSUBROUTINE JULCAL

SUBROUTINE JULCAL(DJ,NM,ND,NY)

IMPLICIT REAL*8(A-H,O-Z,S)

VERSION OF 07/22/63

FORTRAN SUBROUTINE

PURPOSE

COMPUTES CALENDAR DATE FROM JULIAN DATE AT 0 HOURS

UNIVERSAL TIME (OR 0 HOURS EPOCHERIS TIME).

CALLING SEQUENCE

CALL JULCAL(DJ,NM,ND,NY)

IN-UT

DJ = JULIAN DATE AT 0 HOURS UNIVERSAL TIME

OUTPUT

NM = CALENDAR MONTH

ND = CALENDAR DAY

NY = CALENDAR YEAR

REFERENCE

REFER TO MATHEMATICAL DESCRIPTION IN SUBPROGRAM WRITEUP

METHOD

THE NUMBER OF DAYS FROM 12 HOURS UNIVERSAL TIME JAN. 0.

1800 IS CALCULATED. THE INTEGRAL NUMBER OF YEARS IN THIS

NUMBER IS ADDED TO 1800 TO GIVE THE CURRENT CALENDAR YEAR

AND THE NUMBER OF DAYS CONTAINED IN THE INTEGRAL NUMBER

OF YEARS ELAPSED SINCE JAN. 0. 1800 IS SUBTRACTED FROM THE

ORIGINAL NUMBER OF DAYS FROM JAN. 0. 1800. THE INTEGRAL

NUMBER OF MONTHS IN THIS REMAINDER IS CALCULATED TO GIVE

THE CURRENT CALENDAR MONTH. THE NUMBER OF DAYS CONTAINED

IN THIS INTEGRAL NUMBER OF MONTHS IS SUBTRACTED FROM THE

PREVIOUS REMAINING DAYS TO GIVE THE CALENDAR DAY.

APPROPRIATE CONSIDERATION HAS BEEN GIVEN TO THOSE YEARS

WHICH ARE DIVISIBLE BY 4, 100, AND 400.

APPROPRIATE CONSIDERATION HAS BEEN GIVEN TO THOSE YEARS

WHICH ARE DIVISIBLE BY 4, 100, AND 400.

RESTRICTIONS

DATE RESTRICTED TO LIE BETWEEN JANUARY 1, 1801 AND DECEMBER

31, 2000.

100 CONTINUE

ACCURACY

REQUIRED SUBPROGRAMS

07/22/63 DJUL

NAD

C

C

C

C

C

C

C

C

C

C

C

C

C

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS	0011	0012	0013	0014	0015	0016	0017	0018	0029	0029	0050	0034
M	0005 0007 0008 0009 0010	CC11											
N	0025 0026 0029 0030	CC34											
DJ	0002 0019 0020 0022 0023												
ND	0002 0024 0030 0032	CC34	0034										
NM	0002 0006 0031 0031												
NY	0002 0020 0022 0023 0027 0028												
PAD	0028												
NDY	0023 0024												
DJUL	0023												
IDINT	0020 0022												
JULCAL	0002												

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

REFERENCES

DEF INEC

LABEL

0	0006	
6	0007	
9	0008	
10	0009	
11	0010	
12	0011	
13	0012	
14	0013	
15	0014	
16	0015	
17	0016	
18	0017	
19	0018	
20	0019	
21	0020	
22	0021	
23	0022	
24	0023	
25	0024	
26	0025	
27	0026	
28	0027	
29	0028	
30	0029	
31	0030	
32	0031	
33	0032	
34	0033	
100	0004	


```

ISN 0038      EMCOEF( 7,3)=40EC.D 00
ISN 0039      EMCOEF( 7,4)=12700.D 00
ISN 0040      EMCOEF( 7,5)=15750.D 00
ISN 0041      EMCOEF( 8,1)=5.D 00
ISN 0042      EMCOEF( 8,2)=95.D 00
ISN 0043      EMCOEF( 8,3)=375.C 00
ISN 0044      EMCOEF( 8,4)=525.C 00
ISN 0045      EMCOEF( 9,1)=1.D 00
ISN 0046      EMCOEF( 9,2)=1.D 00-2.D 00-E**2
ISN 0047      EMCOEF(10,1)=512.C 00+768.D 00-E**2
ISN 0048      EMCOEF(10,2)=944.D 00-1248.D 00-E**2
ISN 0049      EMCOEF(13,1)=4.D 00
ISN 0050      EMCOEF(13,2)=127.C 00
ISN 0051      EMCOEF(13,3)=48C.D 00
ISN 0052      EMCOEF(13,4)=525.D 00
ISN 0053      EMCOEF(14,1)=40.D 00
ISN 0054      EMCOEF(14,2)=135C.D 00
ISN 0055      EMCOEF(14,3)=1273C.D 00
ISN 0056      EMCOEF(14,4)=38050.D 00
ISN 0057      EMCOEF(14,5)=36750.D 00
ISN 0058      EMCOEF(15,1)=20.D 00
ISN 0059      EMCOEF(15,2)=315.D 00
ISN 0060      EMCOEF(15,3)=1160.D 00
ISN 0061      EMCOEF(15,4)=1225.D 00
ISN 0062      EMCOEF(16,1)=2.D 00-2.D 00-E**2
ISN 0063      EMCOEF(16,2)=2.D 00+3.D 00-E**2
ISN 0064      DO 102 I=1,16
ISN 0065      DC 199 J=1,10
ISN 0066      199 COEF(J)=EMCOEF(I,J)
ISN 0067      102 CALL POLVAL(4,CDEF,THSQ,EM(1))
ISN 0068      EM(1)=E*(1.D 0C-T*SQ)/EM0**2
ISN 0069      EM(5)=2.D 00-E**2/EM0
ISN 0070      EM(6)=EM(6)*EM(3)
ISN 0071      EM(8)=EM(8)*EM(4)
ISN 0072      EM(9)=EM(9)/E
ISN 0073      EM(11)=(4.D 00/3.D 00)*EM(2)/EM0**2
ISN 0074      EM(12)=E**2/EM0**3
ISN 0075      EM(13)=EM(13)*EM(3)
ISN 0076      EM(15)=EM(15)*EM(4)
ISN 0077      EM(16)=EM(16)*TH/DSORT(1.D 00-THSQ)
ISN 0078      IF(NKQZD)7001,7002,7001
ISN 0079      7001 WRITE(6,2)
ISN 0080      WRITE(6,3)
ISN 0081      WRITE(6,11)(EM(I),I=1,16)
ISN 0082      DE1=EM(3)+FJ(4)*EM(4)/FJ(2)
ISN 0083      IF(NKQZD)7003,7004,7003
ISN 0084      7003 WRITE(6,11)DE1
ISN 0085      CE1=DE1**2
ISN 0086      IF(NKQZD)7005,7006,7005
ISN 0087      7005 WRITE(6,11)DE1
ISN 0088      CE1=EM(2)*DE1
ISN 0089      IF(NKQZD)7007,7008,7007
ISN 0090      7007 WRITE(6,11)DE1
ISN 0091      DE2=FJ(4)/FJ(2)**2
ISN 0092      IF(NKQZD)7009,7010,7009
ISN 0093      7009 WRITE(6,11)DE2
00101850
00101860
00101870
00101880
00101890
00101900
00101910
00101920
00101930
00101940
00101950
00101960
00101970
00101980
00101990
00102000
00102010
00102020
00102030
00102040
00102050
00102060
00102070
00102080
00102090
00102100
00102110
00102120
00102130
00102140
00102150
00102160
00102170
00102180
00102190
00102200
00102210
00102220
00102230
00102240
00102250
00102260
00102270
00102280
00102290
00102300
00102310
00102320
00102330
00102340
00102350
00102360
00102370
00102380
00102390
00102400

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```

ISN 0094      7010 DE3=EM(41+EM(7)*DE2+EM(8)*DE2**2
ISN 0095      IF(NK0ZD17011,7012,7011
ISN 0096      7011 WRITE(6,1)DE3
ISN 0097      7012 DE=DE+EM(5)*DE3
ISN 0098      IF(NK0ZD17013,7014,7013
ISN 0099      7013 WRITE(6,1)DE
ISN 0100      7014 DE=EM(1)*DE
ISN 0101      IF(NK0ZD17015,7016,7015
ISN 0102      7015 WRITE(6,1)DE
ISN 0103      7016 DE=(FJ(21)**2/(1024.D 00*SEMA  **2**3))*CE
ISN 0104      IF(NK0ZD17017,7018,7017
ISN 0105      7017 WRITE(6,1)DE
ISN 0106      7018 DE=
ISN 0107      1 ((FJ(3)**2/FJ(21)**2)/(16.D 00*SEMA**P))*EM(9)
ISN 0108      IF(NK0ZD17019,7020,7019
ISN 0109      7019 WRITE(6,1)DE4
ISN 0110      7020 DE=DE*DE4
ISN 0111      7021 DE=EM(10)*EM(11)*EM(3)*EM(4)*FJ(4)/FJ(2)**2
ISN 0112      IF(NK0ZD17021,7022,7021
ISN 0113      7022 DE=EM(12)*EM(13)*EM(14)*FJ(4)/FJ(2)*EM(15)*(FJ(4)/FJ(2))**2
ISN 0114      IF(NK0ZD17023,7024,7023
ISN 0115      7023 WRITE(6,1)D12 ,A110(3)
ISN 0116      7024 D13=(FJ(21)**2/(2048.D 00*EPS**4))*DSIN(2.D 00*A110(3))
ISN 0117      IF(NK0ZD17025,7026,7025
ISN 0118      7025 WRITE(6,1)D13
ISN 0119      7026 C14=(FJ(31)/FJ(21))**2/(16.D 00*P**2)
ISN 0120      IF(NK0ZD17027,7028,7027
ISN 0121      7027 WRITE(6,1)D14
ISN 0122      7028 D14=D14*EM(16)
ISN 0123      IF(NK0ZD17029,7030,7029
ISN 0124      7029 WRITE(6,1)D14
ISN 0125      7030 D1=D13*(D11*D1214D14
ISN 0126      D1H=-DSIN(A110(31)*D1
ISN 0127      IF(NK0ZD17031,7032,7031
ISN 0128      7031 WRITE(6,1)(FJ(1),I=2,4),SEMA,P
ISN 0129      7032 WRITE(6,1)DE,D1,D1H
ISN 0130      GAMMA=0.75D 00*P*(-21)*EPS*(1.D 00-3.D 00*THSQ)
ISN 0131      B220=-15.D 00+16.D 00*EPS+25.D 00*EPS**2
ISN 0132      B21=30.D 00-96.D 00*EPS-90.D 00*EPS**2
ISN 0133      B22=105.D 00+144.D 00*EPS+25.D 00*EPS**2
ISN 0134      IF(NK0ZD17033,7034,7033
ISN 0135      7033 WRITE(6,1)GAMMA,B220,B221,B222
ISN 0136      7034 B2=B220*B221*THSQ*B222*THSQ**2
ISN 0137      IF(NK0ZD17035,7036,7035
ISN 0138      7035 WRITE(6,1)B22
ISN 0139      B22=(3.D 00/128.D 00)*EPS*B22
ISN 0140      IF(NK0ZD17037,7038,7037
ISN 0141      7037 WRITE(6,1)B22
ISN 0142      7038 B4=3.D 00-30.D 00*THSQ+35.D 00*THSQ**2
ISN 0143      IF(NK0ZD17039,7040,7039
ISN 0144      7039 WRITE(6,1)B4
ISN 0145      7040 B4=(65.D 00/128.D 00)*EPS*B4**2*B4
ISN 0146      IF(NK0ZD17041,7042,7041
ISN 0147      7041 WRITE(6,1)B4
ISN 0148      7042 B2=B22*P**(-4)

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15N 0149	B=E+P**(-4)	00102970
15N 0150	IF(NKZD)7043,7044,7045	00102980
15N 0151	7043 WRITE(6,1)B22,B4	00102990
15N 0152	7044 DN1=FJ(2)*J*(GAMMA**3-4.D CC*GAMMA*B22)	00103000
15N 0153	DN2=-4.D CC*B4*GAMMA*FJ(2)*FJ(4)	00103010
15N 0154	DN3=3.L 00*EPS*(-2)*FJ(2)*DE	00103020
15N 0155	DN4=(-6.D CC*TH/(1.D 00-3.D CC*THS0))*FJ(2)*DTH	00103030
15N 0156	DN0=DN1+DN2+DN3+DN4	00103040
15N 0157	DN=ENC*DN0	00103050
15N 0158	IF(NKZD)7045,7046,7047	00103060
15N 0159	7045 WRITE(6,1)DN1,DN2,ENJ,DN4,DN0,ENO,DN	00103070
15N 0160	7046 RETURN	00103080
15N 0161	END	00103090

SYMBOL	INTERNAL STATEMENT NUMBERS
A	0006 0008
B	0006 0008
C	0006 0008
D	0006 0008
E	0017 0018
F	0006 0008
G	0006 0008
H	0006 0008
I	0021 0023
J	0022 0023
K	0020 0103
L	0006 0008
M	0006 0008
N	0006 0008
O	0006 0008
P	0006 0008
Q	0006 0008
R	0006 0008
S	0006 0008
T	0006 0008
U	0006 0008
V	0006 0008
W	0006 0008
X	0006 0008
Y	0006 0008
Z	0006 0008
AA	0006 0008
AB	0006 0008
AC	0006 0008
AD	0006 0008
AE	0006 0008
AF	0006 0008
AG	0006 0008
AH	0006 0008
AI	0006 0008
AJ	0006 0008
AK	0006 0008
AL	0006 0008
AM	0006 0008
AN	0006 0008
AO	0006 0008
AP	0006 0008
AQ	0006 0008
AR	0006 0008
AS	0006 0008
AT	0006 0008
AU	0006 0008
AV	0006 0008
AW	0006 0008
AX	0006 0008
AY	0006 0008
AZ	0006 0008
BA	0142 0145 0146 0147 0149 0149 0151 0153
BB	0099 0100 0102 0103 0103 0105 0109 0129 0154
BC	0125 0125 0125 0125 0125 0125 0125 0125 0125
BD	0002 0157 0159
BE	0006 0008
BF	0004 0006 0009 0070 0070 0070 0071 007

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS
DN3	0154 0156 0159
DN4	0155 0156 0159
DTM	0126 0129 0155
DIE	0008
D11	0008
EM0	0025 0068 0073 0074
ENO	0008 0157 0159
EPS	0018 0130 0131 0132 0132 0133 0133 0139 0145 0154
ERR	0008
REM	0008 0012
ARGO	0007 0008
A110	0006 0008 0016 0017 0019 0115 0116 0126
B220	0131 0135 0136
B221	0132 0135 0136
B222	0133 0135 0136
CJ2S	0008
CDEF	0004 0066 0067
DAJ3	0008
DAJ4	0008
DCGS	0019
DLJ3	0008
DLJ4	0008
DN0	0159 0159
DSIN	0116 0126
DSUP	0007 0008
DUM1	0005 0008
DUM2	0005 0008
ECA1	0008
NARG	0008
NKOZ	0008
NSEC	0008
SEMA	0016
THSQ	0024
CCDEF	0007 0008
DAJ2S	0008
LASUM	0008
DLJ2S	0008
DLSUM	0008
DSORT	0018 0077
GAMA	0130 0135
MCUD	0008 0078 0083 0086 0089 0092 0095 0098 0101 0104 0107 0111 0114 0117 0120 0123 0127 0134 0137
NLONG	0140 0143 0146 0150 0158
NSECC	0008
NSUPP	0008
SCDEF	0007 0008
ARGMOT	0007 0008
ASCHT	0007 0008
DTIMES	0008
EMCUEF	0004 0023 0026 0027 0028 0029 0030 0031 0032 0033 0034 0035 0036 0037 0038 0039 0040 0041 0042
	0043 0044 0045 0046 0047 0048 0049 0050 0051 0052 0053 0054 0055 0056 0057 0058 0059 0060 0061
KCOUNT	0062 0063 0066
KCMAN	0008
KCMAN	0002

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS
NSRWR1	0008
NSHRT	0008
NSUPFD	0008
PCLVAL	0007

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

LABEL	DEFINED	REFERENCES
1	0009	0081 0084 0087 0090 0093 0096 0099 0102 0105 0108 0112 0115 0116 0121 0124 0128 0129
2	0010	0135 0138 0141 0144 0147 0151 0159
3	0011	0079
101	0023	0080 0022
102	0057	0021 0024
199	0066	0065
2001	0079	0078 0079
2002	0082	0078 0079
2003	0084	0078 0083
2004	0085	0083 0086
2005	0087	0086 0086
2006	0088	0086 0089
2007	0090	0089 0089
2008	0091	0089 0091
2009	0093	0092 0092
2010	0094	0092 0095
2011	0096	0095 0095
2012	0097	0095 0098
2013	0099	0098 0098
2014	0100	0098 0101
2015	0102	0101 0101
2016	0103	0101 0104
2017	0105	0104 0104
2018	0106	0104 0107
2019	0108	0107 0107
2020	0109	0107 0111
2021	0112	0111 0111
2022	0113	0111 0114
2023	0115	0114 0114
2024	0116	0114 0117
2025	0118	0117 0117
2026	0119	0117 0120
2027	0121	0120 0120
2028	0122	0120 0123
2029	0124	0123 0123
2030	0125	0123 0127
2031	0126	0127 0127
2032	0130	0127 0134
2033	0135	0134 0134
2034	0136	0134 0137
2035	0138	0137 0137
2036	0139	0137 0140
2037	0141	0140 0140
2038	0142	0140 0143
2039	0144	0143 0143
2040	0145	0143 0146
2041	0147	0146 0146
2042	0148	0146 0150
2043	0151	0150 0150
2044	0152	0150 0158
2045	0159	0158 0158
2046	0160	0158 0160

LEVEL 16 (1 JULY 68)	OS/360 FORTRAN H	DATE 69.206/19.05.47
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COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECT=58,SOURCE,ERCDIC,NOLIST,NODECK,LOAD,MAP,NOEDIT,ID,XREF
CSUBROUTINE MAD
  ISN 0002 FUNCTION MAD(I,J)
  ISN 0003 IMPLICIT REAL*(A-H,O-Z,$)
  ISN 0004 IF (J.EQ.0) GO TO 1C
  ISN 0006 K=I/J
  ISN 0007 MAC=I-K*J
  ISN 0008 RETURN
  ISN 0009 10 MAC=0
  ISN 0010 RETURN
  ISN 0011 ENC
00109960
00109970
00109980
00109990
00110000
00110010
00110020
00110030
00110040
00110050

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***** D H T R A N C R O S S R E F E R E N C E L I S T I N G *****

SYMBOL INTERNAL STATEMENT NUMBERS
 I 0002 0006 0007
 J 0002 0004 0006 0007
 K 0006 0007
 MAD 0004 0007 0009

***** D E T R A N C R O S S R E F E R E N C E L I S T I N G *****

LABEL DEFINED REFERENCES
10 0009 0004

134

***** U P T M A N C O U S S R E F E R E N C E L I S T I N G *****

SYMBOL	INTERNAL STATEMENT NUMBERS
A	0002 0003 0011 0012 0014 0019 0024 0036
E	0034 0036
F	0034 0036
N	0013 0014 0016 0017 0018 0019 0021 0022 0022 0023 0023 0028 0029 0030 0037 0038 0038
AB	0007 0008 0034
AI	0002 0003 0011 0012 0014 0017 0018 0022 0023 0026 0027 0029
BA	0024 0030
CA	0002 0003 0031 0036
CT	0015 0024
EN	0004 0019 0024
GM	0005 0017 0019 0023 0024
RA	0005 0018 0019
XA	0007 0008 0036
MIT	0004 0034 0036
PER	0013 0024
TOH	0004 0023
Allo	0006 0008 0026 0027 0030
DUM1	0006 0008
GDIF	0004 0034 0036
NMAX	0002 0034 0036
PANA	0002
RAOV	0034 0036
REML	0004 0022 0023
ENR-1	0031
ENR-2	0034
ENR-4	0026
INPUT	0002 0010
RVELZ	0015 0024
ASHOUT	0002 0023

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

LABEL	DEFINED	REFERENCES	
		1	2
1	0011	0010	
2	0016	0010	
3	0021	0010	
4	0026	0010	
100	0004		
101	0014	0013	
201	0018	0016	
301	0023	0021	
302	0010		
401	0030	0028	
402	0038	0037	
410	0034	0033	0033
411	0036	0033	
412	0037	0035	
9999	0039	0015	0020 0025

LEVEL 16 (1 JULY 68)		OS/360 FORTRAN H	DATE 69-212/13-36-11
COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=58,SOURCE,EBCDIC,NOLIST,NODECK,LOAD,MAP,NOEDIT,ID,XREF			
TSN 0002	CSUBROUTINE POLVAL		00137200
TSN 0003	SURROUTINE POLVAL(N,A,X,Y)		00137210
TSN 0004	IMPLICIT REAL*8(A-H,O-Z,S)		00137220
TSN 0005	DIMENSION A(10)		00137230
TSN 0006	1 FORMAT(14H N NEGATIVE : IS)		00137240
TSN 0007	2 FORMAT(21H N GREATER THAN 9 = IS)		00137250
TSN 0008	21 FORMAT(1X,15)		00137260
TSN 0009	22 FORMAT(1X,213,3D15.8)		00137270
TSN 0010	IF(N) 3,4,4		00137280
TSN 0011	3 WRITE(6,11N		00137290
TSN 0012	STOP		00137300
TSN 0013	4 IF(N-9) 6,6.5		00137310
TSN 0014	5 WRITE(6,21N		00137320
TSN 0015	STOP		00137330
TSN 0016	6 K = N + 1		00137340
TSN 0017	CALL SSWTCH(1,K000FX)		00137350
TSN 0018	GO TO(7-8)-K000FX		00137360
TSN 0019	7 WRITE (6,21)K		00137370
TSN 0020	8 Y = 0.000		00137380
TSN 0021	DO 11 I = 1,K		00137390
TSN 0022	L = K - I + 1		00137400
TSN 0023	Y = X * Y + A (L)		00137410
TSN 0024	CALL SSWTCH(1,K00CFX)		00137420
TSN 0025	GO TO(10,11)-K00CFX		00137430
TSN 0026	10 WRITE (6,22)1,L,A(L),X,Y		00137440
TSN 0027	11 CONTINUE		00137450
TSN 0028	12 RETURN		00137460
	END		00137470

***** O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

SYMBOL	INTERNAL STATEMENT NUMBERS
A	0002 0004 0022 0025
T	0020 0021 0025
K	0015 0018 0020 0021
L	0021 0022 0025 0025
N	0002 0009 0010 0012 0013 0015
X	0002 0022 0025
Y	0002 0019 0022 0022 0025
K000PK	0016 0017 0023 0024
POLVAL	0002
SSWITCH	0016 0023

***** O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

LABEL	DEFINED	REFERENCES
1	0005	0010
2	0006	0013
3	0010	0009
4	0012	0009 0009
5	0013	0012
6	0015	0012 0012
7	0018	0017
8	0019	0017
9	0022	
10	0025	0024
11	0026	0020 0024
12	0027	
21	0007	0018
22	0008	0025

LEVEL 16 (1 JULY 68) OS/360 FORTRAN H DATE 69-206/19-05-51

COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECT=58,SOURCE,EBCCIC,NOLIST,NODECK,LOAD,MAP,NOEDIT,ID,XREF

CSUBROUTINE RMSZ (TR, ID, IM, TS) 00142440

ISN 0002 SUPEROUTINE RMSZ (TR, ID, IM, TS) 00142450

ISN 0003 IMPLICIT REAL*8 (A-H, O-Z, \$) 00142460

C VERSION OF 03/02/44 00142470

C FORTRAN SUPEROUTINE 00142480

C 00142490

C 00142500

C PURPOSE 00142510

C CONVERTS TIME IN RADIAN (24 HOURS = 2 PI RADIAN) INTO 00142520

C THE INTEGRAL NUMBER OF DAYS, NUMBER OF HOURS, NUMBER OF 00142530

C MINUTES, AND NUMBER OF SECONDS AND DECIMALS OF A SECOND 00142540

C CONTAINED IN THE TIME. 00142550

C 00142560

C CALLING SEQUENCE 00142570

C D CALL RMSZ (TR, ID, IM, TS) 00142580

C 00142590

C INPUT TR = TIME IN RADIAN 00142600

C 00142610

C TIME MUST BE AVAILABLE IN CALLING PROGRAM IN 00142620

C DOUBLE PRECISION FORM. 00142630

C 00142640

C 00142650

C OUTPUT ID = INTEGRAL NUMBER OF DAYS CONTAINED IN THE TIME 00142660

C IM = NUMBER OF HOURS 00142670

C IM = NUMBER OF MINUTES 00142680

C TS = NUMBER OF SECONDS AND DECIMALS OF A SECOND 00142690

C 00142700

C REFERENCE 00142710

C ***** 00142720

C 00142730

C METHOD ***** 00142740

C 00142750

C RESTRICTIONS 00142760

C ***** 00142770

C 00142780

C 00142790

C 00142800

C ACCURACY 00142810

C CONVERSION IS ACCURATE TO AT LEAST .001 SECONDS OF TIME. 00142820

C 00142830

C 00142840

C REQUIRED SUBPROGRAMS 00142850

C NONE 00142860

C 00142870

C 00142880

C 100 CONTINUE 00142890

C TIMING NO ESTIMATE AVAILABLE 00142900

C 00142910

C ANALYSIS 00142920

C 00142930

C 00142940

C 00142950

C ***** START PROGRAM ***** 00142960

C 00142970

1SN 0005	5	ID=TR/6.2831853C7175E86D0	00142980
1SN 0006		GOR=DFLOAT(ID)	00142990
1SN 0007		TS=TS-GOR*6.283185307179586D0	00143000
1SN 0008	7	IN=TR*3.81571863420548D0	00143010
1SN 0009	8	TI=IM	00143020
1SN 0010	9	A=TI/3.81971863420548D0	00143030
1SN 0011	10	B=TR-A	00143040
1SN 0012	11	IM=B*229.183118052329D0	00143050
1SN 0013	12	TS=IM	00143060
1SN 0014	13	IM=IABS(IM)	00143070
1SN 0015	14	C=T2/229.183118052329D0	00143080
1SN 0016	15	D=E-C	00143090
1SN 0017	16	TS=CAES(D*13750.9E70831397D0)	00143100
1SN 0018	27	IF=DABS(TS)-.0005C0) 28.17.17	00143110
1SN 0019	28	TS=C*CC	00143120
1SN 0020	17	IF (TS-S9.9995D0) 20.18.18	00143130
1SN 0021	18	TS=DABS(TS-C0.DC)	00143140
1SN 0022	19	IM=IM+1	00143150
1SN 0023		GO TO 27	00143160
1SN 0024	20	IF (IM-C0) 23.21.21	00143170
1SN 0025	21	IM=IM-C0	00143180
1SN 0026	22	IM=IM+ISIGN(1,IM)	00143190
1SN 0027	23	IF (IABS(IM)-24) 26.24.24	00143200
1SN 0028	24	IM=IM-ISIGN(24,IM)	00143210
1SN 0029	25	ID=ID+ISIGN(1,ID)	00143220
1SN 0030	26	RETURN	00143230
1SN 0031		END	00143240

*****ORTRAN CROSS REFERENCE LISTING*****

SYMBOL	INTERNAL STATEMENT NUMBERS
A	0010 0011
B	0011 0012 0016
C	0015 0016
D	0016 0017
ID	0002 0005 0006 0029 0029 0029
IM	0002 0008 0009 0026 0026 0028
IR	0002 0012 0013 0014 0014 0022
TR	0002 0005 0007 0007 0008 0011
IS	0002 0017 0018 0019 0020 0021
T1	0009 0010
T2	0013 0015
GDR	0006 0007
CAB	0017 0018 0021
IAES	0014 0027
ISIGN	0026 0028 0029
RHMSZ	0002
CFLCAT	0006

***** F O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

LABEL	DEFINED	REFERENCES
5	0005	
7	0007	
8	0008	
9	0009	
10	0010	
11	0011	
12	0012	
13	0013	
14	0014	
15	0015	
16	0016	
17	0017	
18	0018	CC18 C018
19	0019	0020 002C
20	0020	
21	0021	
22	0022	
23	0023	
24	0024	0024 0024
25	0025	
26	0026	
27	0027	0024 0027
28	0028	
29	0029	
30	0030	0027
31	0031	0023
32	0032	
33	0033	
34	0034	
35	0035	
36	0036	
37	0037	
38	0038	
39	0039	
40	0040	
41	0041	
42	0042	
43	0043	
44	0044	
45	0045	
46	0046	
47	0047	
48	0048	
49	0049	
50	0050	
51	0051	
52	0052	
53	0053	
54	0054	
55	0055	
56	0056	
57	0057	
58	0058	
59	0059	
60	0060	
61	0061	
62	0062	
63	0063	
64	0064	
65	0065	
66	0066	
67	0067	
68	0068	
69	0069	
70	0070	
71	0071	
72	0072	
73	0073	
74	0074	
75	0075	
76	0076	
77	0077	
78	0078	
79	0079	
80	0080	
81	0081	
82	0082	
83	0083	
84	0084	
85	0085	
86	0086	
87	0087	
88	0088	
89	0089	
90	0090	
91	0091	
92	0092	
93	0093	
94	0094	
95	0095	
96	0096	
97	0097	
98	0098	
99	0099	
100	0100	

00145160
00145170
00145180
00145190
00145200
00145210
00145220
00145230
00145240
00145250
00145260
00145270
00145280
00145290
00145300

SE=AA*SL*DSORT(1,CC0-E2)
CF=AA*(CE-A(2))
30 A(2)=ATANZ(SI,C1)
31 A(4)=ATANZ(SN,CN)
32 U=ATANZ(SU,CU)
F=ATANZ(SF,CF)
33 E=ATANZ(SE,CE)
ASU=F
A(5)=ALLOT (A5)
34 AG=E-A(2)*SE
35 A(6)=ALLOT (A6)
36 EN=HTGM/RTA*(1)
37 P#6.28318530717956CD0/EN
RETURN
ENC

1SN 0037
1SN 0038
1SN 0039
1SN 0040
1SN 0041
1SN 0042
1SN 0043
1SN 0044
1SN 0045
1SN 0046
1SN 0047
1SN 0048
1SN 0049
1SN 0050
1SN 0051

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS															
A	C002	C005	C022	C023	C025	C026	C028	C035	C036	C038	C039	C040	C045	C046	C047	C048
E	C043	C046														
F	C042	C044														
H	C019	C029	C030	C033												
N	C006	C007	C007	C008	C008											
P	C002	C049														
R	C010	C020	C022	C022	C025	C026										
U	C041	C044														
X	C002	C005	C007													
Y	C005	C007	C009	C009	C013	C013	C013	C013	C014	C014	C015	C015	C016	C016	C033	C034
AR	C025	C037	C038													
A5	C044	C045														
AB	C046	C047														
CE	C036	C038	C042													
CF	C038	C042														
CI	C030	C039														
CN	C032	C040														
CU	C034	C041														
C1	C016	C020	C029	C031	C032											
C2	C017	C018	C019													
EN	C002	C048	C049													
E2	C027	C028	C037													
F1	C021	C027	C035													
F2	C026	C027	C036													
GM	C002	C021	C022	C022	C022											
M1	C014	C017	C031	C034												
M2	C015	C017	C032	C034												
M3	C019	C019	C030													
R2	C009	C010														
SE	C035	C037	C043	C046												
SF	C037	C042														
SI	C029	C035														
SN	C031	C040														
SU	C033	C041														
VX	C002	C005	C008													
VY	C005	C008	C011	C011	C013	C013	C013	C013	C014	C014	C015	C015	C016	C016		
V1	C012															
V2	C011	C012	C022													
RC1	C020	C033	C034													
RIA	C023	C024	C048													
RIGM	C021	C024	C049													
ALLUT	C045	C047														
ATAN2	C039	C040	C041	C042	C043											
DSORT	C010	C012	C018	C019	C021	C023	C028	C037								
RLUT	C013	C024														
RVEL2	C002															

***** O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

DEFINED REFERENCES

LABEL	DEFINED
3	0010
4	0011
5	0012
6	0013
7	0014
8	0015
9	0016
10	0017
11	0018
12	0019
13	0020
14	0021
15	0022
16	0023
17	0024
18	0025
19	0026
20	0027
21	0028
22	0029
23	0030
24	0031
25	0032
26	0033
27	0034
28	0035
29	0036
30	0039
31	0040
32	0041
33	0043
34	0046
35	0047
36	0048
37	0049
100	0054
500	0056
501	0057
502	0058
	0059


```

LEVEL 16 ( 1 JULY 68)                                05/360  FORTRAN H                                DATE 69.212/13.36.14

COMPILED OPTIONS - NAME= MAIN,OPT=00.LINECNT=58.SOURCE=EBCDIC,NOLIST,NODECK,LOAD,NAP,NODEBIT,ID,XREF
CSUBROUTINE SWITCH
SURROUTINE SWITCH(N1,NSW)
C
C
C
TSN 0002
C
C
TSN 0003      IMPLICIT REAL*(A-H,O-Z)
TSN 0004      DIMENSION NSET(6)
TSN 0005      IF(N1.GT.6)GO TO 50
TSN 0007      NSW=NSET(N1)
TSN 0008      GO TO 100
TSN 0009      50 NSET(N1-6)=NSW
TSN 0010      IF(NSET(N1-6).EQ.0)NSET(N1-6)=2
TSN 0012      100 RETURN
TSN 0013      END

```

***** O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

SYMBOL	INTERNAL STATEMENT NUMBERS
NI	0002 0005 0027 0009 0010 0010
NSW	0002 0007 0009
NSET	0004 0007 0009 0010 0010
SSWITCH	0002

***** O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

LABEL	DEFINITION	REFERENCES
50	0009	0005
100	0012	0008

COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=58,SOURCE,EBCDIC,NOLIST,NODECK,LOAD,MAP,NOEDIT,ID,XREF
CSUBROUTINE SUPPO

ISN 0002

ISN 0003

ISN 0004

SUBROUTINE SUPPO

IMPLICIT REAL*8(A-H,O-Z,S)

COMMON CUM1,A110,GP,ERRB,XX,AB,VB,VB,XXX,SS,EL,CUM2,DUMF,

1CCNV,AICH,AG,D1,D2,D3,CMU,T0,CJ0,DTIMES,ARG0,ARGMDT,ARG,CCDEF,

2SCDEF,LSUP,CJ25,CJ3,CJ4,DLJ25,DLJ3,DLJ4,DLSUM,DAJ25,DAJ3,DAJ4,

3IASUM,REM,ASHORT,

4 NBRWRI,NLUNG,NSHURT,NSEC,NSECD,NSUPP,NSUPPC,NARG,KCOUNT,0146770

5,NK0Z,NK0Z0

DIMENSION DUM1(100),A110(6),GP(5),XX(12),AB(6),VB(3),VB(3),

1XX(10),SS(3),EL(13),DUM2(85),DUMF(9),AICH(24),AG(6),DI(6),D2(6),

2C3(6),TLC(99),ARGC(99),ARGMDT(99),ANG(99),CCDEF(99,6),SCDEF(99,6),

JUSUP(6),ASHORT(6)

1 FORMAT(16)

2 FORMAT(D24.17,F10.6,F14.8/G14.10,G12.9,G13.8,3G11.6/

1G14.10,G12.9,G13.8,3G11.6)

3 FORMAT(62X,9HARGUMENTS/39X,53HEPOCH,VALUES IN DEGREES AT EPOCH,AND000146860

1 MULTIONS PER HOUR////)

4 FORMAT(34X,13.5X,D24.17,5X,F10.6,3X,F14.8)

5 FORMAT(58X,19HCOSINE COEFFICIENTS//)

6 FORMAT(58X,17HSINE COEFFICIENTS//)

7 FORMAT(141.52X,27HSUPPLEMENTARY PERTURBATIONS////)

8 FORMAT(18X,

1 13.5X,F14.10,3X,F12.9,3X,F13.8,3X,3(F11.6,3X))

9 FORMAT(72X)

CARDS 81-8J SUPPLEMENTARY PERTURBATIONS INFORMATION CARDS.

NOTE: THESE CARDS ARE REQUIRED IF AND ONLY IF COLUMNS 34-3600146960

OF CARD NO. 2 (THE RUN CONTROL CARD) CONTAIN A QUANTI-00146970

TY OTHER THAN 0.

CARD 81 MAXIMUM NUMBER OF ARGUMENTS IS 99.

COLUMNS 1-6 NUMBER OF ARGUMENTS OCCURRING IN SUPPLEMENTARY

PERTURBATIONS.

EACH ARGUMENT REQUIRES 3 ADDITIONAL CARDS. THESE00147040

3 ADDITIONAL CARDS ARE 8J1,8J2,8J3.

00147050

00147060

00147070

00147080

00147090

00147100

00147110

00147120

00147130

00147140

00147150

00147160

00147170

00147180

00147190

00147200

00147210

00147220

00147230

00147240

00147250

00147260

00147270

00147280

00147290

00147300

00147310

00147320

00147330

00147340

00147350

00147360

00147370

00147380

***** U R T R A N C P O S S R E F E R E N C E L I S T I N G *****

SYMBOL	INTERNAL STATEMENT NUMBERS	0010	0017	0017	0017	0017	0020	0021	0022	0022	0022	0025	0026	0027	0030	0031	0032	0033	0034
I	0010	0017	0017	0017	0017	0017	0020	0021	0022	0022	0022	0025	0026	0027	0030	0031	0032	0033	0034
J	0034	0035	0035																
AB	0017	0017	0017	0017	0017	0017	0027	0027	0027	0027	0032	0032							
AG	0004	0005																	
AI	0004	0005																	
AI	0004	0005																	
D2	0004	0005																	
D3	0004	0005																	
EL	0004	0005																	
GP	0004	0005																	
II	0021	0022	0026	0027	0031	0032													
SS	0004	0005																	
TO	0004	0005	0017	0022															
XX	0004	0005																	
ARG	0004	0005																	
CJ3	0004																		
CJ4	0004																		
CMU	0004																		
DJ0	0004																		
REM	0004																		
RXB	0004	0005																	
VAB	0004	0005																	
XXX	0004	0005																	
AIH	0004	0005																	
ARG0	0004	0005	0017	0022	0034	0034													
AIIC	0004	0005																	
CJ25	0004																		
CCNV	0004	0034	0035																
LAJ3	0004																		
LAJ4	0004																		
DLJ3	0004																		
DLJ4	0004	0005																	
DSUF	0004	0005																	
DUMF	0004	0005																	
DUM1	0004	0005																	
DUN2	0004	0005																	
ERRB	0004																		
NARG	0004	0015	0016	0020	0025	0030	0033												
NKZ	0004																		
NSEC	0004																		
CCDF	0004	0005	0017	0027															
LAJ25	0004																		
DASUM	0004																		
DLJ25	0004																		
DLJ25	0004																		
ULSUM	0004																		
NKZC	0004																		
NLUNG	0004																		
NSEC	0004																		
NSUPP	0004																		
SCDEF	0004	0005	0017	0032															
SUPPO	0004																		
ARGMOT	0004	0005	0017	0022	0035	0035													
ASHORT	0004																		
LTIMES	0004																		

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL INTERNAL STATEMENT NUMBERS

KCOUNT C004
NGRWRI C004
NSHRT C004
NSUPED C004

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

LABEL	DEFINED	REFERENCES
1	0006	0015
2	0007	0017
3	0008	0019
4	0009	0022
5	0010	0024
6	0011	0029
7	0012	0018 0023 0028
8	0013	0027 0032
9	0014	
1001	0017	0016
1002	0022	0020
1003	0027	0025
1004	0032	0030
1005	0035	0033

***** O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

SYMBOL	INTERNAL STATEMENT NUMBERS
I	C012 0013 0014 0015 C015 C015 0016 0017 0018 0020 0020 0020 0020
J	C008 C009 C019 0020 C020 C020 C020
L	C023 C023 C023
AB	C004 C005
AG	C004 C005
C1	C004 C005
C2	C004 C005
DJ	C004 C005
EL	C004 C005
GP	C004 C005
II	C013 C023
SS	C004 C005
TC	C004 C005 C014
XX	C004 C005
ARG	C004 C005 0015 0016 C017 C018 0020 0020
CJ3	C004
CJ4	C004
CHU	C004
DJ0	C004 0014
REM	C004
RXB	C004
TAU	C014 0015
VNE	C004 C005
XX	C004 C005
AICH	C004 C005
ARGO	C004 C005 0015
A110	C004 C005
CJ2S	C004
CONV	C004 0018
CAJ3	C004
CAJ4	C004
ECES	0020
DLJ3	C004
DLJ4	C004
DSIN	0020
DSUP	C004
DUMF	C004 C005
DUM1	C004 C005
DUM2	C004 C005
ERRB	C004
NARG	C004 C012
NKOZ	C004
NSEC	C004
ALLCT	C017
ARGMT	0016
ARGUM	C018 0023
CCCEF	C004 C005 C020
CAJ2S	C004
CASUM	C004
DLJ2S	C004
DLSUM	C004
AKCZD	C004
NLONG	C004
NSECD	C004

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS
NSUPP	0004
SCCEF	0004 0005 0020
SUPPL	0002
ARGMT	0004 0005 0015
ASHORT	0004 0005
DT: PES	0004 0014
KCUNT	0004 0010 0021 0026
NBR#R1	0004
ASHORT	0004
NSUPPD	0004 0022

***** C H O U S R E F E R E N C E L I S T I N C *****

LABEL	DEFINED	REFERENCES
1	0006	0023
101C	0009	0008
1011	0011	0010
1012	0012	0010 001C
1027	0022	0021
1086	0020	0019
1089	0023	0022 0022
1090	0024	0012 0021 0021 0022
1091	0027	0026
1092	0029	0026 0026
1830	0025	0027 0028
1853	0007	0011

LEVEL 16 (1 JULY 68) OS/360 PORTMAN H DATE 69.206/19.06.00

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CCPILER OPTIONS - NAME= MAIN,DPT=00,LINECNT=58,SOURCE=ERCCIC,NOLIST,NODECK,LOAD=MAP,NODEIT,10,XREF
ISN 0002 CSUBROUTINE TIMC4
ISN 0003 IMPLICIT REAL*8(A-H,O-Z,S)

C
C VERSION OF 7/22/63
C
C READS A CARD FROM CARD READER CONTAINING CALENDAR DATE AND UT2 OF
C DESIRED START AND END TIMES FOR CALCULATION OF AN EPHEMERIS, AND
C THE TIME INCREMENT OF THE EPHEMERIS IN SECONDS. CALCULATES TIME
C INTERVAL IN SECONDS FROM SOME EPOCH (CJO,TSEP) TO THE START AND
C END TIMES.
C
C WRITES CALENDAR DATE AND UT2 ON TAPE UNIT A3.
C
C CALLING SEQUENCE
C CALL TIMC4(CJO,TSEP,S,F,UT,KPR,KLAST)
C
C INPUT FROM CALLING SEQUENCE
C DJJ = EPOCH JULIAN DATE AT 0 HOURS UT2.
C TSEP = EPOCH UT2 IN SECONDS
C
C INPUT FROM CARD READER
C NMS,NDS,NYS = MONTH, DAY, YEAR OF START DATE
C NMF,NDF,NYF = MONTH, DAY, AND YEAR OF END DATE
C
C NMS,NMS,NYS = HOUR, MINUTE, SECOND (UT2) OF START TIME
C NMF,NMF,NYF = HOUR, MINUTE, SECOND (UT2) OF END DATE
C DT = TIME INCREMENT OF EPHEMERIS IN SECONDS
C KLAST = INDICATION WHETHER ANOTHER EPHEMERIS IS TO BE
C PRINTED OR NOT
C KPR = FREQUENCY OF PRINTING.
C
C OUTPUT
C S = TIME IN SECONDS FROM EPOCH TO START TIME
C F = TIME IN SECONDS FROM EPOCH TO END TIME
C DT = TIME INCREMENT OF EPHEMERIS IN SECONDS
C
C REQUIRED SUBPROGRAMS
C
C DJJL
C
ISN 0004 6015 FORMAT (2(I1X12),1X,I4,2I3,F7.3,I3,1X,I4,2I3,F7.3,F11.3,I6,
ISN 0005 1I7)
ISN 0006 6016 FORMAT (1X12,1M/12,1M/14,7X11H START DATE/1X12,13,F7.3,5X,
ISN 0007 1 15H START TIME-ET /1X12,3,5X23H TIME INCREMENT-SECONDS//1X12,
ISN 0008 2 1M/12,1M/14,7X5H END DATE/1X12,13,F7.3,5X13H END TIME-ET ///)
ISN 0009 6017 FORMAT(4X,43HDATA FOR EVERY DATE ARE PRINTED ON SYSOUT=A///)
ISN 0010 6018 FORMAT(38X,14HDATA FOR EVERY,3X,16,3X,31HNC DATE ARE PRINTED ON SY00148470
ISN 0011 15OUT=A///)
ISN 0012 6019 FORMAT(///41X,50H ANOTHER EPHEMERIS IS COMPUTED AFTER CURRENT ONC00148490
ISN 0013 1E///)
ISN 0014 6020 FORMAT(///41X,50H NO EPHEMERIS IS COMPUTED AFTER CURRENT ONC00148510
ISN 0015 1E///)
ISN 0016 6021 FORMAT(38X,14HDATA FOR EVERY,3X,16,3X,31HNC DATE ARE PRINTED ON SY00148530
ISN 0017 15OUT=A///)

```

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ISN 0011      6022 FORMAT(3X,14HDATA FOR EVERY,3X,16,3X,31HRD DATE ARE PRINTED ON SYC0148550
              ISOUT=A,/)
C              CAHC NO. 9 EPHEMERIS REQUEST CARD (ONE EPHEMERIS REQUEST CARD IS
C              REQUIRED FOR EACH EPHEMERIS TO BE COMPUTED)
C              COLUMN 1 BLANK
C              COLUMNS 2- 3 MONTH OF START TIME
C              COLUMN 4 BLANK
C              COLUMNS 5- 6 DAY OF START TIME
C              COLUMN 7 BLANK
C              COLUMNS 8-11 YEAR OF START TIME
C              COLUMN 12 BLANK
C              COLUMNS 13-14 HOUR OF START TIME
C              COLUMN 15 BLANK
C              COLUMNS 16-17 MINUTES OF START TIME
C              COLUMN 18 BLANK
C              COLUMNS 19-24 SECONDS OF START TIME TO THOUSANDTHS
C              DECIMAL POINT IN COLUMN 21
C              COLUMN 25 BLANK
C              COLUMNS 26-27 MONTH OF END TIME
C              COLUMN 28 BLANK
C              COLUMNS 29-30 DAY OF END TIME
C              COLUMN 31 BLANK
C              COLUMNS 32-35 YEAR OF END TIME
C              COLUMN 36 BLANK
C              COLUMNS 37-38 HOUR OF END TIME
C              COLUMN 39 BLANK
C              COLUMNS 40-41 MINUTES OF END TIME
C              COLUMN 42 BLANK
C              COLUMNS 43-48 SECONDS OF END TIME TO THOUSANDTHS
C              DECIMAL POINT IN COLUMN 45
C              COLUMNS 49-59 INCREMENT OF TIME IN SECONDS AND THOUSANDTHS OF
C              SECONDS. (DECIMAL POINT IN COLUMN 56)
C              COLUMNS 60-65 FREQUENCY OF DATA IN EPHEMERIS INDICATOR
C              0 OR A NEGATIVE INTEGER = DATA FOR EVERY DATE WILL
C              BE PRINTED.
C              A POSITIVE INTEGER GREATER THAN 1(CALL THIS INTEGER00148900
C              KPR) = DATA FOR EVERY KPR,TH DATE WILL BE PRINTED.
C              COLUMNS 66-72 EPHEMERIS REQUEST SENTINEL.
C              0= THIS IS THE FINAL EPHEMERIS REQUEST
C              OTHER THAN ZERO = THIS IS NOT THE FINAL EPHEMERIS
C              REQUEST.
C              READ
C              0015,NMS,ADS,NYS,NMS,NMS,TSS,NMF,NDF,NYF,NHF,
C              1 NMNF,TSP,DT,KPR,KLAST
C              1 PRINT 6016,NMS,ADS,NYS,NMS,NMS,TSS,DT,NMF,NDF,NYF,NMF,NMF,TSP
C              IF(KPR-1)3,3,30
C              3 WRITE(6,6017)
C              GO TO 5
C              30 IF(KPR-2)31,31,3,
C              31 WRITE(6,6021)KPR
C              GO TO 5
C              32 IF(KPR-3)33,33,4
C              33 WRITE(6,6022)KPR
C              GO TO 5
C              4 WRITE(6,6018)KPR
C              5 IF(KLAST)7,6,7
C              6 WRITE(6,6020)
ISN 0012
ISN 0013
ISN 0014
ISN 0015
ISN 0016
ISN 0017
ISN 0018
ISN 0019
ISN 0020
ISN 0021
ISN 0022
ISN 0023
ISN 0024
ISN 0025
00148550
00148560
00148570
00148580
00148590
00148600
00148610
00148620
00148630
00148640
00148650
00148660
00148670
00148680
00148690
00148700
00148710
00148720
00148730
00148740
00148750
00148760
00148770
00148780
00148790
00148800
00148810
00148820
00148830
00148840
00148850
00148860
00148870
00148880
00148890
00148900
00148910
00148920
00148930
00148940
00148950
00148960
00148970
00148980
00148990
00149000
00149010
00149020
00149030
00149040
00149050
00149060
00149070
00149080
00149090
00149100

```

00149110
00149120
00149130
00149140
00149150
00149160
00149170
00149180
00149190
00149200
00149210
00149220
00149230
00149240
00149250
00149260

GO TO 2
7 *RITE(G,6015)
2 UJSEC JULINMS,NDS,NYS)
DJQEL JULINME,INDF,NYF)
TS=NH\$*J600
TMS=NHNS*60
TS=TH\$TMS*TS-TSEP
TF=NH\$*3600
TVF=NHMF*60
TF=TH\$TMS*TS-TSEP
UJSEC=UJQ-LJQ
UJF=UJFQ-DJQ
S=LJS*80400.00+T3
F=UJF*80400.00+TF
RETURN
END

15N 0026
15N 0027
15N 0028
15N 0029
15N 0030
15N 0031
15N 0032
15N 0033
15N 0034
15N 0035
15N 0036
15N 0037
15N 0038
15N 0039
15N 0040
15N 0041

*****F U R T R A N C R O S S R E F E R E N C E L I S T I N G*****

SYMBOL	INTERNAL STATEMENT NUMBERS
F	0002 0039
S	0002 0038
DT	0002 0012 0013
TF	0025 0039
TS	0032 0038
DJF	0037 0039
DJS	0036 0038
DJO	0002 0036 0037
KPH	0002 0012 0014
NDF	0012 0013 0029
NDS	0014 0013 0028
NHF	0012 0013 0033
NHS	0014 0013 0030
NMF	0012 0013 0029
NMS	0012 0013 0028
NYF	0012 0013 0029
NYS	0012 0013 0028
THF	0023 0035
THS	0030 0032
TSF	0012 0013 0035
TSS	0012 0013 0032
CJF0	0029 0037
DJS0	0028 0036
DJOL	0028 0029
NMNF	0012 0013 0034
NMNS	0012 0013 0031
TMNF	0034 0035
TMRS	0031 0032
TSEP	0002 0032 0035
KLAST	0002 0012 0024
TMCA	0002

***** D F T R A N C R O S S R E F E R E N C E L I S T I N G *****

LABEL	DEF INED	REFERENCES
1	0013	
2	0028	0020
3	0015	0014 0014
4	0023	0020
5	0024	0016 0019 0022
6	0025	0024
7	0027	0024 0024
30	0017	0014
31	0018	0017 0017
32	0020	0017
33	0021	0020 0020
6015	0004	0012
6016	0005	0013
6017	0006	0015
6018	0007	0023
6019	0008	0027
6020	0009	0025
6021	0010	0018
6022	0011	0021

DATE 69.206/19.06.02

OS/360 FORTRAN M

LEVEL 16 (1 JULY 68)

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LCMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=58,SOURCE,EBCDIC,NCLIST,NODECK,LOAD,MAP,NOEDIT,IO,XREF
CSUBROUTINE WRT6
  ISN 0002 SUPEROUTINE WRT6(M,COORD,VELCC,ACC,ELMT,N1,N2,N3,L1,L2,L3) 00151560
  ISN 0003 IMPLICIT REAL*8(A-H,O-Z,*) 00151570
  ISN 0004 DIMENSION COORD(3),VELOC(3),ACC(3),ELMT(6) 00151580
  ISN 0005 22522 FORMAT(1X,15,6D21.14) 00151590
  ISN 0006 22529 FORMAT(1X,15,3D23.16,57X) 00151600
  ISN 0007 IF(L1)100,101,100 00151610
  ISN 0008 100 WRITE(N1,22522) M,(COORD(I),I=1,3),(VELOC(I),I=1,3) 00151620
  ISN 0009 WRITE(N1,22529) M,(ACC(I),I=1,3) 00151630
  ISN 0010 WRITE(N1,22522) M,(ELMT(I),I=1,6) 00151640
  ISN 0011 101 IF(L2)102,103,102 00151650
  ISN 0012 102 WRITE(N2,22522) N,(COORD(I),I=1,3),(VELOC(I),I=1,3) 00151660
  ISN 0013 WRITE(N2,22529) M,(ACC(I),I=1,3) 00151670
  ISN 0014 WRITE(N2,22522) M,(ELMT(I),I=1,6) 00151680
  ISN 0015 103 IF(L3)104,105,104 00151690
  ISN 0016 104 WRITE(N3,22522) M,(COORD(I),I=1,3),(VELOC(I),I=1,3) 00151700
  ISN 0017 WRITE(N3,22529) M,(ACC(I),I=1,3) 00151710
  ISN 0018 WRITE(N3,22522) M,(ELMT(I),I=1,6) 00151720
  ISN 0019 105 RETURN 00151730
  ISN 0020 END 00151740
                                00151750

```


***** F O R T R A N C R O S S R E F E R E N C E L I S T I N G *****

LABEL	DEFINED	REFERENCES
100	U008	0007 0007
101	0011	0007 0011
102	0012	0011 0011
103	0015	0011 0015
104	0016	0015 0015
105	0019	0015
22522	0005	0008 0010 0012 0014 0016 0018
22529	0006	0009 0013 0017

COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=58,SOURCE,EBCDIC,NCLIST,NODECK,LOAD,MAP,NOEDIT,1D,XREF

CSUBROUTINE KKEP

REAL FUNCTION KKEP *B(AM,ECC,SE,CE,ERR,C,NMAX,N)

IMPLICIT REAL*8(A-H,O-Z,S)

C NAME CHANGED 11/12/68 FROM KKEPZ TO KKEP

C PURPOSE

C SOLVES KEPLER'S EQUATION.

C INPUT

C AM = MEAN ANOMALY

C ECC = ECCENTRICITY

C ERR = TOLERANCE FOR (E2-E1)/E2 IN SOLVING KEPLER'S

C EQUATION.

C NMAX = MAXIMUM NUMBER OF ITERATIONS ALLOWED

C IN SOLVING KEPLER'S EQUATION.

C OUTPUT

C SE = SINE OF ECCENTRIC ANOMALY

C CE = COSINE OF ECCENTRIC ANOMALY

C G = LAST CORRECTION (E2-E1)/E2

C IN SOLVING KEPLER'S EQUATION.

C N = SERIAL NUMBER OF CURRENT ITERATION IN SOLVING

C KEPLER'S EQUATION.

C REQUIRED SUBPROGRAMS

C NONE

C

7 E1 = AM + (ECC * CSIN(AM))

N = 0

1 E2 = E1

N = N + 1

SE = CSIN(E1)

CE = CCOS(E1)

E1 = E1 + ((AM - E1 + ECC*SE) / (1.000 - ECC*CE))

IF(E1-FO,C,DO) GO TO 8

G = DAHS((E1 - E2) / E1)

GO TO 9

8 GEC=DOO

9 IF(NMAX)2,2,3

2 IF(G - FRR) 5,5,1

3 PRINT 4

4 PRINT 6,AM,E1,E2,C,N

5 FORMAT (3H0NO CONVERGENCE IN KEPLER'S EQUATION)

6 FORMAT (4D16.8,15)

CALL SLITE(2)

5 KKEP = E1

RETURN

ENC

15N 0004

15N 0005

15N 0006

15N 0007

15N 0008

15N 0009

15N 0010

15N 0011

15N 0012

15N 0013

15N 0014

15N 0015

15N 0016

15N 0017

15N 0018

15N 0019

15N 0020

15N 0021

15N 0022

15N 0023

15N 0024

15N 0025

***** U T R A N C R O S S R E F E R E N C E L I S T I N *****

SYMBOL	INTERNAL STATEMENT NUMBERS
G	CC2 CC13 CC15 CC17 CC19
A	CC2 CC5 CC7 CC7 CC16 CC19
AM	CC2 CC4 CC4 CC10 CC19
CE	CC4 CC4 CC10
E1	CC4 CC6 CC6 CC9 CC10 CC10 CC13 CC19 CC23
E2	CC4 CC13 CC19
SE	CC4 CC6 CC10
ECC	CC2 CC4 CC10
EPF	CC2 CC17
CAS	CC13
CCS	CC4
CSIN	CC4 CC8
NMAX	CC2 CC16
XREP	CC4 CC23
SLITC	CC4

*****F O R T R A N C R O S S R E F E R E N C E L I S T I N G*****

LABEL	DEF INED	REFERENCES	
1	0006	0017	
2	0017	0016	0016
3	0018	0016	
4	0020	0018	
5	0023	0017	0017
6	0021	0013	
7	0004		
8	0015	0011	
9	0016	0014	

APPENDIX B
DESCRIPTION OF INPUT DECK

CARD 1

CARD NO. 1 RUN IDENTIFICATION CARD (CAN CONTAIN ANY DESIRED INFORMATION IN COLUMNS 2 THROUGH 72.)

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CARD 2

CARD NO. 2 RUN CONTROL CARD
 COLUMNS 1-3 INPUT TYPE INDICATOR (INPUT)
 +01 = OSCULATING ELEMENTS
 +03 = INERTIAL POSITION AND VELOCITY
 +04 = BROWER MEAN ELEMENTS
 COLUMNS 4-6 SOURCE OF EARTH CONSTANTS INDICATOR (NCST)
 ANY POSITIVE INTEGER = EARTH CONSTANTS TO BE READ FROM CARDS 4 AND 5.
 ANY NEGATIVE INTEGER OR ZERO : OMIT CARDS 4 AND 5.
 THE FOLLOWING CONSTANTS ARE USED:
 0 = GODDARD EARTH CONSTANTS ARE BEING USED.
 GM = 3.98603200D+05 (KM CUBED)/(SECONDS SQUARED)
 R = 6378.165 KM
 J2 = +1.08230D-03 J3 = -2.300D-06
 J4 = -1.800D-06 J5 = 0.0
 -1 = EARTH CONSTANTS OF THE SIRY PACKAGE ARE BEING USED.
 GM = 3.98626880D+05 (KM CUBED)/(SECONDS SQUARED)
 R = 6378.388 KM
 J2 = +1.08219D-03 J3 = -2.285D-06
 J4 = -2.123D-06 J5 = -2.32D-07
 -2 = GODDARD EARTH CONSTANTS WITH HARMONICS = ZERO ARE BEING USED.
 GM = 3.98603200D+05 (KM CUBED)/(SECONDS SQUARED)
 R = 6378.165 KM
 -3 = INTERNATIONAL EARTH CONSTANTS WITH HARMONICS = ZERO ARE BEING USED.
 GM = 3.98626873D+05 (KM CUBED)/(SECONDS SQUARED)
 R = 6378.388 KM
 COLUMNS 7-9 INDICATOR OF SOURCE FOR TOLERANCE AND MAXIMUM NUMBER OF ITERATIONS ALLOWED IN SOLVING KEPLER'S EQUATION (NERR).
 TOLERANCE IS UPPER LIMIT OF (E2-E1)/E2, WHERE E1 AND E2 ARE VALUES OF THE ECCENTRIC ANOMALY IN TWO SUCCESSIVE ITERATIONS.
 ANY POSITIVE INTEGER = READ TOLERANCE AND MAXIMUM NUMBER OF ITERATIONS ALLOWED IN SOLUTION OF KEPLER'S EQUATION FROM CARD 3.
 ANY NEGATIVE INTEGER OR 0 = OMIT CARD 3. TOLERANCE FOR ECCENTRIC ANOMALY IS -100-13 AND THE MAXIMUM NUMBER OF ITERATIONS IS 50.

CARD 2 (CONTINUED)

COLUMNS 10-12 BROWER TRUNCATION INDICATOR IN SUBROUTINE B9MR
(NDA).

ANY POSITIVE = READ TOLERANCES TO BE USED IN DETERMINING MEAN
INTEGER BROWER ELEMENTS FROM CARD 6.

ANY NEGATIVE
INTEGER OR 0 = OMIT CARD 6. THE FOLLOWING TOLERANCES
ARE USED IN DETERMINING BROWER MEAN
ELEMENTS.

RIGHT ASCENSION OF ASCENDING NODE = .500-10 KM
ECCENTRICITY = .500-14
INCLINATION = .500-11 DEGREES
ARGUMENT OF PERIGEE = .500-11 DEGREES
MEAN ANOMALY = .500-11 DEGREES

COLUMNS 13-15 MAXIMUM NUMBER OF ITERATIONS ALLOWED IN DETERMINING
BROWER MEAN ELEMENTS (IRCI).

COLUMNS 16-18 TYPE OF UNITS USED ON CARDS 8A-8H (UNIT).

0 = UNIT OF LENGTH = MEGAMETERS, UNIT OF TIME = HOUR,
ANGLES IN DEGREES
COEFFICIENTS OF LINEAR TERMS ARE GIVEN PER HOUR
COEFFICIENTS OF QUADRATIC TERMS ARE GIVEN PER
(100 HOURS) SQUARED
COEFFICIENTS OF CUBIC TERMS ARE GIVEN PER
(100 HOURS) CUBED.

OTHER THAN 0 = UNIT OF LENGTH = EARTH'S RADIUS, UNIT OF
TIME = CUT, ANGLES IN RADIANS.

(VANGUARD UNITS)
COEFFICIENTS OF LINEAR TERMS ARE GIVEN PER CUT
COEFFICIENTS OF QUADRATIC TERMS ARE GIVEN PER
(CUT) SQUARED.

COEFFICIENTS OF CUBIC TERMS ARE GIVEN PER
(CUT) CUBED.

COLUMNS 19-21 INTERMEDIATE OUTPUT FROM SUBROUTINE B9MR1 INDICATOR
(NBRWR1).

0 = DO NOT PRINT INTERMEDIATE OUTPUT.

OTHER THAN 0 = PRINT INTERMEDIATE OUTPUT.

COLUMNS 22-24 LONG PERIOD PERTURBATIONS INDICATOR (NLONG)

0 = DO NOT INCLUDE LONG PERIOD PERTURBATIONS

OTHER THAN 0 = INCLUDE LONG PERIOD PERTURBATIONS

COLUMNS 25-27 SHORT PERIOD PERTURBATIONS INDICATOR (NSHORT)

0 = DO NOT INCLUDE SHORT PERIOD PERTURBATIONS

ANY NEGATIVE INTEGER = THE SHORT PERIOD PERTURBATIONS ARE COM-
PUTED WITH E" AND I".

ANY POSITIVE INTEGER = THE SHORT PERIOD PERTURBATIONS ARE COM-
PUTED WITH E' AND I'.

[illegible]

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CARD 3

CARD NO. 3 TOLERANCE AND MAXIMUM NUMBER OF ITERATIONS ALLOWED IN SOLVING KEPLER'S EQUATION. TOLERANCE IS UPPER LIMIT OF $1E2-E1/E2$, WHERE E1 AND E2 ARE VALUES OF THE ECCENTRIC ANOMALY IN TWO SUCCESSIVE ITERATIONS.

NOTE: THIS CARD IS REQUIRED IF AND ONLY IF COLUMNS 7-9 OF CARD 2 (RUN CONTROL CARD) CONTAIN A POSITIVE INTEGER.

COLUMNS 1-8 ECCENTRIC ANOMALY TOLERANCE IN FORMAT D8.2.
COLUMNS 9-12 MAXIMUM NO OF ITERATIONS ALLOWED IN SOLUTION OF KEPLER'S EQUATION.

UU

4 AND 5 EARTH CONSTANTS CARDS

NOTE: THESE CARDS ARE REQUIRED IF AND ONLY IF COLUMNS 4-6 OF CARD NO. 2 (THE RUN CONTROL CARD) CONTAIN A POSITIVE INTEGER.

FOR CARD 5
COLUMNS 1-24 MEAN EQUATORIAL RADIUS OF THE EARTH IN MEGAMETERS
FORMAT IS D24.17

U U U U U U U U U U U U U U U U U U

U U U U U U U U U U U U U U U U U U

U U U U U U U U U U U U U U U U

CARDS NO. 8A THROUGH 8H (8 CARDS)
UNITS ARE DEFINED BY QUANTITY IN COLUMNS 16-18 OF CARD NO. 2 (THE
RUN CONTROL CARD).
THE FORMAT OF EACH OF CARDS 8A-8H IS 3D24.17.

A = SEMI-MAJOR AXIS
E = ECCENTRICITY
I = INCLINATION TO THE EQUATOR
NODE = RIGHT ASCENSION OF ASCENDING NODE
OMEGA = ARGUMENT OF PERIGEE
M = MEAN ANOMALY

CARDS 8A - 8H (CONTINUED)

CARD 8A (COLUMNS 1-3 OF CARD 2 CONTAIN EITHER +01 OR +04)
 COLUMNS 1-24 OSCILLATING (+01) OR BROUWER MEAN (+04) A
 COLUMNS 25-48 OSCILLATING (+01) OR BROUWER MEAN (+04) E
 COLUMNS 49-72 OSCILLATING (+01) OR BROUWER MEAN (+04) I

CARD 8B (COLUMNS 1-3 OF CARD 2 CONTAIN EITHER +01 OR +04)
 COLUMNS 1-24 OSCILLATING (+01) OR BROUWER MEAN (+04) NODE
 COLUMNS 25-48 OSCILLATING (+01) OR BROUWER MEAN (+04) OMEGA
 COLUMNS 49-72 OSCILLATING (+01) OR BROUWER MEAN (+04) M
 NOTE: THESE COMPONENTS ARE AT EPOCH REFERRED TO THE EQUATOR
 AND EQUINOX.

CARD 8A (COLUMNS 1-3 OF CARD 2 CONTAIN +03)
 COLUMNS 1-24 X COMPONENT OF POSITION VECTOR
 COLUMNS 25-48 Y COMPONENT OF POSITION VECTOR
 COLUMNS 49-72 Z COMPONENT OF POSITION VECTOR

CARD 8B (COLUMNS 1-3 OF CARD 2 CONTAIN +03)
 COLUMNS 1-24 X COMPONENT OF VELOCITY VECTOR
 COLUMNS 25-48 Y COMPONENT OF VELOCITY VECTOR
 COLUMNS 49-72 Z COMPONENT OF VELOCITY VECTOR

NOTE: THESE COMPONENTS ARE AT EPOCH REFERRED TO THE EQUATOR
 AND EQUINOX.

CARDS 8A - 8H (CONTINUED)

FOR ALL VALUES IN COLUMNS 1-3 OF CARD 2:

CARD 8C	COLUMNS 1-24	COEFFICIENT OF LINEAR TERM IN A	IN NODE
	COLUMNS 25-48	COEFFICIENT OF LINEAR TERM IN E	IN OMEGA
	COLUMNS 49-72	COEFFICIENT OF LINEAR TERM IN I	IN M
CARD 8D	COLUMNS 1-24	COEFFICIENT OF ADDITIONAL LINEAR TERM	IN NODE
	COLUMNS 25-48	COEFFICIENT OF ADDITIONAL LINEAR TERM	IN OMEGA
	COLUMNS 49-72	COEFFICIENT OF ADDITIONAL LINEAR TERM	IN M
CARD 8E	COLUMNS 1-24	COEFFICIENT OF QUADRATIC TERM IN A	IN NODE
	COLUMNS 25-48	COEFFICIENT OF QUADRATIC TERM IN E	IN OMEGA
	COLUMNS 49-72	COEFFICIENT OF QUADRATIC TERM IN I	IN M
CARD 8F	COLUMNS 1-24	COEFFICIENT OF QUADRATIC TERM	IN NODE
	COLUMNS 25-48	COEFFICIENT OF QUADRATIC TERM	IN OMEGA
	COLUMNS 49-72	COEFFICIENT OF QUADRATIC TERM	IN M
CARD 8G	COLUMNS 1-24	COEFFICIENT OF CUBIC TERM IN A	IN NODE
	COLUMNS 25-48	COEFFICIENT OF CUBIC TERM IN E	IN OMEGA
	COLUMNS 49-72	COEFFICIENT OF CUBIC TERM IN I	IN M
CARD 8H	COLUMNS 1-24	COEFFICIENT OF CUBIC TERM	IN NODE
	COLUMNS 25-48	COEFFICIENT OF CUBIC TERM	IN OMEGA
	COLUMNS 49-72	COEFFICIENT OF CUBIC TERM	IN M

[illegible]

COLUMN	1	BLANK
COLUMNS	2-3	MONTH OF START TIME
COLUMN	4	BLANK
COLUMNS	5-6	DAY OF START TIME
COLUMN	7	BL/NK
COLUMNS	8-11	YEAR OF START TIME
COLUMN	12	BLANK
COLUMNS	13-14	HOUR OF START TIME
COLUMN	15	BLANK
COLUMNS	16-17	MINUTES OF START TIME
COLUMN	18	BLANK
COLUMNS	19-24	SECONDS OF START TIME IN COLUMN
COLUMN	25	BLANK
COLUMNS	26-27	MONTH OF END TIME
COLUMN	28	BLANK
COLUMNS	29-30	DAY OF END TIME
COLUMN	31	BLANK
COLUMNS	32-35	YEAR OF END TIME
COLUMN	36	BLANK
COLUMNS	37-38	HOUR OF END TIME
COLUMN	39	BLANK
COLUMNS	40-41	MINUTES OF END TIME
COLUMN	42	BLANK
COLUMNS	43-48	SECONDS OF END TIME IN COLUMN
COLUMNS	49-59	INCREMENT OF TIME IN COLUMN SECONDS - DECIMAL POINT
COLUMNS	60-65	FREQUENCY OF DATA IN MINUTE OR A NEGATIVE INTEGER
		A POSITIVE INTEGER GREATER THAN ZERO
		KPR) = DATA FOR EVERY KPR)
COLUMNS	60-72	EMPHASIS REQUEST SETTING - THIS IS THE FINAL EMPHASIS OTHER THAN ZERO = THIS REQUEST

ON NEGATIVE INTEGER BE PRINTED.
A POSITIVE INTEGER GREATER THAN 1 WILL BE PRINTED.
(KPK) = DATA FOR EVERY KPTH DATE WILL BE PRINTED.
72 EPHEMERIS REQUEST SENTINEL.
0= THIS IS THE FINAL EPHEMERIS REQUEST
OTHER THAN ZERO = THIS IS NOT THE FINAL EPHEMERIS
REQUEST.

APPENDIX C
MATHEMATICAL DETAILS OF MODIFICATION
OF THE BROUWER THEORY

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APPENDIX C

Mathematical Details of Modification of the Brouwer Theory

The unit of length in this section is the mean equatorial radius of the earth. Let the quantities designated in Section 2 by

$$a_0, e_0, I_0$$

be designated here by

$$A'', E'', I''$$

They are the constant terms in the expressions in the Brouwer theory for a, e, I . Introduce the quantities

$$\epsilon'', P'', \Theta''$$

by means of

$$\begin{aligned}\epsilon'' &= (1 - E''^2)^{1/2} \\ P'' &= A'' \epsilon''^2 \\ \Theta'' &= \cos I''\end{aligned}\tag{C-1}$$

The Keplerian mean motion n_0 is then given by

$$n_0 = \mu^{1/2} A''^{-3/2}\tag{C-2}$$

where μ is the constant appearing in (1). With the adopted unit for lengths and any unit for time μ is determined.

The mean motion of the mean anomaly ℓ_B as used in the Brouwer theory is the first time derivative of the secular portion ℓ_B'' of ℓ_B and is given by

$$\frac{d\ell_B''}{dt} = n_0 \{1 + J_2 B_{21} + J_2^2 B_{22} + J_4 B_4\}\tag{C-3}$$

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where

$$\begin{aligned}
 B_{21} &= \frac{3}{4} P''^{-2} \epsilon'' (-1 + 3 \Theta''^2) \\
 B_{22} &= \frac{3}{128} P''^{-4} \epsilon'' [(-15 + 16 \epsilon'' + 25 \epsilon''^2) \\
 &\quad + (30 - 96 \epsilon'' - 90 \epsilon''^2) \Theta''^2 \\
 &\quad + (105 + 144 \epsilon'' + 25 \epsilon''^2) \Theta''^4] \\
 B_4 &= -\frac{45}{128} P''^{-4} \epsilon'' E''^2 [3 - 30 \Theta''^2 + 35 \Theta''^4]
 \end{aligned} \tag{C-4}$$

Kozai like Brouwer uses as basic dependent variables the Delaunay variables

$$L, G, H, l, g, h$$

defined by (13).

The complete solutions for L, G, H, l, g, h according to Kozai are of the form

$$\begin{aligned}
 L &= L'' + \delta L, \quad l = l'' + \delta l \\
 G &= G'' + \delta G, \quad g = g'' + \delta g \\
 H &= H'' \quad , \quad h = h'' + \delta h
 \end{aligned} \tag{C-5}$$

Here L'', G'', H'' are constants and l'', g'', h'' are linear functions of the time. Whereas $\delta G, \delta l, \delta g, \delta h$ contain long- and short-period perturbations, there are only short-period terms present in δL .

The secular portions of a, e, i are designated by double primes and equations for the quantities

$$a'', e'', i''$$

are obtained from (13) by replacing L, G, H by L'', G'', H'' and a, e, i by a'', e'', i'' .

Then p'' , η'' , θ'' are introduced by means of

$$\begin{aligned} p'' &= a'' (1 - e''^2) \\ \eta'' &= \frac{G''}{L''} = (1 - e''^2)^{1/2} \\ \theta'' &= \cos i'' = H''/G'' \end{aligned} \quad (C-6)$$

The quantity n'' is defined by

$$n'' = \mu^2 L''^{-3} = \mu^{1/2} a''^{-3/2} \quad (C-7)$$

For the computation of the mean motion of the mean anomaly from equation (7.1) in Kozai's article seventeen functions M_i , $i = 0, 1, \dots, 16$ are introduced. They are defined by

$$\begin{aligned} M_0 &= 1 - 5 \theta''^2 & M_1 &= e'' (1 - \theta''^2) (1 - 5 \theta''^2)^{-2} = e'' (1 - \theta''^2)/M_0^2 \\ M_2 &= 3 (1 - \theta''^2) & M_3 &= 1 - 15 \theta''^2 & M_4 &= 5 (1 - 7 \theta''^2) \\ M_5 &= 2 e''^2 (1 - 5 \theta''^2)^{-1} = 2 e''^2/M_0 \\ M_6 &= (1 - 43 \theta''^2 + 155 \theta''^4 - 225 \theta''^6) (1 - 15 \theta''^2) = (1 - 43 \theta''^2 + 155 \theta''^4 - 225 \theta''^6) M_3 \\ M_7 &= 10 (1 - 42 \theta''^2 + 408 \theta''^4 - 1270 \theta''^6 + 1575 \theta''^8) & (C-8) \\ M_8 &= 25 (1 - 19 \theta''^2 + 75 \theta''^4 - 105 \theta''^6) (1 - 7 \theta''^2) = 5 (1 - 19 \theta''^2 + 75 \theta''^4 - 105 \theta''^6) M_4 \\ M_9 &= e''^{-1} (1 - \theta''^2 - 2 e''^2 \theta''^2) & M_{10} &= 16 [(32 - 59 \theta''^2) + 6 e''^2 (8 - 13 \theta''^2)] \\ M_{11} &= 4 (1 - \theta''^2) (1 - 5 \theta''^2)^{-2} = \frac{4}{3} M_2/M_0^2 & M_{12} &= e''^2 (1 - 5 \theta''^2)^{-3} = e''^2/M_0^3 \\ M_{13} &= (4 - 127 \theta''^2 + 480 \theta''^4 - 525 \theta''^6) M_3 \\ M_{14} &= 10 (4 - 139 \theta''^2 + 1273 \theta''^4 - 3805 \theta''^6 + 3675 \theta''^8) \\ M_{15} &= 5 (4 - 63 \theta''^2 + 232 \theta''^4 - 245 \theta''^6) M_4 \\ M_{16} &= - (1 - \theta''^2)^{-1/2} \theta'' [2 (1 - \theta''^2) + e''^2 (2 - 3 \theta''^2)] \end{aligned}$$

Then the auxiliary constants

$$a_0, e_0, i_0, p_0,$$

defined by equations (6.27) - (6.30) in Kozai's article and the constants

$$\eta_0, \theta_0$$

are obtained by successive approximations from

$$\begin{aligned} a_0 &= a'' \left[1 + \frac{3}{4 p_0^2} (1 - e_0^2)^{1/2} (1 - 3 \theta_0^2) J_2 \right] \\ e_0 &= e'' + (J_2^2 / 1024 a'' p''^3) M_1 \left\{ M_2 \left[M_3 + \frac{J_4}{J_2} M_4 \right]^2 \right. \\ &\quad \left. + M_5 \left[M_6 + M_7 \frac{J_4}{J_2^2} + M_8 \left(\frac{J_4}{J_2^2} \right)^2 \right] \right\} \\ &\quad + (J_3^2 / J_2^2) (1/16 a'' p''^4) M_9 \end{aligned} \tag{C-9}$$

$$\begin{aligned} i_0 &= i'' - (J_2^2 / 2048 p''^4) \sin 2 i'' \\ &\quad \times \left\{ M_{10} + M_{11} \left[M_3 + M_4 \frac{J_4}{J_2^2} \right]^2 \right. \\ &\quad \left. + M_{12} \left[M_{13} + M_{14} \frac{J_4}{J_2} + M_{15} \left(\frac{J_4}{J_2} \right)^2 \right] \right\} \\ &\quad + (J_3^2 / J_2^2) (1/16 p''^2) M_{16} \end{aligned}$$

$$p_0 = a_0 (1 - e_0^2)$$

$$\eta_0 = (1 - e_0^2)^{1/2}$$

$$\theta_0 = \cos i_0$$

The mean motion of the mean anomaly as used in the Kozai theory will be denoted now by $d\ell_K''/dt$ and is equal to the first derivative of the secular portion ℓ_K'' of ℓ_K and is given according to equation (7.1) in Kozai's article by

$$\frac{d\ell_K''}{dt} = n'' \cdot \{1 + K_{21} J_2 + K_{22} J_2^2 + K_4 J_4\} \quad (C-10)$$

where

$$\begin{aligned} K_{21} &= -\frac{3}{4 p_0^2} \eta_0 (1 - 3 \theta_0^2) \\ K_{22} &= \frac{3}{128 p_0^4} \eta_0 [10 (1 - 6 \theta_0^2 + 13 \theta_0^4) \\ &\quad - 5 e_0^2 (5 - 18 \theta_0^2 + 5 \theta_0^4) \\ &\quad - 32 \eta_0 (1 - 3 \theta_0^2)^2] \end{aligned} \quad (C-11)$$

$$K_4 = -\frac{45}{128 p_0^4} e_0^2 \eta_0 (3 - 30 \theta_0^2 + 35 \theta_0^4)$$

In the expression for K_4 Kozai's p_0^2 has been replaced by p_0^4 .

If the Brouwer theory is modified so that the second order short-period terms in a are included, then it may not be unreasonable to assume that

$$a'' = A'' \quad (C-12)$$

and that

$$e'' = E'', \quad i'' = I'' \quad \text{to order } J_2 \quad (C-13)$$

Then from (C-2), (C-7), (C-12)

$$n_0 = n'' \quad (C-14)$$

so that (C-3) and (C-10) lead to

$$\begin{aligned}\delta n &= \frac{d\ell''_{\mathbf{K}}}{dt} - \frac{d\ell''_{\mathbf{B}}}{dt} \\ &= n_0 \{ (K_{21} - B_{21}) J_2 + (K_{22} - B_{22}) J_2^2 + (K_4 - B_4) J_4 \}\end{aligned}\tag{C-15}$$

According to (C-13) $e'' - E''$ and $i'' - I''$ are of order J_2^2 and according to (C-9) the quantities δe and δi defined by

$$\begin{aligned}\delta e &= e_0 - e'' \\ \delta i &= i_0 - i''\end{aligned}\tag{C-16}$$

and

$$\delta \Theta = -\sin I'' \delta i\tag{C-17}$$

are of order J_2^2 .

It may be conceivable that the differences $e'' - E''$ and $i'' - I''$ are even smaller than the second order of J_2 . Only if we make this assumption will we be able to derive a fairly simple approximation for $d\ell''_{\mathbf{K}}/dt - d\ell''_{\mathbf{B}}/dt$. The assumption implies

$$\begin{aligned}e_0 &= E'' + \delta e \\ i_0 &= I'' + \delta i \\ \theta_0 &= \Theta'' + \delta \Theta\end{aligned}\tag{C-18}$$

where δe and δi are computed from (C-16) and $\delta \theta$ from (C-17). Using two successive iterations of (C9) we find that to order J_2^2

$$a_0 = A'' [1 + \gamma J_2 - 2 \gamma^2 J_2^2]\tag{C-19}$$

where

$$\gamma = \frac{3 \epsilon''}{4 P''^2} (1 - 3 \Theta''^2)\tag{C-20}$$

Then we obtain the following correction

$$\delta n = n_0 \left\{ J_2^3 (\gamma^3 - 4 \gamma B_{22}) - 4 B_4 \gamma J_2 J_4 + \frac{3 E''}{\epsilon''^2} J_2 \delta e - \frac{6 \Theta''}{1 - 3 \Theta''^2} J_2 \delta \Theta \right\} \quad (C-21)$$

to the mean motion of the mean anomaly computed according to Brouwer. This correction will probably make the mean motion to be closer to the value according to Kozai.

To free oneself from above assumptions, one would have to make use of more portions of the Kozai theory. This probably would mean so much involvement with the Kozai theory that it might just as well be used exclusively and the purpose of deriving a reduction from Brouwer's to Kozai's mean motion would be defeated.

The long period portions L' , G' , H' , ℓ' , g' , h' of L , G , H , ℓ , g , h are given by

$$\begin{aligned} L' &= L'' & , & \ell' = \ell'' + \delta_L \ell \\ G' &= G'' + \delta_L G & , & g' = g'' + \delta_L g \\ H' &= H'' & , & h' = h'' + \delta_L h \end{aligned} \quad (C-22)$$

there being no long-period perturbations in L and H .

The quantities

$$\delta_L G, \delta_L \ell, \delta_L g, \delta_L h$$

are the long-period perturbations in G , ℓ , g , h .

We define

$$a', e', i'$$

by means of (13) by replacing L , G , H , a , e , i by the primed quantities L' , G' , H' , a' , e' , i' . We also introduce η' , θ' by means of

$$\begin{aligned} \eta' &= (1 - e'^2)^{1/2} = G'/L' \\ \theta' &= H''/G' = \cos i' \end{aligned} \quad (C-23)$$

Let f' be the true anomaly and $\rho' = r'/a'$ the ratio r/a computed from e' and ℓ' . Finally define g^*, B'_{20}, B'_{22} by

$$g^* = g' - \frac{3\mu^2 J_2}{4G'^4} (1 - 5\theta'^2) (f' - \ell')$$

$$B'_{20} = -\frac{1}{4} (1 - 3\theta'^2) \quad (C-24)$$

$$B'_{22} = \frac{3}{4} (1 - \theta'^2)$$

The perturbations δL of L , which consist of short-period terms, are then given by (see equation (3.8) in Kozai's article)

$$\delta L = \frac{\mu^2 J_2}{L'^3} \left\{ B'_{20} \left[\frac{a'^3}{r'^3} - \eta'^{-3} \right] + B'_{22} \frac{a'^3}{r'^3} \cos 2(f' + g^*) \right\} + \delta_2 L \quad (C-25)$$

where $\delta_2 L$ are second order terms due to J_2^2, J_3, J_4 . They have been programmed for an electronic computer by Agreen and Fisher (1968).

The first term of the right hand member of (C-25) represents the first order term in δL according to Kozai. Brouwer who uses only first order terms in δL which he then converts to a perturbation in a uses a similar expression but instead of g^* he uses g' . In order to facilitate comparison of Kozai's expression for δL with Brouwer's expression to be given below we rewrite (C-25) and make use of (C-24). We then obtain

$$\delta L = \delta_1 L + \delta'_2 L + \delta_2 L \quad (C-26)$$

where

$$\delta_1 L = \frac{\mu J_2}{4L'a'} \{ (-1 + 3\theta'^2) (\rho'^{-3} - \eta'^{-3}) + 3(1 - \theta'^2) \rho'^{-3} \cos 2(f' + g') \} \quad (C-27)$$

and

$$\delta'_2 L = \frac{9\mu^4 J_2^2}{8G'^7} \frac{a'^3}{r'^3} \eta'^3 (1 - 5\theta'^2) (1 - \theta'^2) (f' - \ell') \sin 2(f' + g') \quad (C-28)$$

The quantity $\delta'_2 L$ represents the reduction of the first order term of (C-25) with g^* replaced by g' to the form as it appears in (C-25).

From (13) and (C-26) we obtain to the second order

$$a = a'' + \delta_1 a + \frac{1}{\mu} 2L'' (\delta'_2 L + \delta_2 L) + \frac{\delta_1^2 a}{4 a''} \quad (C-29)$$

where

$$\begin{aligned} \delta_1 a &= \frac{2L' \delta_1 L}{\mu} = \frac{1}{2} \frac{J_2}{a'} \{ (-1 + 3\theta'^2) (\rho'^{-3} - \eta'^{-3}) + 3(1 - \theta'^2) \rho'^{-3} \cos 2(f' + g') \} \\ &= J_2 \phi \left(a', \theta', \eta', \frac{r'}{a'}, f', g' \right) \end{aligned} \quad (C-30)$$

Here

$$\begin{aligned} \phi \left(a, \theta, \eta, \frac{r}{a}, f, g \right) &= \frac{1}{2a} \left\{ (-1 + 3\theta^2) \left(\frac{a}{r} - \eta^{-3} \right) \right. \\ &\quad \left. + 3(1 - \theta^2) \frac{a^3}{r^3} \cos 2(f + g) \right\} \end{aligned} \quad (C-31)$$

Here r/a and f are the ratio of radius vector and semi-major axis and the true anomaly computed from the eccentricity and mean anomaly according to the laws of Keplerian motion.

According to the Brouwer theory

$$a = A'' + \delta_s A \quad (C-32)$$

where $\delta_s A$ are the short-period terms in a assumed to be computed with e', I' . This expression is

$$\delta_s A = J_2 \phi \left(A', \Theta', \epsilon', \frac{r'}{A'}, f', g'_B \right) \quad (C-33)$$

where ϕ is the same function as the one appearing in (C-30), where g'_B is the long-period portion of g , according to Brouwer, and where r'/A' and f' are the ratio of radius vector and semi-major axis and true anomaly computed from the long-period portions of the eccentricity and mean anomaly.

From (C-29) and (C-32) follows

$$a'' = A'' + \{(\delta_s A)_0 - (\delta_1 a)_0\} - \frac{1}{\mu} 2L'' \left[(\delta'_2 L)_0 + (\delta_2 L)_0 \right] - \frac{(\delta_1 a)_0^2}{4a''} \quad (C-34)$$

where parentheses with subscripts 0 indicate values at the epoch.

Let us assume that the long-period portions of e , i , Ω , ω , M as computed by Brouwer and Kozai differ at most by terms of order J_2 , which is a reasonable assumption.

Since in (C-30) and (C-33) occur the same function ϕ the difference $\delta_1 a - \delta_s A$ is at least of order J_2^2 . Then (C-34) implies that $a'' - A''$ will be at least of order J_2^2 . Also, the first two terms in the right hand member of (C-29) will be of the same form as the right hand member of (C-32).

Thus while we cannot prove it, it may be conceivably true that to order J_2^2 (C-29) may be replaced by

$$a = A'' + \delta_s A + \frac{1}{\mu} 2L'' (\delta'_2 L + \delta_2 L) + \frac{\delta_s^2 A}{4A''} \quad (C-35)$$

where $\delta'_2 L$, $\delta_2 L$ are based on quantities computed according to Brouwer instead of Kozai. The terms in (C-35) additional to A'' and $\delta_s A$ consisting of second order short-period terms may lead to an improvement of the Brouwer expression for a .

We might note again, that the term with $\delta'_2 L$ allows for the fact that Brouwer uses g' while Kozai uses g^* in the first order perturbations of L . The term $\delta_2 L$ represents the second order terms in L in the Kozai theory due to J_2^2 , J_3 , J_4 but based on quantities obtained in the Brouwer theory. The last term in (C-35) is a second order term due to the conversion from L to a .

The modifications of the Brouwer theory given by (C-21) and (C-35) will be tested experimentally. Results will be reported in a later report.

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